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Experimental and numerical assessment of fracture toughness of dual-phase austempered ductile iron

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ABSTRACT The effects of the microstructure topology on the fracture toughness of dual-phase austempered ductile iron are studied in this paper by means of finite element modelling and experimental testing. To this end, specimens with matrix microstructures ranging from fully ferrite to fully ausferrite were studied and the preferential zones and phases for crack propagation were identified in every case. The effectiveness of the ausferrite phase as a reinforcement of the ferritic matrix via the encapsulation of the brittle and weak last-to-freeze (LTF) zones was confirmed. The toughening mechanism is consequence of the increment in the crack path longitude as it avoids the encapsulated LTF zones. Besides, the presence of small pools of allotriomorphic ferrite increase the crack propagation resistance of the ausferrite-ferrite matrices.

> **Keywords** dual-phase ADI; ductile iron; finite element modelling; fracture toughness; matrix microstructure.

NOMENCLATURE

E =Young modulus

 $G_n^0 = \text{mode-I fracture energy}$

 $G_{\circ}^{0} = \text{mode-II fracture energy}$

 $K_{\rm IC}$ = Fracture toughness

 $t_n^0 = \text{maximum normal traction}$

 $t_{s}^{0} = \text{maximum shear traction}$

 ε_{ut} = ultimate strain

 $\sigma_{0,2}$ = yield stress

 σ_{ut} = ultimate tensile stress

 $\nu = Poisson ratio$

INTRODUCTION

The engineering community is strongly pressed to produce lighter, stronger and stiffer metallic parts. Ductile iron (DI) can be a material of choice to fabricate numerous parts, because it is suitable to produce high resistance cast parts of complex shape and relatively inexpensive. For this reason DI is used to successfully replace cast and forged steels parts in a large number of applications. The mechanical properties of sound DI cast parts depends mainly on the matrix microstructure (type, amount and distribution of microconstituents), the graphite phase and the cast defects present mainly in the last-to-freeze (LTF) regions. A wide range of microstructures, and consequently

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mechanical properties, can be obtained by using different heat treatments.²

Currently, DI producers work in the continuous improvement of DI properties while seeking for new applications. In particular, the market of safety critical parts in the automotive industry is a main target where high strength and toughness are customary requirements. In this regard special thermal cycles have been developed in order to obtain mixed microstructures. Wade et al.,3 He et al.4 and Rashidi et al.5 have obtained mixed microstructures starting from rapid uncompleted austenitizations into the austenitic field. The austenite nucleated mainly surrounding graphite nodules is transformed into pearlite, bainite or martensite depending on the cooling rate. The final phases obtained consisted of ferrite and pearlite, bainite or martensite. The relationship between

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the relative quantities of each phase is controlled by means of the austentizing time (the longer the time, the higher the amount of austenite in the matrix). On the other hand, Galarreta et al.⁶ have studied the procedures to control the morphology of small amounts of free ferrite in pearlitic DI by using heat treatment cycles involving solid phase transformations within the intercritical interval of the Fe-C-Si diagram.

A new kind of DI usually referred to as dual-phase austempered ductile iron (dual-phase ADI) is an active field of research and development. The matrix of dualphase ADI is composed of ausferrite (regular ADI microstructure) and free (or allotriomorphic) ferrite, which is obtained by subjecting DI to special heat treatments. Different methodologies have been used to obtain this kind of microstructures. Aranzabal et al. obtained dualphase ADI by means of changes in the chemical composition of the melt by heat treating ferritic DI at different austenitizing temperatures. In that process the austenitizing stage is followed by an austempering step in order to obtain a final microstructure composed by ferrite and ausferrite. On the other hand, Wade et al.³ and Verdu et al.⁸ have obtained dual-phase ADI microstructures by means of heat treatments similar to those used by He et al.4 and Rashidi et al.,5 based in quick and uncompleted austenitizations in the austenitic field. However, in these cases the austenite obtained at high temperature is transformed into ausferrite as a result of an austempering step.

