Numerical Stability Analysis and Control of Numerical Stability Analysis and Control of

Umbilical–ROV Systems in 1-DOF Taut-Slack Condition

Mario Alberto Jordán^{*} and Jorge Luis Bustamante

Dto. de Ingeniería Eléctrica y de Computadoras, merical Stability Analysis and Control of
-ROV Systems in 1-DOF Taut-Slack Condition
Mario Alberto Jordán* and Jorge Luis Bustamante
Dto. de Ingeniería Eléctrica y de Computadoras,
Universidad Nacional del Sur
Av. Alem 125 nerical Stability Analysis and Control of
ROV Systems in 1-DOF Taut-Slack Condition
fario Alberto Jordán^{*} and Jorge Luis Bustamante
Dto. de Ingeniería Eléctrica y de Computadoras,
Universidad Nacional del Sur
Av. Alem 12 Stability Analysis and Control of

ystems in 1-DOF Taut-Slack Condition

arto Jordán* and Jorge Luis Bustamante

ngeniería Eléctrica y de Computadoras,

Universidad Nacional del Sur

1253 - (8000) Bahía Blanca, Argentina.
 merical Stability Analysis and Control of

-ROV Systems in 1-DOF Taut-Slack Condition

Mario Alberto Jordán^{*} and Jorge Luis Bustamante

Dto. de Ingeniería Eléctrica y de Computadoras,

Universidad Nacional del Sur

Av. A s in 1-DOF Taut-Slack Condition

far* and Jorge Luis Bustamante

Eléctrica y de Computadoras,

dad Nacional del Sur

8000) Bahía Blanca, Argentina.

June 5, 2006
 ABSTRACT

mumbilical-ROV system under nonlinear oscil-

d

ABSTRACT

In this paper is alternatively and computadoras,

Universidad Nacional del Sur

Av. Alem 1253 - (8000) Bahía Blanca, Argentina.

June 5, 2006

ABSTRACT

In this paper the stability of an umbilical-ROV system under nonlinea Dto, dc Ingeniería Eléctrica y de Computadoras,

Universidad Nacional del Sur

Av. Alem 1253 - (8000) Bahía Blanca, Argentina.

June 5, 2006

ABSTRACT

In this paper the stability of an umbilical-ROV system under nonlinear Universidad Nacional del Sur

Av. Alem 1253 - (8000) Bahía Blanca, Argentina.

June 5, 2006

ABSTRACT

In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using n $\rm A v.~A lcm~1253~\cdot~(8000)~Bahfa~Blanca,~Argentina.~\rm June~5,~2006$
 $\rm ABSTRACT$
 $\rm ABSTRACT$
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using numerical methods for the uncontrolle June 5, 2006
ABSTRACT
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using numerical methods for the uncontrolled
and controlled cases comparatively. Mainly **Sigerical ABSTRACT**
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using numerical methods for the uncontrolled and controlled cases comparatively. Mainly ABSTRACT
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using numerical methods for the uncontrolled
and controlled cases comparatively. Mainly the appearan In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave motion is analyzed using numerical methods for the uncontrolled and controlled cases comparatively. Mainly the appearance of the They indicate a wide diversity in qualitative bahaviours, both in the periodicity considered and controlled cases comparatively. Mainly the appearance of the so-called tautslack phenomenon on the umbilical cable produced ions in heave motion is analyzed using numerical methods for the uncontrolled

d controlled cases comparatively. Mainly the appearance of the so-called taut-

ek phenomenon on the umbilical cable produced by interactions o and controlled cases comparatively. Mainly the appearance of the s
slack phenomenon on the umbilical cable produced by interactions of m
waves and an operated the ROV is specially focused. Nonlinear eleme
sidered as nonlin

and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first and possible routes to chaos from the stability regions to outside. For detection
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first that the control system is extended and outside the stability regions a
of periodicity of the nonlinear oscillations inside and outside the stability regions a
method based on Cauchy series is developed. A first part of th the taut-slack phenomenon despite the presence of wave perturbations and the presence of the results are dedicated
to the stability of the uncontrolled dynamics. These suggest the design of a control
system that be able to method stack of a cataly of the tale of any proper. There give the results are denoted to the stability of the uncontrolled dynamics. These suggest the design of a control
system that be able to counteract hefty hauls of t as the stating of the uncontrined synthmetric and sugged the dength of a contradent departure and the stating operation under perturbation. A combination of a force and cinematic controlle based on nonlinear model-referenc Early distributed moty mand of the canted during the similarly intersect ratio and controller end on nonlinear model-reference control is proposed. Through a comparative dy of the stability regions for uncontrolled and con dynamics, nonlinear model-reference control is proposed. Through a comparative study of the stability regions for uncontrolled and controlled dynamics it is shown that the control system can extend considerably these regio enated on momintain model reactions cannot in proposal. Through a sample study of the stability regions for uncontrolled and controlled dynamics it is
that the control system can extend considerably these regions without a

I. INTRODUCTION
e operated underwater vehicles (ROVs) are widely used
sstigations and rescue operations (Fossen [8], Kijima and
he inherent nonlinear equations of motions, ROVs require I. INTRODUCTION
Tethered subsea units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Hawa I. INTRODUCTION
Tethered subsea units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Hawa I. INTRODUCTION
Tethered subsea units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Hawa 1. INTRODUCTION
Techerced subsca units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Ha Fethered subsea units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Hawary [6]). Due to 1. INTRODUCTION

I. INTRODUCTION

IT (ROVs) are widely us

in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima a

Fossen [16], El-Hawary [6]). Due to the inherent nonlinear equatio I. INTRODUCTION

I. INTRODUCTION

Also the umbilical capacity of the umbilical capacities (ROVs) are widely used

Also the umbilical cable investigations and researc operations (Fossen [8], Kijima and

sea [16], EL-Hawary Tethered subsea units such as remote operated underwater vehicles (ROVs) are widely used
in offshore engineering, scientific investigations and rescue operations (Fossen [8], Kijima and
Fossen [16], El-Hawary [6]). Due to

The top end of the cable investigations and research of the cable in the cable in the cap of the subject of the cable investigations and resear (Fosen [8], Kijima and Fosen [16], El-Hawary [6]). Due to the inherent nonline in similar engineering, elements in to the single-time responds to the seate (cosen polycins ROVs require
Fossen [16], El-Hawary [6]). Due to the inherent nonlinear equations of motions, ROVs require
the design of sophist directly along the cable and cause strumming oscillations due to vortex stress for dynamic positioning and tracking, as well as autopilot systems for automatic steering of depth and altitude (see Fossen [8], for basic det The main effect on the vehicle is that its forward speed values is dynamical distribution (see Fossen [8], for basic details).

Also the dynamics of the umbilical cable interacts with the body and the environment in a com Mothemann, and transmiss are induced parameter and anti-management of the environment in a
complex way, mainly at low and middle levels of depths, where waves and currents are significa-
tive. The top end of the cable is g Fascar (s)) as saked decadar).
Also the dynamics of the umbilical cable interacts with the body and the environment in a
plex way, mainly at low and middle levels of depths, where waves and currents are significa-
c. The t Find the umbilical cable may be significant when a complex way, mainly at low and middle levels of depths, where waves and currents are significative. The top end of the cable is generally subject to motions of the suppor

can
pear any, manny as is an anti-measure of the other or the porton and surface and surface ship-which in turn responds to the sea excitation. Additionally, strong currents may act
directly along the cable and cause stru surface ship- which in turn responds to the sea excitation. Additionally, strong currents may act
directly along the cable and cause strumming oscillations due to vortex shedding phenomena.
The main effect on the vehicle motion and the call to take control action is also and the control of the control action phenomena.
The main effect on the vehicle is that its forward speed is reduced and undesirable rotational
motions are induced (Feng a The main effect on the vehicle is that its forward speed is reduced and undesirable rotational
motions are induced (Feng and Allen [4]).
Even when vortex shedding in cables could be not so important (Indiveri [11]), a tau From the economic solution is easy to formal speak of the introduction of forces due to the six-slack
Even when vortex shedding in cables could be not so important (Indiveri [11]), a taut-slack
momenon of the umbilical ca Even when vortex shedding in cables could be not so important (Indiveri [11]), a taut-slack
phenomenon of the umbilical cable may be significant when a combination of forces due to strong
currents or thrustess and superfi to the cable which changes are the cable which changes the cable of the position of the strong
phenomenon of the umbilical cable may be significant when a combination of forces due to strong
currents or thrusters and supe

parameters or the unished value and visual constraints of the commutation of elector of alternal direction
currents or thrusters and superficial waves produces acceleration in the port/starboard direction
up to the advent deployed in a sea with weak currents, the dynamics can be simplified to a simple heave motion predictability, which makes difficult to take control actions by the operator. Commonly, high frequent and hefty oscillations o

alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou *et al.*
[18], Jordán and Beltrán-A alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou *et al.*
[18], Jordán and Beltrán-Ag

alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou *et al.*
[18], Jordán and Beltrán-Ag 2. In such situation, the rate at which the cable varies its length can affect the dynamics
the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou *et al.*,
Jordán and Beltrán-Aguedo [13]). alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglon *et al.*
[18], Jordán and Beltrán-A alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou et al.
[18], Jordán and Bellrán-Agu alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglon *et al.*
[18], Jordán and Beltrán-A alone. In such situation, the rate at which the cable varies its length can affect the dynamics
of the cable-body system, manifesting quite qualitatively different behaviors (Papazoglou *et al.*
[18], Jordán and Beltrán-A body with punctual geometry (cf. Smith [21]). Using a dimensionless different behaviors (Papazoglou *et al.* [18], Jordán and Beltrán-Aguedo [13]).
A special aspect of the operation is that the cable tension can become mu (18), Jordán and Beltrán-Aguelo [13]).

A special aspect of the operation is that the cable tension can become mull if the ascent of the

ROV occurs sufficiently fast. This can also arise for relatively large frequencies A special aspect of the operation is that the cable tension can become null if the asce
ROV occurs sufficiently fast. This can also arise for relatively large frequencies of the Huang [9], the dynamics of a cable-body sys The simplifical model of Huang, however, does not cover the nonlinear effects of the wave. In any [9], the dynamics of a cable-body system under taut-slack conditions is analyzed through is existing the nonlinear equation Hydrodynamic solutional and and the contractive response in the basis may increase in the procedure differential procedure procedure and mass process in a procedure mass in a procedure in the caller body with punctual geo

a piecewise linear equation of the cable force. Herein, the cable stiffness is assumed to have the
a piecewise linear equation of the cable force. Herein, the cable stiffness is assumed to have
blow with punctual geometry and added mass. Hydrodynamic structure the damping is considered constant and the body with punctual geometry (cf. Smith [21]). Using a dimensionless differential equation and considering an harmonic motion of the upper c Another not considered point in the Huang's model is the nonlinear density and considering an harmonic motion of the upper cable extreme, the system behavior is shown to manifest nonlinear oscillations.

The simplified mo oody with pulsival geometry (c) sharen pap), edug a anticialization direction equation and
considering an harmonic motion of the upper cable extreme, the system behavior is shown to
manifest nonlinear oscillations.
The sim In order to simplified model of Huang, however, does not cover the nonlinear effects of damping

Infest nonlinear oscillations.

In simplified model of Huang, however, does not cover the nonlinear effects of damping

idde The simplified model of Huang, however, does not cover the nonlinear effects of damping
and added mass. Hydrodynamic aspects may influence the qualitative response of the ROV
decisively (Kleczka and Kreuzer [17], Ellerman

In our contribution we aim to provide a more realistic stability enspires and added mass. Hydrodynamic aspects may influence the qualitative response of the ROV decisively (Kleezka and Kreuzer [17], Ellermann *et al.* [7] the decisively (Kleezka and Kreuzer [17], Ellermann et al. (7]). For instance, the body can radiate and hydrodynamics memory effects can influence the damping and the inertia at small depths.
Another not considered point Extrained the simulation of the figure control of the figure and hydrodynamics memory effects can influence the damping and the inertia at small depths.
Another not considered point in the Huang's model is the nonlinear dr Another not considered point in the Huang's model is the onlinear drag term in the equation
of the forced oscillator, which depends on the body shape and Reynolds number.
In order to simplify our analysis of stability and Since there does not exists an analytic solution for the nonlinear ends solution. In order to simplify our analysis of stability and controller design, we examine simple spheric more to simplify our analysis of stability a In order to simplify our analysis of stability and controller design, we examine simple spheric
forms of the ROV with different diameters providing varying drag resistance and inertia forces.
In our contribution we aim to

analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of t analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of t analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of t vertical operation can already in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of the ROV and magnitudes of thruster forces. From them, fundamental
 Lysis extracted from Poincaré maps. Regions that characterize stability are also constructed
space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
frequency, wet area of the ROV and m

analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of t analysis extracted from Poincaré maps. Regions that characterize stability are also constructed
in a space of different bifurcation parameters, namely cable length and stiffness, wave amplitude
and frequency, wet area of t the approached from Content mapper. They are that the following the state constrates the and frequency, wet area of the ROV and magnitudes of thruster forces. From them, fundamental information about the dynamic behavior o ments in the velocity of the ROV and magnificulates of thruster forces. From them, fundamental
information about the dynamic behavior of the unibilical-ROV system in taut-slack condition and
vertical operation can already Information about the dynamic behavior of the umbilical-ROV system in taut-slack condition and
vertical operation can already be obtained in physical models with different degrees of knowledge.
A second aim in this paper w system and a set of a set of bifurcation is a set of bifurcation and the state of the stability representing a control system for the sinking/lifting process. Nonlinearities that affect unfavorably the performance of the c The control and the paper will consist in designing a control system for the sinking/lifting
process. Nonlinearities that affect unfavorably the performance of the control will be compen-
sated by nonlinear feedback. The m While and the former in an entity of the formula of the control will be compen-
process. Nonlinearities that affect unfavorably the performance of the control will be compen-
sated by nonlinear feedback. The main control o In control objective will be focused on rejecting eventual
purpose thrusters for vertical displacements and adjust-
e motor unit. Another control objective will be looked at
to get it away from critical values. A basis for ments in the velocity of the hoisting crane motor unit. Another control objective will be looked at the control of the cable tension in order to get it away from critical values. A basis for the control system analysis wil

the control of the cable tension in order to get it away from critical values. A basis for the control
system analysis will be stability regions for a set of bifurcation parameters. Numerous simula-
tions has been planed i expect analysis will be stability regions for a set of bifurcation parameters. Numerons simula-
tions has been planed in the paper to illustrate the features of the control system in comparison
with the former uncontrolled From a vertical model of the ship. The cable remains of the control system in comparison
with the former uncontrolled umbilical-ROV system.
IT. DYNAMICS
Let us consider the following scenario for our study. During the sink by the action of a current as shown in Fig. (1). Depending the sinking/lifting operation of a current as shown in Fig. (1). Depending on the action of a monochromatic wave causes taut-slack transitions in the unbilical cor II. DYNAMICS
ILet us consider the following scenario for our study. During the sinking/lifting operation of a
ROV, harmonic vertical motion of the hoisting crane due to the action of a monochromatic wave
causes taut-slack II. DYNAMICS
Let us consider the following scenario for our study. During the sinking/lifting operation of a
ROV, harmonic vertical motion of the hoisting crane due to the action of a monochromatic wave
causes taut-slack t Let us consider the following scenario for our study. During the sinking/lifting operation of a ROV, harmonic vertical motion of the hoisting crane due to the action of a monochromatic wave causes taut-slack transitions i ROV, harmonic vertical motion of the hoisting craate of our staty. But the same of a reason of a reason of the section of the section of a reason of a reason of a reason of a reason in the umbilical cord. Usually the ROV i

example in this paper we analyze the phenomenon in one degree of freedom namely vertically, as for the sinking/lifting phase and assume that there exists a static equilibrium tension of the eable given by the weight of th The cable given by the weight of the vehicle surface ship.
The cable given by the vehicle in water (see Fig. (2)). Additionally we describe
the cable given by the weight of the vehicle in water (see Fig. (2)). Additionally $\begin{tabular}{lllllllllll} \multicolumn{3}{l}{{\small\textbf{S}}_{\small on}} & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet & \bullet \\ \hline \end{tabular} \begin{tabular}{lllllllllllllllllll} \multicolumn{3}{l}{{\small\textbf{C}}_{\small on}} & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet & \bullet & \bullet \\ & \bullet & \bullet & \bullet & \bullet & $\bullet$$ monochromatic excitation, complete model of the model of the model of the model of the sinking/lifting phase and assume that there exists a static equilibrium tension of the cable given by the weight of the vehicle in wat Figure 1: Remotely operated vehicle and surface ship

In this paper we analyze the phenomenon in one degree of freedom namely vertically, as for

mple in the sinking/lifting phase and assume that there exists a static equi Figure 1: Remotely operated vehicle and surface sl
In this paper we analyze the phenomenon in one degree of freedom
example in the sinking/lifting phase and assume that there exists a static
the cable given by the weight In this paper we analyze the phenomenon in one degree of freedom namely vertically, as for mple in the sinking/lifting phase and assume that there exists a static equilibrium tension of cable given by the weight of the ve mass cable is inappreciable to the mass care and assume that there exists a static equilibrium to
cable given by the weight of the vehicle in water (see Fig. (2)). Additionally we
nonlinear hydrodynamics completely, in ord rese in the slight positive buoyance interest (see Fig. (2)). Additionally we describe
condinear hydrodynamics completely, in order to show that from this simple operation with
nochromatic excitation, complex behaviors can nonlinear hydrodynamics completely, in order to show that from this simple operation with
nonlinear hydrodynamics completely, in order to show that from this simple operation with
nochromatic excitation, complex behaviors

-
-
-
-
-

relatives a
procedynamics complex behaviors can happen.

For the analysis of the dynamics of the umbilical-ROV system, following general assumptions

considered:

a) motion takes place vertically (heave mode)

b) mass cabl For the analysis of the dynamics of the umbilical-ROV system, following general assumptions

considered:

a) motion takes place vertically (heave mode)

b) mass cable is inappreciable

c) ROV has a slight positive buoyanc attached. b) mass cable is inappreciable
c) ROV has a slight positive buoyancy and its hull is spherical
c) cROV has a slight positive buoyancy and its hull is spherical
d) mass of the surface ship is very large in comparison with c) ROV has a slight positive buoyancy and its hull is spherical

d) mass of the surface ship is very large in comparison with ROV mass

e) cable has a null stiffness in slack condition

f) sea provides a persistent monochr

purposes.