An alternative methodology to obtain dual-phase ADI has been used by Basso et al. 9,10 and Kilicli et al. 11,12 It consists of an incomplete austenitization stage at temperatures within the intercritical interval followed by an austempering step in a salt bath to transform the austenite into ausferrite. This heat treatment has been allowed to obtain microstructures composed of different combinations of free ferrite and ausferrite, depending on the austenitizing temperature. The main characteristic of this material is the improvement of the fracture and fatigue properties by controlling the relative amount of the present phases and their morphologies. 4–12

ROLE OF MIXED MICROSTRUCTURES ON MECHANICAL PROPERTIES OF DUCTILE IRON

He et al.,⁴ Rashidi et al.,⁵ Dai et al.¹³ and Verdu et al.⁸ focused the development of thermal cycles to encapsulate the graphite nodules with different types of microconstituents (pearlite, bainite and martensite) in order to improve mechanical properties. The goal of these authors was to improve the mechanical properties of ferritic DI surrounding graphite nodules (phase with no resistance) with a high resistant second phase. Based on the

Table 1 Tensile properties for dual-phase ADI compared with ferritic and ausferritic matrices. (austempering temperature 350 °C). Data from reference⁹

Matrix	$\sigma_{\rm max}$ (MPa)	$\sigma_{0.2}$ (MPa)	δ (%)
100% ferrite	455	332	26.5
92% ferrite-8% ausferrite	486	348	26
80% ferrite-20% ausferrite	521	354	24
50% ferrite-50% ausferrite	583	389	19.8
35% ferrite-65% ausferrite	728	410	18
5% ferrite-95% ausferrite	1054	707	10.8
100% ausferrite	1085	729	10.6

same tactic, Wade et al.³ and Verdu et al.⁸ have studied the mechanical properties of dual-phase ADI microstructures, encapsulating graphite nodules with ausferrite. They found that the yield stress, tensile strength and hardness increase with increasing the ausferrite volume fraction.³ In particular, microstructures composed of 80% ferrite and 20% ausferrite allowed reaching increments in tensile strength and fatigue life when compared with fully ferritic DI.⁸

Kilicli et al. 11,12 have studied the mechanical properties of dual-phase ADI austempered at 365 °C, on a wide range of microstructures composed by different quantities of ferrite and ausferrite. The resulting microstructures exhibited higher ductility than the fully ausferritic ones. In particular, samples with 45% ausferrite and 65% ferrite showed the best combinations of strength and toughness.

The authors of this paper have also contributed in the field of dual-phase ADI, analysing the effect of austempering temperature and other variables on the final material microstructure and properties, focusing particularly on microstructures where the LTF zones are encapsulated by ausferrite. ^{9,10} These previous results are reported in Table 1 and Fig. 1 for tensile and fracture properties,

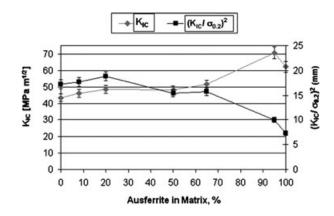


Fig. 1 Fracture properties for dual-phase ADI matrices compared with ferritic and ausferritic matrices (austempering temperature $350\,^{\circ}\text{C}$). Data from reference.⁹

respectively. As it can be seen, the fracture toughness, $K_{\rm IC}$, increases with the ausferrite content. Besides, the crack critical size, defined as the square of the ratio between the fracture toughness and the yield stress, $(K_{\rm IC}/\sigma_{0.2})$, shows it highest value for dual-phase ADI matrices composed of allotriomorphic ferrite and small amounts of ausferrite (less than 25%). This has been attributed to the encapsulating effect of the ausferrite located around the LTF zones. In fact, these are the regions where the alloying elements with direct segregation tendency concentrates and most of the cast defects, such as inclusions, carbides, pores, etc. are present. These regions are generally weak zones expecting them to be the preferential and fast path for crack propagation.

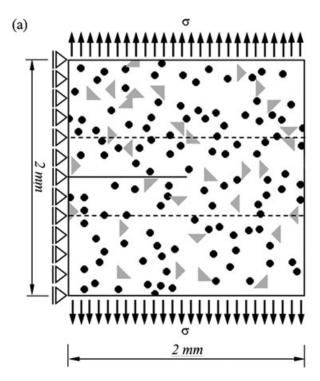
The promising obtained results motivated the continuity of the research. In this work, numerical and experimental assessments on the effect of the encapsulation of LTF zones on the fracture properties of dual-phase ADI as well as the mechanics of crack propagation are studied.

COMPUTATIONAL MODELLING

The effect of the dual-phase microstructure topology on the crack behaviour is studied in this section by means of finite element modelling. An antecedent in this field is the work by Ortiz et al.¹⁴ who used boundary element models and weight-function analyses to study the effect of microcracking on the mechanics of fatigue crack growth in ADI.

The crack behaviour was analysed by means of a series of two-dimensional finite element models for the material microstructure in the vicinity of the crack tip. The models were designed to mimic the microstructures of the materials, which will be described later in Section 4. Typical model geometry is illustrated in Fig. 2a. It consists in a $4 \text{ mm} \times 2 \text{ mm}$ sample with a central crack of length 2 mmunder remote uniaxial tension. As it is shown in Fig. 2a, only one half of the specimen is modelled after assuming symmetry with respect to the direction perpendicular to the crack at its midpoint. Each model contains 113 nodules of diameter 50 μ m. The nodules are randomly distributed and they constitute 8% of the sample volume. The LTF zones are idealized of isosceles triangular-shape with dimensions 80 μ m base and 40 μ m height. Thirtyone LTF zones constituting 5% of the sample volume are randomly distributed over the model domain. The LTF zones are surrounded with a layer of material representing the encapsulations. Two cases were considered for the encapsulation extent: (1) 2.5% and (2) 8.5% of the sample volume. It is worth mentioning that the model does not account for the effect of grain boundaries.

Material properties assigned to the different model regions are reported in Table 2. The matrix was assigned



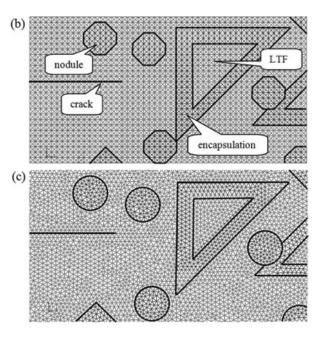


Fig. 2 (a) Model geometry and dimensions; strategies used for the discretization of the FE model: (b) regular and (c) irregular mesh.

an elastoplastic constitutive law with the properties of a fully ferritic DI. The nodules were considered to behave linear-elastic with the properties of bulk graphite. LTF zones were assimilated to pearlite DI in the ascast condition (this is before the heat treatment) and to ADI after the heat treatment. In order to account for the

Table 2 Material parameters for the model regions. Data from references^{1,21}

Model region	Description	Young modulus <i>E</i> (GPa)	Poisson ratio, v	Yield stress σ_y (MPa)	Ultimate stress, σ_{ut} (MPa)	Ultimate strain, ε_{ut}	Fracture strength, $K_{\rm IC}$ (MPa· \sqrt{m})
Matrix	Ferrite	175	0.3	320	450	0.27	40
Nodules	Graphite	16	0.26	60	60	_	2
LTFs	Degraded pearlite	175	0.3	200	200	_	4
Transformed LTFs		175	0.3	250	367	0.1	22
Encapsulation	ADI	175	0.3	750	1100	0.10	65

solidification defects present in LTF zones (microsegregation, porosity, inclusions, etc.), they were assigned the properties of degraded pearlite DI or ADI whether the model represented their conditions before or after the heat treatment, respectively. Degraded properties were arbitrarily defined with the yield stress and the fracture energy equal to one third those of regular pearlite DI or ADI, respectively. The encapsulations were assigned an elastoplastic constitutive law with the properties of ADI.

The finite element models were discretized using threenode linear triangular elements. The triangles were connected to each other using four-node cohesive elements, which account for material damage and crack propagation. The material properties assigned to the cohesive elements are reported in Table 3. The details about the cohesive element formulation and setting are given in the Appendix. In order to verify the independence of the results on the mesh size and orientation the models were discretized using regular and irregular meshes (see Fig. 2b & c, respectively) and elements with sizes ranging from 5 to 8 μ m. The models were solved for plane stress and plane strain conditions. In every case it was verified that the computed results were independent of the element size, the mesh orientation and the assumption of the plane stress or plain strain condition.

Preliminary tests not reported in this work showed that the material damage occurs in the region next to the crack tip. So that, and in order to reduce the model computational cost, the cohesive elements were only used in the central third of the model (the region delimited by dashed lines in Fig. 2a). Models were solved using Abaqus/Explicit running on a Beowulf Cluster with 8

Pentium 4 PCs. The remote traction load was applied incrementally until the complete failure of the sample using a 500 second-long time step. In every case the appropriate mass scaling procedures were used in order to ensure quasi-static loading conditions. To this end it was verified that the model kinetic energy was always less than 2% the external work. For further details on the model implementation please refer to the Abaqus Theory and Analysis manuals.¹⁵

Models were used to study the following microstructure configurations:

- (a) Ferritic microstructure: ferritic matrix with graphite nodules and LTF zones.
- (b) Dual-phase ADI case 1: ferritic matrix with graphite nodules and encapsulated LTF zones. Encapsulation extent: 2.5% of the sample volume.
- (c) Dual-phase ADI case 2: ferritic matrix with graphite nodules and encapsulated LTF zones. Encapsulation extent: 8.5% of the sample volume.
- (d) ADI microstructure: limiting case consisting in a completely ausferritic matrix with graphite nodules and LTF zones.