Eigance 2: Umbilical-ROV system in taut condition (left) and slack condition (right)
mersion depth d with respect to the water line. When the top end of the cable moves upwards
in the quantity $b = a \sin \omega t$ and the bottom en Figure 2: Umbilical-ROV system in taut condition (left) and slack condition (right)
mersion depth *d* with respect to the water line. When the top end of the cable moves upwards
in the quantity $b = a \sin \omega t$ and the bottom e Figure 2: Umbilical-ROV system in taut condition (left) and slack condition (right)
mersion depth *d* with respect to the water line. When the top end of the cable moves upwards
in the quantity $b - a \sin \omega t$ and the bottom e sion depth d with respect to the water line. When the top end of the cable moves upwards
be quantity $b - a \sin \omega t$ and the bottom end downwards in the quantity d (cf. Fig. (2)), the
sion of the cable is
 $F_c = \frac{E A_0}{L} (d -$

$$
F_c = \frac{EA_0}{L} \left(d - L + b - a \sin(\omega t) \right),\tag{1}
$$

 $F_c = \frac{EA_0}{L} \left(d - L + b - a \sin(\omega t) \right), \eqno(1)$
 is $F_c = \frac{EA_0}{L} \left(d - L + b - a \sin(\omega t) \right), \eqno(1)$
 (1) the cable, *b* the crane jib elevation, *a* sin (ωt) the oscillation about it due
 wave, *E* the Young's modulus of the cable and is
 $F_c = \frac{EA_0}{L} (d - L + b - a \sin(\omega t)),$ (1)

the cable, *b* the crane jib elevation, *a* sin (ωt) the oscillation about it due

wave, *E* the Young's modulus of the cable and A_0 its cross section. On the

remains loose whe

$$
d - L + b - a\sin(\omega t) > 0 \to \text{taut condition } F_c > 0 \tag{2}
$$

$$
d - L + b - a\sin(\omega t) \leq 0 \to \text{slack condition } F_c = 0. \tag{3}
$$

It is noticely and the characteristic stress-deformation about 1 and L_0 is the characteristic value of the cable and A_0 its cross section. On the er side the cable conditions are summarized as
 $d - L + b - a \sin(\omega t) > 0$ be a monoclinomatic wave, L the Found is motion as the calibratized and Eq. and the calibration from the side evaluation from the side evaluation from $d - L + b - a \sin(\omega t) > 0 \rightarrow$ taut condition $F_c > 0$ (2)
 $d - L + b - a \sin(\omega t) \leq 0$ These cable conditions are summarized as
 $d - L + b - a \sin(\omega t) > 0 \rightarrow \text{taut condition } F_c > 0$
 $d - L + b - a \sin(\omega t) \leq 0 \rightarrow \text{slack condition } F_c = 0.$

It is noticing from (1) and (2)-(3) that the characteristic stress-deformation is continuous

broken at $d - L +$ These cable conditions are summarized as
 $d - L + b - a \sin(\omega t) > 0 \rightarrow \text{taut condition } F_c > 0$ (2)
 $d - L + b - a \sin(\omega t) \leq 0 \rightarrow \text{slack condition } F_c = 0.$ (3)

It is noticing from (1) and (2)-(3) that the characteristic stress deformation is continuous but

ken $\begin{array}{lcl} d-L+b-a\sin{(\omega t)} & > & 0 \rightarrow \text{tant condition } F_c > 0 \qquad \qquad & (2) \\[2mm] d-L+b-a\sin{(\omega t)} & \leq & 0 \rightarrow \text{slack condition } F_c = 0. \qquad \qquad & (3) \end{array}$ It is noticing from (1) and (2)-(3) that the characteristic stress-deformation is continuous but broken at $d-L+b=a\sin{(\omega t$

increment of physical knowledge, starting from a coarse characterization with a simple model of the hydrodynamics up to a more refined model including a velocity-depending drag coefficient and radiation-potential forces. E the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. Each increment of physical knowledge, starting from a coarse characterization with a simple model of
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. increment of physical knowledge, starting from a coarse characterization with a simple model of
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. increment of physical knowledge, starting from a coarse characterization with a simple model of
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. E increment of physical knowledge, starting from a coarse characterization with a simple model of
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. increment of physical knowledge, starting from a coarse characterization with a simple model of
the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. Exercise the hydrodynamics of the amore characterization with a simple model of hydrodynamics up to a more refined model including a velocity-depending drag coefficient radiation-potential forces. Each one of these descri

the hydrodynamics up to a more refined model including a velocity-depending drag coefficient
and radiation-potential forces. Each one of these descriptions is analyzed separately and then
comparatively under the same sett radiation-potential forces. Each one of these descriptions is analyzed separately and then
paralively under the same setting of common parameters.
Equations of motion - Model 1
c quations of motion in vertical z axis are comparatively inder the same setting of common parameters.

A. Equations of motion in vertical z axis are subject to the rigid body mechanics and to the

hydrodynamics given by Potential Flow Theory and Morison's law.

Le A. Equations of motion - Model 1
The equations of motion in vertical z axis are subject to the rigid body mechanics and to the
hydrodynamics given by Potential Flow Theory and Morison's law.
Let the hydrodynamics of the m The equations of motion in vertical z axis are subject to the rigid body mechanics and to the
hydrodynamics given by Potential Flow Theory and Morison's law.
Let the hydrodynamics of the unbilical-ROV system be described Let the hydrodynamics of the unbifold Flow Theory and Morison's law.

Let the hydrodynamics of the unbifical-ROV system be described uniquely by the so-called

led mass of the ROV geometry and by the drag force with a con Let the hydrodynamics of the umbilical-ROV system be described uniquely by the so-called added mass of the ROV geometry and by the drag force with a constant drag coefficient.
The parameters of the system are the ROV mass with a constant drag coefficient.

, the so-called added mass m_{∞} due to

e ROV surface, the gravity acceleration

coefficient C_D , the diameter of the ROV
 F_t .

cording to (2)-(3), the equations can be

for the t neters of the system are the ROV mass m , the so-called added mass m_{∞} due to the water particles in the surrounding of the ROV surface, the gravity acceleration or clensity ρ , the hydrodynamic drag force coeffic acceleration of the water particles in the surrounding of the ROV suriace, the gravity acceleration
 g , the scalar density ρ , the hydrodynamic drag force coefficient C_D , the diameter of the ROV
 D and finally the

As the cable characteristic has two linear portions according to (2)-(3), the equations can be
established separately for these two states. On one side, for the taut condition (2), it is valid

$$
(m+m_{\infty}) \text{ Red} + \frac{\pi \rho D^2}{8} C_D \text{ Red} |\text{Red}| + \frac{EA_0}{L} (d - L + b) + \frac{\pi \rho D^3}{6} g + F_t =
$$

 $m g + \frac{EA_0 a}{L} \sin(\omega t)$, (4)
and, on the other side, for the slack condition (3), it is accomplished
 $(m+m_{\infty}) \text{ Red} + \frac{\pi \rho D^2}{8} C_D \text{ Red} |\text{Red}| + \frac{\pi \rho D^3}{6} g + F_t =$
 mg . (5)
Then a solution $d(t)$ can be composed piecewise from the solutions of (4) and (5).
The approximation of the hydrodynamics through the constant coefficient m_{∞} is sufficiently
accurate for large depths. Practically, this is fulfilled for $d >> D$. For the spherical surface

$$
mg + \frac{EA_0a}{L} \sin(\omega t),
$$
\n(4)
\nand, on the other side, for the slack condition (3), it is accomplished
\n
$$
(m + m_{\infty}) \text{ Bed} + \frac{\pi \rho D^2}{8} C_D \text{ Bed} |\text{Red}| + \frac{\pi \rho D^3}{6} g + F_t =
$$
\n
$$
mg.
$$
\n(5)
\nThen a solution $d(t)$ can be composed piecewise from the solutions of (4) and (5).
\nThe approximation of the hydrodynamics through the constant coefficient m_{∞} is sufficiently
\naccurate for large depths. Practically, this is fulfilled for $d >> D$. For the spherical surface
\n9

considered, the added mass is equal to the half of the displaced fluid mass by the body, *i.e.*, $m_{\infty} = \frac{\rho \pi D^3}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is valid $\frac{m}{m_{\infty}}$ 12 . As in the case study, the ROV is assumed with a slightly positive buoyancy, it is $\frac{rD^3}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is ≥ 2 .
ations of motion - Model 2
i valid $\frac{m}{m_{\infty}} \gtrapprox 2$. $\gtrapprox 2$. considered, the added mass is equal to the half of the displaced fluid mass by the body, *i.e.*,
 $m_{\infty} = \frac{\rho \pi D^3}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is

valid $\frac{m}{m_{\in$ considered, the added mass is equal to the half of the displaced fluid mass by the body, *i.e.*,
 $m_{\infty} = \frac{\rho \pi D^3}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is

valid $\frac{m}{m_{\in$ considered, the added mass is equal to the balf of the displaced fluid mass by the body, *i.e.*,
 $m_{\infty} = \frac{\omega_{D}^{ab}}{2}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is

valid $\frac{m}{m_{\in$

considered, the added mass is equal to the half of the displaced fluid mass by the body, *i.e*
 $m_{\infty} = \frac{mD^3}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it

valid $\frac{m}{m_{\infty}} \gtrapprox$ sidered, the added mass is equal to the half of the displaced fluid mass by the body, *t.e.*,
 $= \frac{\rho n D^2}{12}$. As in the case study, the ROV is assumed with a slightly positive buoyancy, it is
 $d \frac{n_0}{m_\infty} \gtrapprox 2$.

Equ $m_{\infty} = \frac{m_{\infty}}{12}$. As in the case study, the KOV is assumed with a sugnity positive buoyancy, it is wild $\frac{m}{m_{\infty}} \gtrapprox 2$.

In addition to the added mass for the ROV geometry, we can incorporate a velocity-depende While $\frac{m}{2m\omega} \leq 2$.

B. Equations of motion - Model 2

In addition to the added mass for the ROV geometry, we can incorporate a velocity-dependent

drag coefficient for the same spherical geometry (see Figs. (3) and B. Equations of motion - Model 2

In addition to the added mass for the ROV geometry, we can incorporate a velocity-dependent

drag coefficient for the same spherical geometry (see Figs. (3) and (4)). This leads to a bett for the ROV geometry, we can incorporate a velocity-dependent
bherical geometry (see Figs. (3) and (4)). This leads to a better
mics.
 D depends basically on the shape of the ROV along the motion
will be classified as a NV geometry, we can incorporate a velocity-dependent
metry (see Figs. (3) and (4)). This leads to a better
basically on the shape of the ROV along the motion
sified as a design parameter. However, as it depends
during ope drag coefficient for the same spherical geometry (see Figs. (3) and (4)). This
description of the system dynamics.
It is worth noticing that C_D depends basically on the shape of the ROV
direction and consequently it wil spherical geometry (see Figs. (3) and (4)). This leads to a better
amics.
 C_D depends basically on the shape of the ROV along the motion

will be classified as a design parameter. However, as it depends

sich changes du and (4)). This leads to a better

be of the ROV along the motion

meter. However, as it depends

s also an operation parameter.
 l (6)

= 10^{-3} [Kg/ms] water dynamic

by calculated in the range Re \in (7) %. This leads to a better

ROV along the motion

However, as it depends

an operation parameter.

(6)

[Kg/ms] water dynamic lated in the range Re \in (7) description of the system dynamics.

It is worth noticing that C_D depends basically on the shape of the ROV along the motion direction and consequently it will be classified as a design parameter. However, as it depends by moth noticing that C_D depends basically on the shape of the ROV along the motion
and consequently it will be classified as a design parameter. However, as it depends
teynolds number, which changes during operation, i

$$
\text{Re} = \frac{\rho D}{\eta_{H_2 O}} \text{Pcd} = 1.026 \times 10^6 D \text{Pcd}
$$
\n
$$
\tag{6}
$$

with $\rho = 1.026 \times 10^3 [\text{Kg/m}^3]$ the sea water density and $\eta_{H_2O} = 10^{-3} [\text{Kg/ms}]$ water dynamic $[10^{-1}, 10^7]$ by means of a linear regression lik

$$
C_D = \varphi_{\text{Re}}^T \theta_{\text{Re}},\tag{7}
$$

with $\sum_{i=1}^{n}$

$$
\varphi_{\text{Re}}^T = [(\log_{10} \text{Re})^{21}, (\log_{10} \text{Re})^{20},
$$

..., $(\log_{10} \text{Re})^2$, $(\log_{10} \text{Re})$, 1] (8)

isity. Taken Fig. (3) into account,
$$
C_D
$$
 can be approximately calculated in the range $Re \in$

\n1, 10⁷ by means of a linear regression like

\n
$$
C_D = \varphi_{Re}^T \theta_{Re}, \qquad (7)
$$
\n2.905 × 10⁻⁹, 5.389 × 10⁻⁷, -1.592 × 10⁻⁵, 2.841 × 10⁻⁴, -3.412 × 10⁻³, 2.905 × 10⁻², -1.798 × 10⁻¹, 8.132 × 10⁻¹, -2.648, 5.925, -7.871, 2.121, 12.407, -20.641, 6.411, 17.352, -26.194, 28.856, -45.340, 62.735, -56.695, 27.193]

\n10

 $\begin{aligned} \frac{\ddot{\tilde{\xi}}}{\tilde{\xi}} & \frac{\ddot{\tilde{\xi}}}{2} \frac{\ddot{\tilde{\xi}}}{2} & \frac{\ddot{\tilde{\xi}}}{2} \frac{\ddot{\tilde{\xi}}}{2} + \frac{1}{2} \frac{\ddot{\tilde{\xi}}}{2} + \frac{1}{2$ function of Reynolds number
 $d = L + b$ + $\frac{\pi \rho D^3}{6}g + F_t =$

(a)
 $d = L + b$ + $\frac{\pi \rho D^3}{6}g + F_t =$

(9)

(a)

(4)
 $d = \frac{\pi \rho D^3}{6}g + F_t =$ 3: Drag coefficient for a spherical-shape body as function of Reynolds number
seribes a polynomial approximation of degree 21 of the curve in Fig. (3) based on
al data in steady state.
we model, equations of motion are gi Figure 3: Drag coefficient for a spherical-shape body as function of Reynolds number

Eq. (7) describes a polynomial approximation of degree 21 of the curve in Fig. (3) based on

experimental data in steady state.

In the

Eq. (7) describes a polynomial approximation of degree 21 of the curve in Fig. (3) based on
experimental data in steady state.
In the new model, equations of motion are given first for the taut condition (2) as

$$
(m+m_{\infty}) \text{ Red} + \frac{\pi \rho D^2}{8} C_D (\text{Red}) \text{ Red} |\text{Red}| + \frac{EA_0}{L} (d - L + b) + \frac{\pi \rho D^3}{6} g + F_t =
$$

 $mg + \frac{EA_0 a}{L} \sin(\omega t)$, (9)
and then for slack condition (3)
 $(m+m_{\infty}) \text{ Red} + \frac{\pi \rho D^2}{8} C_D (\text{Red}) \text{ Red} |\text{Red}| + \frac{\pi \rho D^3}{6} g + F_t =$
 mg . (10)
In Fig. (5) the drag force characteristic based on the relation $F_v = -\frac{\pi \rho D^2}{8} C_D (\text{Red}) \text{ Red} |\text{Red}|$
is described for different volumes.
C. Equations of motion - Model 3
A better characterization of the cable-ROV dynamics will include the radiation capability of the
submeresed body in motion. The radiation is significant mainly at small immersion depths. It

$$
(m+m_{\infty}) \text{ Red} + \frac{\pi \rho D^2}{8} C_D \text{ (Bed) } \text{Bed} \, | + \frac{\pi \rho D^3}{6} g + F_t =
$$

$$
m g .
$$
 (10)

In Fig. (5) the drag force characteristic based on the relation $F_v = -\frac{\pi \rho D^2}{8} C_D$ (Pcd) Pcd |Pcd|

 $m\,g+\frac{E A_0 a}{L}\sin(\omega t)\;,\eqno(9)$ and then for slack condition (3)
 $(m+m_\infty)\;\text{R}rd+\frac{\pi\rho D^2}{8}C_D\left(\text{R}rd\right)\text{R}rd\left[\text{R}rd\right]+\frac{\pi\rho D^3}{6}g+F_{\rm e}=\eqno(10)$ $\text{In Fig. (5) the drag force characteristic based on the relation }F_e=-\frac{\pi\rho B^2}{8}C_D\left(\text{R}rd\right)\text{R}cl\left[\text{R}rd\right]$ is described and then for slack condition (3)
 $(m+m_{\infty}) \text{ Br}d + \frac{\pi \rho D^2}{8} C_D \text{ (Brd)} \text{ Br}d[|\text{Brd}| + \frac{\pi \rho D^3}{6} g + F_t =$
 mg . (10)

In Fig. (5) the drag force characteristic based on the relation $F_v = -\frac{\omega \rho D^2}{8} C_D \text{ (Fed)} \text{ Br}d[\text{Fed}]$

i $(m+m_{\infty})\ \text{Red} + \frac{\pi\rho D^2}{8}C_D\left(\text{Red}\right)\ \text{Red}\left|\text{Red}\right| + \frac{\pi\rho D^3}{6}g + F_t =$ $m\ g\ . \eqno{(10)}$ In Fig. (5) the drag force characteristic based on the relation $F_r = -\frac{\pi\rho D^2}{8}C_D\left(\text{Red}\right)\ \text{Red}\left|\text{Red}\right|$ is described for different v order in Fig. (5) the drag force characteristic based on the relation $F_v = -\frac{\pi \rho D^2}{8} C_D$ (Red) Red Red is described for different volumes.
C. Equations of motion - Model 3
A better characterization of the cable-ROV dyna

$$
F_r(t) = -m_\infty \operatorname{Bcd}(t) - \int_{-\infty}^t \kappa(\tau; D, d) \operatorname{Bcd}(t - \tau) d\tau,
$$
\n(11)

0.3

0.1

0.5 1 1.5 2 2.5 3 3.5

damping of a submersed spherical body with $D = 2[m]$ and $d = 15[m]$

he dynamics is affected by a new force, namely the so-called induced-

y

F_r(t) = -m_∞ Ped(t) - $\int_{-\infty}^{t} \kappa(\tau; D, d)$ 2 2.5 3 3.5

erical body with $D = 2[m]$ and $d = 15[m]$

a new force, namely the so-called induced-
 $(\tau; D, d) \text{Red}(t - \tau) d\tau$, (11)

counting for the memory of the fluid response

e geometry of the wet part of the submersed

e Figure 6: Potential damping of a submersed spherical body with $D - 2[m]$ and $d - 15[m]$

In this situation the dynamics is affected by a new force, namely the so-called induced-

and interesponse function accounting for the Figure 6: Potential damping of a submersed spherical body with $D = 2[m]$ and $d = 15[m]$
In this situation the dynamics is affected by a new force, namely the so-called induced-
radiation force given by
 $F_r(t) = -m_{\infty} \text{Re}d(t)$ Figure 6: Potential damping of a submersed spherical body with $D = 2[m]$ and $d = 15[m]$
In this situation the dynamics is affected by a new force, namely the so-called induced-
radiation force given by $F_r(t) = -m_\infty \text{Re}d(t) = \$

$$
\kappa(\tau; D, d) = \frac{2}{\pi} \int_0^\infty \gamma(\omega; D, d) \cos(\omega t) d\omega,
$$
\n(12)

 $e = -m_{\infty} \text{Re}d(t) - \int_{-\infty}^{t} \kappa(\tau; D, d) \text{Re}d(t - \tau) d\tau,$ (11)

E-response function accounting for the memory of the fluid response

ent. It depends on the geometry of the wet part of the submersed

sion depth. For a sphere $\kappa(\tau; D, d)$ $\text{Re}d(t-\tau) d\tau$, (11)

accounting for the memory of the fluid response

the geometry of the wet part of the submersed

phere, the geometry is parametrized by *D*.

neans of the so-called damping function as
 $F_r(t) = -m_\infty \operatorname{Red}(t) - \int_{-\infty}^t \kappa(\tau; D, d) \operatorname{Red}(t-\tau) \, d\tau, \tag{11}$
where $\kappa(\tau; D, d)$ is an impulse-response function accounting for the memory of the fluid response
to a sudden body displacement. It depends on the geometry o where $\kappa(\tau; D, d)$ is an impulse-response function accounting for the memory of the fluid response
to a sudden body displacement. It depends on the geometry of the wet part of the submersed
body as well as on the immersio where $\kappa(\tau; D, d)$ is an impulse-response function accounting for the memory of the fund response
to a sudden body displacement. It depends on the geometry of the wet part of the submersed
body as well as on the immersion to a sudden body displacement. It depends on the geometry of the wet part of the submersted
body as well as on the immersion depth. For a sphere, the geometry is parametrized by *D*.
A straightforward form to calculate $\$ for the submersed
netrized by *D*.
(12)
t can be calculated
ltrán-Aguedo [14]).
body for particular
for hydrodynamics body as well as on the immersion depth. For a sphere, the geometry is parametrized by D