EXPERIMENTAL ANALYSES

It is presented in this section an experimental test devoted to monitor the crack path within the material microstructure. Since it is practically impossible to monitor the crack path during a traction test, the experimental analyses in this section consists in observations performed on stable

Table 3 Material parameters for cohesive interfaces

Interface	Description	Max. normal traction t_n^0 (MPa)	Max. shear traction t_s^0 (MPa)	Mode-I fract. energy G_n^0 (J/m ²)	Mode-II fract. Energy G_s^0 (J/m ²)
Matrix-matrix	Ferrite	480	554	8320	11 090
Nodule-nodule	Graphite	60	69	233	310
LTF-LTF	Degraded pearlite	200	231	924	1233
Transformed LTF-LTF	Degraded ADI	375	433	2441	3255
Encapsulation	ADI-ADI	1125	1299	21 970	29 300

Table 4 Chemical composition of the material (weight%)

%C	%Si	%Mn	%Cu	%Mg	%S, P	%CE
3.35	3.21	0.46	0.94	0.04	< 0.02	4.26

growing cracks induced by means of cyclic loading. It is worth mentioning that it is not the intention of these analyses to evaluate the material fatigue properties. Although the differences between the mechanisms governing crack propagation for monotonous and cyclic loads, it is expected that the comparison of the propagations paths could provide useful information to better understand the effect of the microstructure on the material fracture toughness.

Material and sample preparation

A ductile cast iron melt was prepared using a medium frequency induction furnace at an industrial foundry plant (Megafund S.A., Argentina). Copper content was chosen 0.90% in order to improve the material austemperability. The melt was poured in 25 mm-thick Y-block-shaped sand moulds. Round and prismatic samples were cut from the Y-blocks and used to prepare test specimens. The chemical composition of the material is listed in Table 4. The as-cast microstructure was pearlitic-ferritic, and the nodularity exceeded 90%, according to the ASTM A-247 standard.

All the samples were firstly ferritized by means of an annealing heat treatment. Four dual-phase ADI microstructures containing different relative quantities of ferrite and ausferrite were obtained by heat treatments. The heat treatments consisted of a partial austenitization within the intercritical temperature interval at 780, 800, 820 and 860 °C during one hour followed by an austempering at 350 °C for 90 min in all the cases. The intercritical temperature interval for the melt was established by employing the methodology described in previous papers by the authors. 9,10

In addition, samples with entirely ferritic and ausferritic matrices were also produced to be used as references. Standard ADI samples were obtained by means of a heat treatment consisting in a complete austenitizing stage at 910 °C for one hour followed by an austempering at 350 °C for 90 min.

The ferritizing and the austenitizing treatments were both performed using electric furnaces, while a 500 kg salt bath was used for the austempering. Microconstituents quantification was performed by means optical microscopy and image-processing using the software Image-Pro Plus.¹⁶

Table 5 shows the identification of the samples, their intercritical austenitization temper-

 Table 5
 Sample designation, austenitizing temperature and phase percentages

Sample	Austenitization, T (°C)	Phases in the matrix (%)
F	_	100% ferrite
DP1	780	90% ferrite-10% ausferrite
DP2	800	80% ferrite-20% ausferrite
DP3	820	50% ferrite-50% ausferrite
DP4	860	5% ferrite-95% ausferrite
ADI	910	100% ausferrite

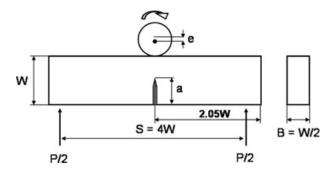


Fig. 3 Experimental set up of crack propagation test.

atures and the final percentages of the matrix microconstituents.

Fractographic analyses

Crack propagation tests were performed using single edge notched specimens under bending load, SEN (B), similar to those specified by ASTM E399 for fracture toughness tests. The sample dimensions were: thickness, B=10 mm; width, W=20 mm and span, S=80 mm. The machined notch was 9 mm length and 1.5 mm width, with an angle of 45° (see Fig. 3).

Crack propagation was induced by means of displacement-controlled cyclic load by using a mechanical testing machine with a double-eccentric actuator (see Fig. 3). A constant eccentricity e=0.145 mm was chosen for all tests. This eccentricity resulted in stable crack propagation under small-scale plasticity condition. To this end it was verified that the crack plastic zone extent, r_p , never exceeded 1% the crack length, accordingly to procedures reported in the literature.¹⁷ Cyclic load was applied at a frequency of 6 Hz.

The experimental methodology consisted in: (a) to stop the fatigue test periodically, (b) to polish the lateral specimen faces, (c) to etch the specimen lateral faces with Nital 2%, (d) to perform optical microscopic observations and (e) to document the crack propagation length and path by means of micrographs. The stops for observations were performed at intervals of 15 min until the sample broke apart.

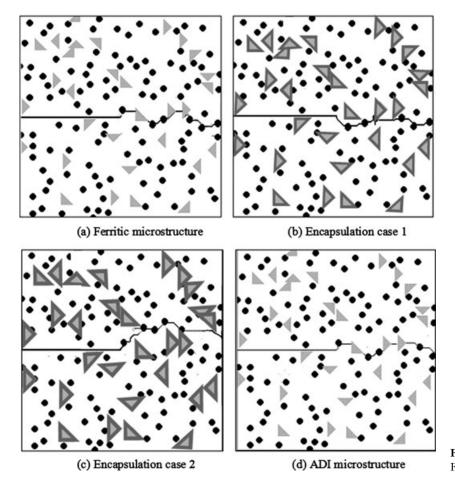


Fig. 4 Crack propagation paths for the four FE models.

RESULTS AND DISCUSSION

Numerical modelling

Figure 4 illustrates the resulting crack propagation paths for the four modelled microstructures. It can be observed from Fig. 4a that the crack path in the ferritic microstructure follows the LTF zones. On the other hand, for the encapsulated 1 and encapsulated 2 models the crack advances across the ferritic matrix and avoiding the LTF zones (see Fig. 4b & c). The greater the encapsulation extent, the more intricate the resulting crack paths are. Finally, the crack path for the ADI microstructure (see Fig. 4d) is almost identical to that of the ferritic microstructure, since the transformed LTF zones are once again the regions of the model with the weakest fracture properties.

Table 6 reports the failure stress and damage energy results computed for the four configurations. The failure stress is the maximum traction attained by the sample up to the moment the crack propagates across the complete ligament (this is, the sample splits into two parts). The damage energy is the total energy dissipated by the cohesive elements (variable ALLDMD in Abaqus/Explicit) up to the moment of the sample splits into two parts. Both

 $\begin{tabular}{ll} \textbf{Table 6} & Failure traction and damage energy results from the finite element analyses \\ \end{tabular}$

	Normalized failure stress	Normalized damage energy
Ferrite microstructure	1.00	1.00
Encapsulation case (1)	1.32	3.89
Encapsulation case (2)	1.36	5.28
ADI microstructure	2.14	5.47

sets of results are reported normalized with respect to the value computed for the ferritic microstructure. The increment in the failure stress for the encapsulated case (1) is 32% with respect to the ferritic matrix, while the increment in the damage energy is nearly 390%. Besides, the encapsulation extent has a marginal effect on the failure stress. The failure stress for the encapsulation case 2 exhibits only an increment of 4% with respect to the case 1. On the other hand, the encapsulation extent has a strong influence on the damage energy, which rises 35% with respect to the encapsulation case 1 and up to nearly 530% when compared to that of the ferritic matrix. This result

can be justified by the increment in the crack path longitude, which avoids the encapsulated LTF zones. Finally, the ADI microstructure results in an important increment of the failure stress (214% larger than that of the ferritic matrix) while it has almost no effect on the damage energy. This last result can be explained arguing that the relative reduction in the longitude of the crack path in the ADI microstructure with respect to the encapsulated cases compensates the increment in the matrix fracture toughness as it is has been transformed from ferrite to ADI (see Table 2).

It is worth to mention that the numerical values for the relative increments or reductions in the failure stress and damage energy reported in the previous paragraph are only indicative. The computed values depend on the material properties specified to the different model regions (see Tables 2 & 3), which were not measured for the actual tests specimens but set using the best criteria of the authors. However, the trend of the results can be credited. To this extent it is observed that the numerical results for the damage energy can qualitative explain the trend of the experimental tests reported by Basso et al.9 and depicted in Fig. 1. Starting from a fully ferrite matrix, the increment of the ausferrite content leads to the encapsulation of the LTF zones what results in increments in the fracture energy (see results for the ferrite and encapsulated cases 1 and 2 in Table 6) but has a marginal effect on the $\sigma_{0.2}$ (see the $\sigma_{0.2}$ data for the 100, 95 and 80% ferrite samples in Table 1). This behaviour can explain the rise in the values of $(K_{\rm IC}/\sigma_{0.2})^2$ in Fig. 1 when the ausferrite content is incremented from 0 to 20%. However, this trend for the $(K_{\rm IC}/\sigma_{0.2})^2$ is reverted for matrices with high contents of ausferrite forming a continuous austenitic phase in the material microstructure. The change in the crack path behaviour does not result in further increments in the damage energy (consider as a limiting case the damage energy for the ADI model in Table 6), but there are important improvements in the $\sigma_{0,2}$ (see the $\sigma_{0,2}$ data for the 33 and 5% ferrite samples in Table 1). As a consequence the $(K_{\rm IC}/\sigma_{0.2})^2$ diminishes after an ideal amount of ausferrite is introduced into the material microstructure.