A straightforward form to calculate κ is by means of the so-called damping function as
 $\kappa(\tau; D, d) = \frac{2}{\pi} \int_0^\infty \gamma(\omega; D, d) \cos$

$$
\gamma(\omega; D, d) = f(D, d)\,\gamma(g(d)\omega; 2, 15),\tag{13}
$$

$$
f(D,d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+7.05)^6}
$$
 (14)

$$
g(d) = \frac{8.28}{(d+2.51)^{0.73}},\tag{15}
$$

 $\frac{6}{6}$ $\frac{6}{6}$ $\frac{1}{7}$ $\frac{6}{8}$ $\frac{1}{10}$ $\frac{1}{10}$ $\frac{3}{10}$ $\frac{4}{10}$ $\frac{5}{10}$ $\frac{6}{10}$

e responses for different diameters *D* and depths *d*

2) and *d* can be approximated for $d > D/2$ by
 $(\omega; D, d) = f(D$ t diameters D and depths d

t diameters D and depths d

imated for $d > D/2$ by
 $g(d)\omega; 2, 15$, (13)
 $D^{4.58}$
 $\frac{D^{4.58}}{(d+7.05)^6}$ (14)
 $\frac{D^{6.78}}{(d+7.05)^6}$ (15)

terpolating various curves $\gamma(\omega; D, d)$ for a

t to sponses for different diameters D and depths d

and d can be approximated for $d > D/2$ by
 $D, d) = f(D, d) \gamma(g(d)\omega; 2, 15),$ (13)

iunctions
 $D, d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d + 7.05)^6}$ (14)
 $g(d) = \frac{8.28}{(d + 2.51)^{0.73}},$ (15)
 or different diameters *D* and depths *d*

(D, d) $\gamma(g(d)\omega; 2, 15)$, (13)

(D, d) $\gamma(g(d)\omega; 2, 15)$, (13)

4.8 × 10⁶ $\frac{D^{4.58}}{(d+7.05)^6}$ (14)

8.28

($d+2.51$)^{0.73}</sub>, (15)

and by interpolating various curves $\gamma(\omega;$

Figure 7: Impulse responses for different diameters D and depths d

The dependency of γ with D and d can be approximated for $d > D/2$ by
 $\gamma(\omega; D, d) = f(D, d) \gamma(g(d)\omega; 2, 15),$ (13)

with attenuation and contraction fu The dependency of γ with D and d can be approximated for $d > D/2$ by
 $\gamma(\omega; D, d) = f(D, d) \gamma(g(d)\omega; 2, 15)$, (13)

with attenuation and contraction functions
 $f(D, d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+7.05)^6}$ (14)
 $g(d) = \frac{8.28}{(d+$ $\gamma(\omega; D, d) - f(D, d) \gamma(g(d)\omega; 2, 15)$, (13)

in attenuation and contraction functions
 $f(D, d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+2.51)^{0.79}}$, (14)
 $g(d) = \frac{8.28}{(d+2.51)^{0.79}}$, (15)

bectively. Relations (14)-(15) were obtained by inter with attenuation and contraction functions
 $f(D, d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+7.05)^6}$ (14)
 $g(d) = \frac{8.28}{(d+2.51)^{0.73}}$, (15)

respectively. Relations (14)-(15) were obtained by interpolating various curves $\gamma(\omega; D, d)$ for with attenuation and contraction functions
 $f(D,d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+7.05)^6}$ (14)
 $g(d) = \frac{8.28}{(d+2.51)^{0.73}}$, (15)

respectively. Relations (14)-(15) were obtained by interpolating various curves $\gamma(\omega; D, d)$ for $f(D,d) = 4.8 \times 10^6 \frac{D^{4.58}}{(d+7.05)^6}$ (14)
 $g(d) = \frac{8.28}{(d+2.51)^{0.73}}$, (15)

respectively. Relations (14)-(15) were obtained by interpolating various curves $\gamma(x; D, d)$ for a

set of values of D and d and normalizing w $g(d) = \frac{8.28}{(d+2.51)^{0.73}}$,
respectively. Relations (14)-(15) were obtained by interpolating various curves $\gamma(\omega; D)$
set of values of D and d and normalizing with respect to $\gamma(\omega; 2, 15)$.
Using Fig. (6) and putting E bestively. Relations (14)-(15) were obtained by interpolating various curves $\gamma(\omega; D, d)$ for a
of values of D and d and normalizing with respect to $\gamma(\omega; 2, 15)$.
Using Fig. (6) and putting Eq. (13) into Eq. (12) the imp evaluate the induced-radiation force F_r for the sinking/lifting values can be interpretently induced for a set of values of D and d and normalizing with respect to $\gamma(\omega)$; 2, 15).

Using Fig. (6) and putting Eq. (13) i Using Fig. (6) and putting Eq. (13) into Eq. (12) the impulse response function is numerically
Using Fig. (6) and putting Eq. (13) into Eq. (12) the impulse response function is numerically
nd. Fig. (7) shows different im -response functions for a set of values of D and d . One
se for low depths and relatively large diameters. Other
olution of the response with a resonance frequency that
oody diameter and depth, (11) can be applied so a

$$
\kappa(\tau, t) = \kappa(\tau, d(t))\tag{16}
$$

Thus the equations of motion in vertical z axis become first for the taut condition
\n
$$
[m+m_{\infty}] \text{ Red} + \frac{\pi \rho D^2}{8} C_D(\text{Red}) \text{ Red} |\text{Red}| + \frac{EA_0}{L} (d - L + b) + \frac{\pi \rho D^3}{6} g + F_t =
$$
\n
$$
m g + \frac{EA_0 a}{L} \sin(\omega t) - \int_{-\infty}^t \kappa(\tau, t) \text{Red}(t - \tau) d\tau, \qquad (17)
$$
\nand for the slack condition
\n
$$
[m+m_{\infty}] \text{ Red} + \frac{\pi \rho D^2}{8} C_D(\text{Red}) \text{ Red} |\text{Red}| + \frac{\pi \rho D^3}{6} g + F_t =
$$
\n
$$
m g - \int_{-\infty}^t \kappa(\tau, t) \text{ Red}(t - \tau) d\tau. \qquad (18)
$$
\nIt is noticing that
\n
$$
f_o = - \int_{-\infty}^0 \kappa(\tau, 0) \text{ Red}(t - \tau) d\tau, \qquad (19)
$$
\ndescribes the effect of the past evolution of the hydrodynamics at $t = 0$, *i.e.*, it describes the

$$
[m+m_{\infty}] \operatorname{Red} + \frac{\pi \rho D^2}{8} C_D(\operatorname{Red}) \operatorname{Red} |\operatorname{Red}| + \frac{\pi \rho D^3}{6} g + F_t =
$$

$$
m g - \int_{-\infty}^t \kappa(\tau, t) \operatorname{Red}(t - \tau) d\tau .
$$
 (18)

$$
f_o = -\int_{-\infty}^{0} \kappa(\tau, 0) \text{P}cd(t - \tau) d\tau,
$$
\n(19)

is the equations of motion in vertical z axis become first for the taut condition
 $[m+m_{\infty}] \text{ Red} + \frac{\pi \rho D^2}{8} C_D(\text{Red}) \text{ Red} |\text{Red}| + \frac{EA_0}{L} (d - L + b) + \frac{\pi \rho D^3}{6} g + F_t$
 $mg + \frac{EA_0 a}{L} \sin(\omega t) - \int_{-\infty}^t \kappa(\tau, t) \text{Red}(t - \tau) d\tau$,

for $mg + \frac{E\Delta_0 a}{L} \sin\left(\omega t\right) = \int_{-\infty}^{\infty} \kappa(\tau,t) \operatorname{Red}(t-\tau) \, d\tau \; , \eqno{(17)}$

and for the slack condition $[m+m_{\infty}] \det + \frac{\pi \rho D^2}{8} C_D(\operatorname{bed}|\operatorname{Red}|\operatorname{Red}| + \frac{\pi \rho D^3}{6} g + F_t =$
 $m_g - \int_{-\infty}^t \kappa(\tau,t) \operatorname{Red}(t-\tau) \, d\tau \; . \eqno{(18)}$

It is no and for the slack condition
 $[m+m_{\infty}]$ ieal $+\frac{\pi\rho D^2}{8}C_D(\text{bcd})$ ieal $|\text{acd}| + \frac{\pi\rho D^3}{6}g + F_t =$
 $mg - \int_{-\infty}^t \kappa(\tau,t)\text{Red}(t-\tau)\,d\tau$. (18)

It is noticing that
 $f_o = -\int_{-\infty}^0 \kappa(\tau,0)\text{Red}(t-\tau)\,d\tau$, (19)

describes the effec $[m+m_{\infty}] \text{ red} + \frac{\pi \rho D^2}{8} C_D(\text{red}) \text{ red} |\text{red}| + \frac{\pi \rho D^3}{6} g + F_t =$
 $m g - \int_{-\infty}^t \kappa(\tau, t) \text{med}(t - \tau) d\tau. \tag{18}$ noticing that
 $f_o = - \int_{-\infty}^0 \kappa(\tau, 0) \text{med}(t - \tau) d\tau, \tag{19}$ $s \text{ the effect of the past evolution of the hydrodynamics at } t = 0, i.e., \text{ it describes the
condition for the differential equations (17)-(18). Fortunately, the existence of $\$$ $[m+m_{\infty}]$ if $\det + \frac{c_{\infty}}{8}C_D(\text{bcd})$ if $\det + \frac{c_{\infty}}{6}g + F_k =$
 $mg - \int_{-\infty}^6 \kappa(\tau, t) \text{P}d(t - \tau) d\tau$. (18)

It is noticing that
 $f_0 = -\int_{-\infty}^0 \kappa(\tau, 0) \text{P}d(t - \tau) d\tau$, (19)

describes the effect of the past evolution of $mg = \int_{-\infty}^{\infty} \kappa(\tau, t) \cdot \mathbf{B} \cdot d(t - \tau) d\tau$. (18)

It is noticing that
 $f_o = -\int_{-\infty}^0 \kappa(\tau, 0) \cdot \mathbf{B} \cdot d(t - \tau) d\tau$, (19)

describes the effect of the past evolution of the hydrodynamics at $t = 0$, *i.e.*, it describes th $=-\int_{-\infty}^{0} \kappa(\tau,0) \text{ } \text{bed}(t-\tau) \, d\tau,$ (19)

olution of the hydrodynamics at $t=0$, *i.e.*, it describes the

1 equations (17)-(18). Fortunately, the evanescence of $\kappa(\tau)$ for

m (17)-(18) indicate that the fact of su describes the effect of the past evolution of the hydrodynamics at $t = 0$, *i.e.*, it describes the initial condition for the differential equations (17)-(18). Fortunately, the evanescence of $\kappa(\tau)$ for $\tau \to \infty$ and th initial condition for the differential equations (17)-(18). Fortunately, the evanescence of $\kappa(\tau)$ for $\tau \to \infty$ and the passivity of system (17)-(18) indicate that the fact of supposing $f_v = 0$ has no effect in the acc

$$
Pcd + f(Pcd, d, \mu_i) = h(u, \mu_j),\tag{20}
$$

en (17)-(18) indicate that the fact of supposing $f_o = 0$ has no
ion $d(t)$ at steady state (see Jordán [12]). So, for the following
red null.
III. STABILITY ANALYSIS
al solution for the different nonlinear equations (4)-(5 effect in the accuracy of the solution d(t) at steady state (see Jordan [12]). So, for the following
studies in steady state it is assumed mill.
III. STABILITY ANALYSIS
An attempt to obtain an analytical solution for the studies in steady state it is assumed mul.