Experimental analyses

The results of the fractographic analyses show that the fracture path of entirely ferritic matrices (samples F) exhibits transgranular propagation (see Fig. 5) and it mainly follows LTF zones (see Fig. 6 & 7). Zones with high plastic deformation are evident in the fracture path and in its vicinity (see Fig. 5). Branching advance of the main crack is also observed (see Fig. 7). It can also be seen in Fig. 5–7 that although the crack crosses several nodules, nodules do not constitute the preferential propagation path.

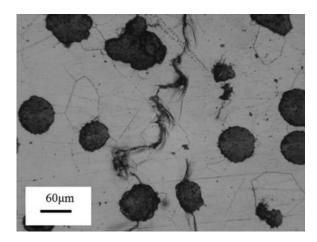


Fig. 5 Plastic deformation produced by the advancing crack.

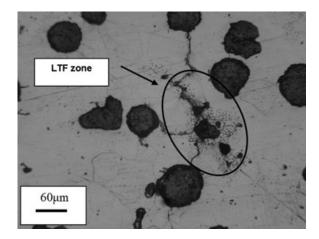


Fig. 6 Crack in progress through LTF zones in ferritic matrix.

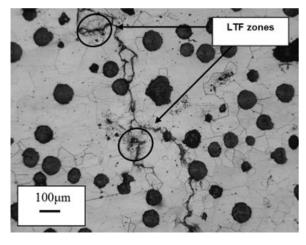


Fig. 7 Branched path of fracture in the ferritic matrix.

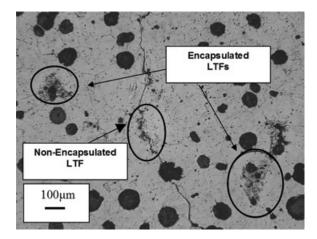


Fig. 8 Advancing crack through LTF in sample DP1.

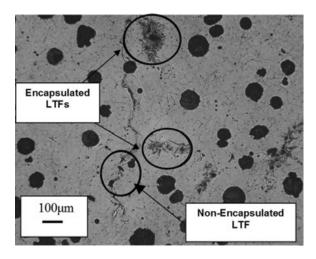
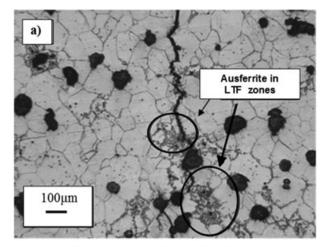


Fig. 9 Crack avoiding LTF zones encapsulated with ausferrite in sample DP1.

For samples DP1 (matrices containing approximately 90% ferrite and 10% ausferrite) the ausferrite is located mainly at the LTF zones. It is important to point out that in this case the crack propagates avoiding the ausferrite-encapsulated LTF zones (see Fig. 8 & 9). Thus, the crack follows an intricate and transgranular crack path, which exhibits zones with noticeable plastic deformation.

Samples DP2 (matrices composed by approximately 80% ferrite and 20% ausferrite) show that an increment in ausferrite content results in larger encapsulation for LTF zones and also the ausferrite starts to be present along the ferrite grain boundaries. Crack propagation is transgranular in the ferritic zones and it follows an intricate path, which avoids most of the encapsulated LTF zones (see Fig. 10).

For samples DP3 (matrices composed by 50% ferrite and 50% ausferrite) it is found that the crack advances through the ausferrite, which is now located on all the



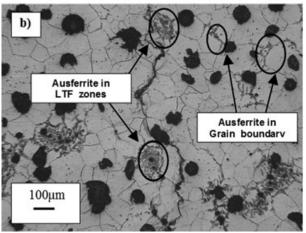
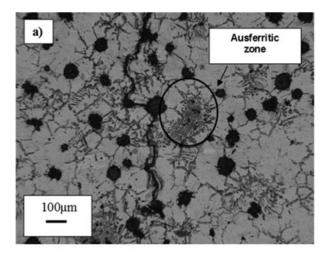


Fig. 10 Crack avoiding LTF zones encapsulated with ausferrite in sample DP2.

previous ferrite grain boundaries and LTF zones. The plastic deformation observed in these samples has been reduced and the fracture path takes a straighter geometry (see Fig. 11).

Figure 12 shows the crack propagation in samples DP4, which has an ausferritic matrix containing small amounts of allotriomorphic ferrite as a dispersed second phase. It is clearly observed that crack propagation avoids the ferrite pools; and as it happened for fully ferritic matrices, the crack chooses LTF as preferential zones to advance. The aspect of the crack in this case is straight and no signs of plastic deformation are evident in the vicinity of the fracture path.

Although the limitations because of the different propagation mechanisms, the correlation between the crack paths resulting from the numerical models and observed in the experimental analyses is evident. It is clear that the encapsulations of the LTF zones help the crack to avoid the most brittle and weakest zones into the material



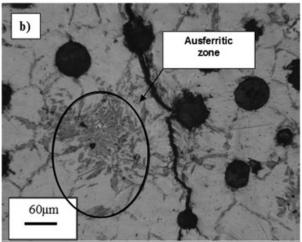


Fig. 11 Crack advancing through ausferrite in sample DP3.

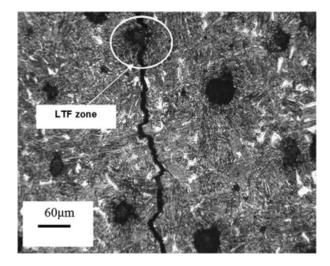


Fig. 12 Crack advancing through ausferrite and LTF zones avoiding the ferritic phase in sample DP4.

Table 7 Total number cycles until fracture as a function of the microestructure

Sample	Phases in the matrix%	Total number cycles until fracture
F	100% ferrite	792 000
DP1	90% ferrite-10% ausferrite	1818 000
DP2	80% ferrite-20% ausferrite	3528 000
DP3	50% ferrite-50% ausferrite	1764 000
DP4	5% ferrite-95% ausferrite	1872 000
ADI	100% ausferrite	702 000

microstructure, modifying the fracture path and increasing the resistance to crack propagation.

Table 7 displays the number cycles to fracture for the different test samples. Despite that these values are not the result of conventional fatigue tests it is interesting to compare their behaviour and to correlate them with the other analyses. Thus, it is interesting to observe in Table 7 that the number of cycles-to-break of the DP2 sample (80% ferrite–20% ausferrite microstructure) doubled those of the other microstructures. Note that both, the maximum time to fracture in Table 7 and the maximum in the $(K_{\rm IC}/\sigma_{0.2})^2$ values in Fig. 1 are for the specimen with the 80% ferrite–20% ausferrite microstructure.

On the other hand, the total number of cycles-to-break of the ADI sample was shorter, when compared to the sample DP4 (5% ferrite–95% ausferrite microstructure). The ductile disperse second phase (ferrite) within a harder phase (ausferrite), makes more difficult the crack growth due to changes in the propagation direction and longer fracture paths for the DP4. These results show the same trend than that of the $K_{\rm IC}$, which is higher for the DP4 specimen than for the ADI one (see Fig. 1). This improvement mechanism is called crack deflection, and it can be found in other materials such as ceramics or polymer based composites. ¹⁸

CONCLUSIONS

They have been presented in this paper numerical and experimental analysis to assess the effect of the microstructure topology on the fracture toughness of dual-phase ADI. To this end, the crack propagation behaviour was analysed numerically by means of cohesive FEA models of the material microstructure subjected to a monotonic traction load; and experimentally by performing a fractographic analysis of stable propagating cracks induced by means of cyclic loading. Both, the numerical and fractographic analyses were used to study the microstructures of a fully ferritic matrix with graphite nodules and LTF zones; dual-phase

ADI microstructures with ferritic matrices and different amounts of ausferrite encapsulating the LTF zones; and a fully ADI microstructure consisting in a completely ausferritic matrix with graphite nodules and LTF zones.

Although the limitations because of the different propagation mechanisms considered, the correlation between the crack path results computed via the numerical models and observed in the experimental analyses is evident. In fully ferritic matrices cracks grow following mainly LTF zones which present the weakest fracture toughness because they posses the highest concentrations of casting defects (such as inclusions, micro voids and segregation). On the other hand, dual-phase ADI samples with matrices containing small amounts of ausferrite encapsulating LTF zones allow changing the brittle and weak behaviour of LTF zones, which are avoided by the crack. This mechanism hinders the crack propagation due to the changes in the propagation direction and longer fracture paths. Besides, the crack path for the ADI microstructure is similar to that of the ferritic microstructure, since the transformed LTF zones are once again the regions of the model with the weakest properties. These results allow confirming the effectiveness of the ausferrite phase as a reinforcement of the ferritic matrix via the encapsulation of the brittle and weak LTF zones. The toughening mechanism is consequence of the increment in the crack path longitude as it avoids the encapsulated LTF

In addition it has been also found that the presence of small pools of allotriomorphic ferrite increase the crack propagation resistance of the ausferrite–ferrite matrices with respect to that of the completely ausferritic matrices. This improvement is attributed to the so-called crack deflection mechanism.