III. STABILTTY ANALYSIS

An attempt to obtain an analytical solution for the different nonlinear equations (4)-(5), (9)-(10)

and (17)-(18) generally fails. The motion equations III. STABILITY ANALYSIS

An attempt to obtain an analytical solution for the different nonlinear equations (4)-(5), (9)-(10)

and (17)-(18) generally fails. The motion equations can be put generically as
 $\text{Pcd} + f(\text{Pcd}, d$ An attempt to obtain an analytical solution for the different nonlinear equations (4)-(5), (9)-(10)
and (17)-(18) generally fails. The motion equations can be put generically as
 $\text{Red} + f(\text{Red}, d, \mu_i) = h(u, \mu_j)$, (20)
with $u =$

for period-one solutions and approximated methods of solution are discussed for instance in
Rossenwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability of fo for period-one solutions and approximated methods of solution are discussed for instance in
Rossenwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability of fo period-one solutions and approximated methods of solution are discussed for instance in
senwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability of forced p for period-one solutions and approximated methods of solution are discussed for instance in Rossenwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability of fo for period-one solutions and approximated methods of solution are discussed for instance in Rossenwasser [20], Guckenheimer and Holmes [5].

In Huang [9] was established an analytical procedure for detecting stability of for period-one solutions and approximated methods of solution are discussed for instance in Rossenwasser [20], Guckenheimer and Holmes [5].

In Huang [9] was established an analytical procedure for detecting stability of f for period-one solutions and approximated methods of solution are discussed for instance in Rossenwasser [20], Guckenheimer and Holmes [5].

In Huang [9] was established an analytical procedure for detecting stability of for period-one solutions and approximated methods of solution are discussed
Rossenwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability
one stable orbits ba Bernord can be betalled and approximated intended of solidation are distended for intended in
Schwasser [20], Guckenheimer and Holmes [5].
In Huang [9] was established an analytical procedure for detecting stability of fo In Huang [9] was established an analytical procedure for detecting stability of forced period-
one stable orbits based on the observation of eigenvalues of a discrete system that relates cross
points through zero of perio one stable orbits based on the observation of eigenvalues of a discrete system that
points through zero of periodic orbits. The method is complemented with an itera
for enhancing the information given by the eigenvalues a points through zero of periodic orbits. The method is complemented with an iterative algorithm
for enhancing the information given by the eigenvalues about stability. The domain of attraction
is extremely sensible to bad

for enhancing the information given by the eigenvalues about stability. The domain of attraction
is extremely sensible to bad initial conditions, so that the result is not always reliable to be
extended here.
In this pape and its state trajectory sampled at a rate $T = \frac{2\pi}{\omega}$. The resulting time-discrete dynamics is dethe about stability. The domain of attraction

o that the result is not always reliable to be

in order to establish stability. These are based

measures (Guckenheimer and Holmes [5]).

stem starts from an initial condi is extremely sensible to bad initial conditions, so that the result is not a
extended here.
In this paper we develop numerical procedures in order to establish stabil
on Poincaré maps, time averaging and asymptotic measur merical procedures in order to establish stability. These are based

ng and asymptotic measures (Guckenheimer and Holmes [5]).

e umbilical-ROV system starts from an initial condition $(d(0), \text{Re}d(0))$

d at a rate $T = \frac{2\$

$$
\begin{bmatrix} d(k+1) \\ \mathrm{Pcd}(k+1) \end{bmatrix} = \mathbf{F} \left(\begin{bmatrix} d(k) \\ \mathrm{Pcd}(k) \end{bmatrix}, \boldsymbol{\mu} \right). \tag{21}
$$

merical procedures in order to establish stability. These are based

img and asymptotic measures (Guckenheimer and Holmes [5]).

he umbilical-ROV system starts from an initial condition $(d(0), \text{Red}(0))$

ed at a rate $T = \frac{2$ on Poincaré maps, time averaging and asymptotic measures (Guckenheimer and Holmes [5]).

A. Periodic solutions

Let us assume the behavior of the umbilical-ROV system starts from an initial condition $(d(0), \text{Re}(0))$

and it A. Periodic solutions

Let us assume the behavior of the umbilical-ROV system starts from an initial condition $(d(0), \text{B-}d(0))$

and its state trajectory sampled at a rate $T = \frac{2\pi}{\sigma}$. The resulting time-discrete dynam Let us assume the behavior of the umbilical-ROV system starts from an initial condition $(d(0), \text{Re}d(0))$
and its state trajectory sampled at a rate $T - \frac{2\pi}{\omega}$. The resulting time-discrete dynamics is de-
scribed by
 \begin Let us assume the behavior of the umbilical-ROV system starts from an initial condition $(d(0), 8rd(0))$
and its state trajectory sampled at a rate $T = \frac{2\pi}{\omega}$. The resulting time-discrete dynamics is de-
scribed by
 $\begin{pmatrix$ and its state trajectory sampled at a rate $T = \frac{2\pi}{\omega}$. The resulting time-discrete dynamics is described by

scribed by
 $\begin{bmatrix}\n d(k+1) \\
 \text{Re}d(k+1)\n \end{bmatrix} = \mathbf{F}\left(\begin{bmatrix}\n d(k) \\
 \text{Re}d(k)\n \end{bmatrix}, \boldsymbol{\mu}\right)$. (21)

with k a posit bed by
 $\begin{bmatrix}\nd(k+1) \\
\text{Re}d(k+1)\n\end{bmatrix} = \mathbf{F}\left(\begin{bmatrix}d(k)\\ \text{Re}d(k)\n\end{bmatrix}, \boldsymbol{\mu}\right).$ (21)
 $\text{h } k$ a positive integer, \mathbf{F} a nonlinear vector-valued function that is smooth in both regions

mited by $d - L + b - a\sin(\omega t) > 0$ a with k a positive integer, **F** a nonlinear vector-valued function that is smooth in both regions delimited by $d - L + b - a \sin(\omega t) > 0$ and $d - L + b - a \sin(\omega t) \le 0$, and μ a vector that describes the control parameters for bifurca with k a positive integer, **F** a nonlinear vector-valued function that is smooth in both regions
delimited by $d - L + b - a \sin(\omega t) > 0$ and $d - L + b - a \sin(\omega t) \le 0$, and μ a vector that
describes the control parameters for bifurca

instead.

B. Identification of periodic solutions

Periodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

point of the discrete system (21) described in the Poincaré map.

So for a particul

B. Identification of periodic solutions

Periodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

point of the discrete system (21) described in the Poincaré map.

So for a particul B. Identification of periodic solutions

Periodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

point of the discrete system (21) described in the Poincaré map.

So for a particul Identification of periodic solutions

iodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

ant of the discrete system (21) described in the Poincaré map.

So for a particular value From an arbitrary initial conditions

From an arbitrary initial conditions (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

point of the discrete system (21) described in the Poincaré map.

So for a particular value B. Identification of periodic solutions

Periodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18)

point of the discrete system (21) described in the Poincaré map.

So for a particular valued μ , there Identification of periodic solutions

iodic orbits of the continuous systems (4)-(5), (9)-(10) and (17)-(18) correspond to a fixed

at of the discrete system (21) described in the Poincaré map.

So for a particular valued solutions

(a)-(5), (9)-(10) and (17)-(18) correspond to a fixed

21) described in the Poincaré map.
 μ , there exists a solution $d(t)$ and a state trajectory that starts

ition $\zeta(0)$ in an attraction domain and is a point of the discrete system (21) described in the Poincaré map.

So for a particular valued μ , there exists a solution $d(t)$ and a state trajectory that starts

from an arbitrary initial condition $\zeta(0)$ in an attra So for a particular valued μ , there exists a solution $d(t)$ and a state trajectory that starts
from an arbitrary initial condition $\zeta(0)$ in an attraction domain and is asymptotically periodic
with period $nT = \frac{u_0^$

$$
\zeta(k) = \begin{bmatrix} \zeta_1(k) \\ \zeta_2(k) \end{bmatrix} = \begin{bmatrix} d(t_0 + kT) \\ \text{Pcd}(t_0 + kT) \end{bmatrix},
$$
\n(22)

from an arbitrary initial condition $\zeta(0)$ in an attraction domain and is asymptotically periodic
with period $nT = \frac{n2\pi}{\omega}$.
Considering the sampled trajectory conformed as
 $\zeta(k) = \begin{bmatrix} \zeta_1(k) \\ \zeta_2(k) \end{bmatrix} = \begin{bmatrix} d(t_$ In period $nT = \frac{u_{2n}}{\omega}$.