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APPENDIX

Cohesive element formulation

Cohesive elements were placed along the interfaces between the triangular elements in order to account for material damage and crack propagation. The progressive

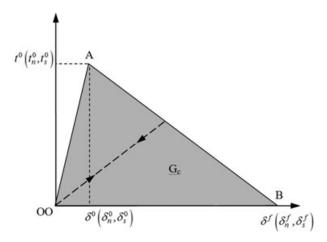


Fig. A.1 Traction separation law for the cohesive elements.

damage and failure in the cohesive elements is defined in terms of the traction-separation law depicted in Fig. A.1. Failure mechanism consists of three ingredients:

(i) Damage initiation criterion: As the name implies, damage initiation refers to the beginning of degradation of the response of a material point. In this work the damage is assumed to initiate when the following nominal stress ratios (a quadratic interaction function) reaches a value of one (Abaqus option *DAMAGE INITIATION, CRITERION = QUADS):

$$\left(\frac{t_n}{t_n^0}\right)^2 + \left(\frac{t_s}{t_s^0}\right)^2 = 1 \tag{A.1}$$

where t_n and t_s are the components of the traction vector in the directions normal and tangential to the element direction, and t_n^0 and t_s^0 are the critical values. (see Fig. A.1).

The critical traction values used for the different model regions are given in Table 3. For fragile materials (these are graphite and degraded pearlite) the normal critical traction, t_n^0 , was set equal to the material yield stress, so that no plastic deformation takes place before damage initiation. On the other hand, for ductile materials (ferritic DI and ADI) the normal critical traction was set $t_n^0 = 1.5 \sigma_y$. This choice only allows for very small scale plasticity at the initial stages of the crack growth.¹⁹ The values for the tangential critical traction were set for every material $t_s^0 = \beta^{-1} t_n^0$, where following Pandolfi et al.²⁰ the weighting coefficient $\beta = 0.866$.

The damage model assumes a linear elastic traction-separation law prior to damage initiation. In

this work initial interface stiffness (slope of the line OA in Fig. A.1) was considered very high (thousand times that of the associated material Young modulus) in order to avoid introducing an extra compliance in the model (or in other words, the critical displacements, $\delta^0(\delta_n^0, \delta_c^0)$ in Fig. A.1, were set nearly zero)

(ii) Damage evolution law: A scalar damage variable, D, represents the overall damage in the material. It initially has a value of 0 and it monotonically evolves from 0 to 1 upon further loading after the initiation of damage. The stress components of the traction-separation model are affected by the damage according to

$$t_n = \begin{cases} (1-D)\overline{t_n}, & \overline{t_n} \ge 0\\ \overline{t_n}, & \end{cases}$$

$$t_s = (1-D)\overline{t_s}$$
(A.2)

where $\overline{t_n}$ and $\overline{t_s}$ are the traction components predicted by the elastic traction-separation behavior for the current strains without damage. The second component for the definition of damage evolution is the specification of the nature of the evolution of the damage variable, D, between initiation of damage and final failure (points O and B in Fig. A.1). In this work it was defined linear softening law using the Abaqus option *DAMAGE EVOLUTION, TYPE = DISPLACE-MENT, SOFTENING = LINEAR.

(iii) Element removal criterion: The final element failure occurs when the dissipated damage energy equals the fracture energy. Note that the fracture energy equals the area under the traction-separation curve (see Fig. A.1). The dependence of the fracture energy on the mode mix is defined in this model based on a power law fracture criterion (Abaqus option *DAMAGE EVOLUTION, TYPE = ENERGY, MIXED MODE BEHAVIOR = POWER LAW, POWER = α)

$$\left(\frac{G_n}{G_n^0}\right)^{\alpha} + \left(\frac{G_s}{G_s^0}\right)^{\alpha} = 1 \tag{A.3}$$

where G_n and G_s are the work done by the traction in the normal and shear directions and G_n^0 and G_s^0 are the critical fracture energies to produce fracture in the normal (mode-I) and tangential (mode-II) directions, respectively. The exponent was set $\alpha=1$.

The critical fracture energy values for the different model regions are reported in Table 3. Mode-I critical fracture energies for every material were computed using the well-known formula $G_n^0 = K_{IC}^2/E$, while for mode-II $G_s^0 = \beta^{-2} \cdot G_n^0$ where $\beta = 0.866$ is the weighting coefficient introduced earlier.²⁰