Considering the sampled trajectory conformed as
 $\zeta(k) = \begin{bmatrix} \zeta_1(k) \\ \zeta_2(k) \end{bmatrix} = \begin{bmatrix} d(t_0 + kT) \\ \text{Red}(t_0 + kT) \end{bmatrix}$, (22)

say the system is asymptotically stable and has a fixed point, whe Considering the sampled trajectory conformed as
 $\zeta(k) = \begin{bmatrix} \zeta_1(k) \\ \zeta_2(k) \end{bmatrix} = \begin{bmatrix} d(t_0 + kT) \\ w d(t_0 + kT) \end{bmatrix}$, (22)

we say the system is asymptotically stable and has a fixed point, when the series $\{\zeta(k)\}_{k=0}^{\infty$ $\zeta(k) = \begin{bmatrix} \zeta_1(k) \\ \zeta_2(k) \end{bmatrix} = \begin{bmatrix} d(t_0 + kT) \\ \text{ed}(t_0 + kT) \end{bmatrix}$, (22)
we say the system is asymptotically stable and has a fixed point, when the series $\{\zeta(k)\}_{k=0}^{\infty}$
converges to a periodic series. Moreover, th

fulfilled the contract of the

fulfilled
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - n + 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - n) \right\| < \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - n) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - n - 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - 2n + 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - 2n + 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - 2n - 1) \right\| > \varepsilon
$$
\n
$$
\left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - 2n - 1) \right\| > \varepsilon
$$
\n
$$
\vdots
$$
\nand the series of the previous system within the band ε will also accomplish\n
$$
\lim_{k \to \infty} \left\| \mathbf{f}(\mathbf{k}) - \mathbf{f}(\mathbf{k} - v n) \right\| = 0,
$$
\n(24)\nfor $v = 0, \pm 1, \pm 2, \ldots$ Moreover (23) and (24) are independent on t_0 except for a set with measure zero of series that are identically zero or constant.\nEquation system (23) is equivalently to the autocorrelation function $\vartheta(\tau) - \sum_{k=0}^{\infty} \mathbf{f}(k) \mathbf{f}^T(k + \tau),$ for $\tau = 0, 1, \ldots$. The calculus of $\vartheta(\tau)$ for large τ will reveal the periodicity of ζ . The detection method developed above can also be used to identify a chaotic state. In this case, there does not exist any finite

$$
\lim_{k \to \infty} \|\zeta(k) - \zeta(k - v n)\| = 0,\tag{24}
$$

Equation system (23) is equivalently to the autocorrelation function $\vartheta(\tau) = \sum_{k=0}^{\infty} \zeta(k) \zeta^{T} (k +$ $(k+$

 $\label{eq:2.1} \|{\mathbf{c}}(k)-{\mathbf{c}}(k-2n-1)\|>\varepsilon.$
 $\label{eq:2.1} \|{\mathbf{c}}(k)-{\mathbf{c}}(k-q)\|\varepsilon.$
 the series of the previous system within the band ε
will also accomplish $\lim_{k\to\infty}\|{\mathbf{c}}(k)-{\mathbf{c}}(k-vn)\|=0, \qquad \qquad (24)$
 $v=0,\pm 1,\pm 2,... \text{$:
 $|\zeta(k) - \zeta(k-q)| > \varepsilon.$

and the series of the previous system within the band ε will also accomplish
 $\lim_{k \to \infty} |\zeta(k) - \zeta(k - v n)| = 0,$ (24)
 $v v = 0, \pm 1, \pm 2, ...$ Moreover (23) and (24) are independent on t_0 except fo and the series of the previous system within the band ε will also accomplish
 $\lim_{k \to \infty} ||\zeta(k) - \zeta(k - v n)|| = 0,$ (24)

for $v = 0, \pm 1, \pm 2, ...$ Moreover (23) and (21) are independent on t_0 except for a set with measure
 $\lim_{k \to \infty} \|\zeta(k) - \zeta(k - v\,n)\| = 0, \tag{24}$ for $v = 0, \pm 1, \pm 2, \ldots$ Moreover (23) and (24) are independent on t_0 except for a set with measure zero of series that are identically zero or constant.

Equation system (23) i (23) and (24) are independent on t_0 except for a set with measure
lly zero or constant.
livalently to the autocorrelation function $\vartheta(\tau) = \sum_{k=0}^{\infty} \zeta(k)\zeta^T(k+\tau)$
v of peaks of $\vartheta(\tau)$ for large τ will reveal th zero of series that are identically zero or constant.

Equation system (23) is equivalently to the autocorrelation function $\vartheta(\tau) = \sum_{k=0}^{\infty} \zeta(k) \zeta^{T}(k + \tau)$, for $\tau = 0, 1, ...$ The cadency of peaks of $\vartheta(\tau)$ for larg Equation system (23) is equivalently to the autocorrelation function $v(\tau) = \sum_{k=0}^{\infty} \zeta(k)\zeta^k(k + k)$, for $\tau = 0, 1, \dots$. The cadency of peaks of $\vartheta(\tau)$ for large τ will reveal the periodicity of ζ .
The detection

$$
\lim_{q \to \infty} \|\zeta(k) - \zeta(k - j)\| > 0, \text{ with } j = 0, ..., q,
$$
\n(25)

 $\begin{bmatrix}\n\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0}\n\end{bmatrix}$
 $\begin{bmatrix}\n\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0}\n\end{bmatrix}$
 $\begin{bmatrix}\n\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \mathbf{0}\n\end{bmatrix}$
 $\begin{bmatrix}\n\mathbf{0} & \mathbf{0} \\
\mathbf{0} & \$ **Example 12** \overline{t} **F** \overline{t} **12** \overline{t} **13** \overline{t} **13** \overline{t} **13** \overline{t} **13** \overline{t} **140 460 480 480 580 540 560 580 580 680 150 150 150 140 440 460 480 480 480 6** ^{-0.31}
 $\frac{1}{440-460-460}$
 $\frac{1}{440-460-460}$

Figure 8: Period-4 behaviour of the ROV dynamics

It is observing that the detection method proposed above performs well in the transition of

the transitory to the stead -6.5

Figure 8: Period-4 behaviour of the ROV dynamics

Figure 8: Period-4 behaviour of the ROV dynamics

Figure 8: Period-4 behaviour of the ROV dynamics

It is observing that the detection method proposed above perfor **1.** and $\frac{4}{40}$ and $\frac{4}{400}$ and $\frac{4}{500}$ so $\frac{520}{500}$ so $\frac{560}{500}$ so $\frac{40}{500}$ so $\frac{1}{500}$ so $\frac{1}{500}$ so $\frac{1}{500}$ so $\frac{1}{500}$ so $\frac{1}{500}$ the ROV dynamics ing that the detection meth Figure 8: Period-4 behaviour of the ROV dynamics

It is observing that the detection method proposed above performs well in the transition of

transitory to the steady state. The application of the identification method i Eignie 8: Period-4 behaviour of the ROV dynamics

It is observing that the detection method proposed above performs well in the transition of

the transitory to the steady state. The application of the identification meth Figure 8: Period-4 behaviour of the ROV dynamics

It is observing that the detection method proposed above performs well in the transition of

transitory to the steady state. The application of the identification method i wing that the detection method proposed above performs well in the t

to the steady state. The application of the identification method is

or a P4 behavior of the cable-ROV system with free parameter: D
 ${}^6[N]$ and $C_D =$ tion method proposed above performs well in the transition of
e. The application of the identification method is illustrated
f the cable-ROV system with free parameter: $D = 0.85[m]$,
whose time evolution is depicted in Fig

$$
\varepsilon = 10^{-6} \max_{t \in [0,\infty], \tau \in (0,\infty]} |\zeta(t) - \zeta(t-\tau)| = 10^{-6} \left(\max_{t \in [0,\infty]} \zeta(t) - \min_{t \in [0,\infty]} \zeta(t) \right),\tag{26}
$$

the transitory to the steady state. The application of the identification method is illustrated
in Fig. (9) for a P4 behavior of the cable-ROV system with free parameter: $D = 0.85$ [m],
 $E A_0 = 5 \times 10^6$ [N] and $C_D = 0.2$, w in Fig. (9) for a 14 behavior of the cable-ROV system with free parameter: $D = 0.88$ [m],
 $EA_0 = 5 \times 10^6$ [N] and $C_D = 0.2$, whose time evolution is depicted in Fig. (8).

The detection of this period is performed on the s $E A_0 = 5 \times 10^{\circ}$ [N] and $U_D = 0.2$, whose time evolution is depicted in Fig. (8).

The detection of this period is performed on the sampled series on $\zeta_1(t)$ of Fig. (8) at a rate equal to the wave period $T = 6.5$ [s].
 The detection of this period is performed on the sampled series on $\zeta_1(t)$ of Fig. (8) at a rate equal to the wave period $T = 6.5[s]$.

According to the restrictions (23) and for a tolerance selected as
 $\varepsilon = 10^{-6} \sum_{t \$

In this paper a stability region is defined as a zone in the free-parameter space, in which the
Bigger a stability region is defined as a zone in the free-parameter space, in which the
Bigger a stability region is defined **Example 1.1 behavior of the umbilical-ROV** system is defined as a zone in the free-parameter space, in which the universe state of the umbilical-ROV system is characterized by a bounded oscillation in steady state
subj Subsection of bifurcations for a P4-case upon Cauchy series

Figure 9: Detection of bifurcations for a P4-case upon Cauchy series

IV. STABILITY REGIONS

ID. STABILITY REGIONS

ID. STABILITY REGIONS

ID. STABILITY REGIONS **PRECISION CONTROLLED AND THE SET OF STARTLE SET ON AN ABOVE AT A SIMULTY REGISTARY PROTONS**
 PRECISION SAFELL TYPE ROTONS
 PRECISION STABILITY REGIONS
 PRECISION STARTLEY REGIONS
 PRECISION SUMPLE AND SUMPLE AND Figure 9: Detection of bifurcations for a P4-case upon Canchy series
IV. STABILITY REGIONS
In the space, in which the
stability region is defined as a zone in the free-parameter space, in which the
avior of the umblical-R which the
eady state
naracterize
, which is
arameters,
d stability Figure 9: Detection of bifurcations for a P4-case upon Cauchy series
IV. STABILITY REGIONS
In this paper a stability region is defined as a zone in the free-parameter space, in
behavior of the umbilical-ROV system is char in the set of bifurcations for a P4-case upon Cauchy series

IV. STABILITY REGIONS

is defined as a zone in the free-parameter space, in which the

system is characterized by a bounded oscillation in steady state
 > 0 . IV. STABILITY REGIONS

In this paper a stability region is defined as a zone in the free-parameter space, in which the

behavior of the umbilical-ROV system is characterized by a bounded oscillation in steady state

subje In this paper a stability region is defined as a zone in the free-parameter space, in which the behavior of the umbilical-ROV system is characterized by a bounded oscillation in steady state subject to the taut condition

behavior of the umbilical-ROV system is characterized by a bounded oscillation
subject to the taut condition $F_c > 0$. From a practical point of view, such repredictable and safe ROV operations.
The boundary of a stability subject to the taut condition $r_c > 0$. From a practical point of view, such regions characterize
predictable and safe ROV operations.
The boundary of a stability region depends on the initial vector $(d(0), \text{P}d(0))^T$, whic predictable and sale ROV operations.

The boundary of a stability region depends on the initial vector $(d(0), \text{Re}d(0))^T$, which is

assumed equal to $(L(0) - b, 0)^T$ in the set of experiments. For specific values of free para The boundary of a stability region depends on the initial vector $(d(0), \text{t}cd(0))$
assumed equal to $(L(0) - b, 0)^T$ in the set of experiments. For specific values of free
the dynamics can also bifurcate showing high period os

med equal to $(L(0) - b, 0)$ in the set of experiments. For specific values of tree parameters,
dynamics can also bifurcate showing high period oscillations or even chaos. To find stability
ions, the three models stated befo the dynamics can also bifurcate showing high period oscillations or even chaos. To find stability
regions, the three models stated before will be employed.
A. Free parameters
First, let us distinguish between design and o

	Design	Operation	
	$D: \mbox{ROV diameter}$	${\cal L}$: cable length	
	A_0 : cable cross section	ω : wave frequency	
	m : ROV mass	$a:$ wave amplitude	(27)
	C_D (shape): drag coefficient	$C_D(\text{Re})$: drag coefficient	
	m_∞ : added mass	F_t : vertical thruster force	
	\boldsymbol{E} : Young's modulus	$b:$ crane jib elevation	
	Table 1: Design and operation parameters stability regions with a set of them conforming the vector		
	$\pmb{\mu} = [D,EA_0, C_D(\text{shape}), C_D(\text{Re}), L, F_t, a, \omega]^T$.		(28)
	The ROV mass m and added mass m_{∞} are not directly employed in (28), but through the		
	relations $\frac{m}{\rho \pi D^3/6}$ = $c_1 > 1$ and $\frac{m_{\infty}}{\rho \pi D^3/6}$ = $c_2 = 0.5$, respectively, with c_1 and c_2 being specified		
constants.			
	In order to perform simulations for several kinds of operations and for a wide class of umbilical-		
	ROVs with spherical shell, the basic settings are prescribed mostly in intervals (see Table (2).		

$$
\boldsymbol{\mu} = [D, EA_0, C_D(\text{shape}), C_D(\text{Re}), L, F_t, a, \omega]^T. \tag{28}
$$

relations $\frac{m}{\rho \pi D^3/6}$ = $c_1 > 1$ and $\frac{m_{\infty}}{\rho \pi D^3/6}$ = $c_2 = 0.5$, respectively, with c_1 and c_2 being specified m_{∞} : added mass F_t : vertical thruster force
 E : Young's modulus b : crane jib elevation

Table 1: Design and operation parameters

gions with a set of them conforming the vector
 $\mu = [D, EA_0, C_D(\text{shape}), C_D(\text{Re}), L, F_t, a,$ constants. Table 1: Design and operation parameters

ility regions with a set of them conforming the vector
 $\mu = [D, EA_0, C_D(\text{shape}), C_D(\text{Re}), L, F_t, a, \omega]^T$. (28)

FROV mass m and added mass m_{∞} are not directly employed in (28), but thr

	stability regions with a set of them conforming the vector			
			$\mu = [D, EA_0, C_D(\text{shape}), C_D(\text{Re}), L, F_t, a, \omega]^T$.	
			The ROV mass m and added mass m_{∞} are not directly employed in (28), but through the	
			relations $\frac{m}{\rho \pi D^3/6}$ = $c_1 > 1$ and $\frac{m_{\infty}}{\rho \pi D^3/6}$ = $c_2 = 0.5$, respectively, with c_1 and c_2 being specified	
constants.				
			In order to perform simulations for several kinds of operations and for a wide class of umbilical-	
			ROVs with spherical shell, the basic settings are prescribed mostly in intervals (see Table (2).	
	Design parameters	Span	Operation parameters	S pan
	$D=1[\mathrm{m}]$	[0.5:2]	$L = 50[\text{m}]$	$[1:10^2]$
	$EA_0 = 10^6 [N]$	$[10^5:10^7]$	$\omega = 1 \text{[rad/s]}$	$[10^{-1}:5]$
	$\frac{m}{\rho \frac{\pi D^3}{6}} = 1.1$ -		$a=1[\mathrm{m}]$	[0:3]
	$C_D(\text{shape}) = 0.2$		$C_D(\text{Re})$	see Fig. (3)
	$b=3[m]$		$F_t = 0[N]$	$[-600:600]$
			Table 2: Basic simulation parameters	
			Due to the large dimension of the free-parameter space, stability regions are constructed in	
			21	

subspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate orbi subspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate orbi subspaces conformed by pairs of components of μ , while the complement of each of them is maintained constant. In order to identify the kind of oscillation, particular stability regions are shaded so as to indicate orbi subspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate orbi is
ubspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate or (26). subspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate orbi subspaces conformed by pairs of components of μ , while the complement of each of them is
maintained constant. In order to identify the kind of oscillation, particular stability regions are
shaded so as to indicate orbi mantamed constant. In order to identity the kind of oscillation, particular stability reshaded so as to indicate orbits with the same periodicity. Also each orbit is depicte
symbol that identifies its periodicity. The dete

ded so as to indicate orbits with the same periodicity. Also each orbit is depicted with a
hold that identifies its periodicity. The detection of periodicity is performed according to the
attification method developed prev symbot that identifies its periodicity. The detection of periodicity is performed according to the identification method developed previously on the basis of Cauchy series for a tolerance given by (26).

B. Stability accor 1. Generally speaking to Model 1

1. Stability according to Model 1

1. Generally seed in the taut-slack phenomenon and its stability properties, simulations are carried out on

1. Generally speaking, it is seen that the (26).

B. Stability according to Model 1

To study the taut-slack phenomenon and its stability properties, simulations are carried out on

the basis of model (4)-(5).

Figures from (10) up to (16) illustrate the stability B. Stability according to Model 1
To study the taut-slack phenomenon and its stability properties, simulations are carried out on
the basis of model (4)-(5).
Figures from (10) up to (16) illustrate the stability region in To study the taut-slack phenomenon and its stability properties, simulations are carried out on
the basis of model $(4)-(5)$.
Figures from (10) up to (16) illustrate the stability region in different subspaces correspon the basis of model (4)-(5).

Figures from (10) up to (16) illustrate the stability region in different subspaces corresponding

to an experiment series for a constant drag coefficient, which is the main particularity of m Figures from (10) up to (16) illustrate the stability region in different subspaces corresponding
to an experiment series for a constant drag coefficient, which is the main particularity of model
1. Generally speaking, it Figure from (E) up to (25) matrials the chability region in sinterin subspace corresponding
to an experiment series for a constant drag coefficient, which is the main particularity of model
1. Generally speaking, it is se 1. Generally speaking, it is seen that the stability region is composed by definition of behaviors of periodicity one, termed P1, with a taut condition fulfilled. Outside the stability region the diversity of behavior is of periodicity one, termed P1, with a taut condition fulfilled. Outside the stability
diversity of behavior is wide, ranging from P1 up to chaos. The presence of perio
is not a characteristic of the stability regions as, Fig. (10) shows the role of the monochromatic wave excitation through its parameters of period doubling
to a characteristic of the stability regions as, for instance, this occurs in related ODEs like
Mathieu and Duffing q not a characteristic of the stability regions as, for instance, this occurs in related ODEs like
not a characteristic of the stability regions as, for instance, this occurs in related ODEs like
e-Mathieu and Duffing quadr the Mathieu and Duffing quadratic nonlinear differential equations. The reason for that is the presence of two actuating nonlinearities, *i.e.*, due to the bilinear and the quadratic characteristics for the dable force an the stability region and that of chaos is thin and composed most
stability in the stability characteristics for the cable force and drag, respectively. Moreover, the behavior diversity in the subspace is
characterized wit

 $\left[\text{N}\right]$ and $% \left[\text{N}\right]$ and $% \left[\text{N}\right]$ and $\left[\text{N}\right]$ and $\left[\text{N}\right]$ and $\left[\text{N}\right]$ and $\left[\text{N}\right]$ For a middle cable length. Accordingly, one sees that the larger is the diameter of the ROV, the larger

more elastic has to be the cable in order to avoid the taut-slack condition. Fig. (12) shows a marked insensitivity of the oscillation with depth. This occurs is a matching of the oscillation with depth. Accordingly, one sees that the larger is the diameter of the ROV, the original and cable stiff Figure 10: Stability region, wave amplitude vs. frequency for $C_D = 0.2$, $EA_0 = 10^6$ [N] and $L = 50$ [m]

Fig. (11) demonstrates the balance between the ROV mass through D and cable stiffness

for a middle cable length, A $L = 50$ [m]

Fig. (11) demonstrates the balance between the ROV mass through D and cable stiffness

for a middle cable length. Accordingly, one sees that the larger is the diameter of the ROV, the

more clastic has to be Fig. (11) demonstrates the balance between the ROV mass through D and cable stiffness
for a middle cable length. Accordingly, one sees that the larger is the diameter of the ROV, the
more elastic has to be the cable in A similar insensitivity, one sees that the larger is the diameter of the ROV, the
re-distic has to be the cable in order to avoid the taut-slack condition.
Fig. (12) shows a marked insensitivity of the oscillation with de relation of the depth with the wave amplitude, see Fig. (13). It is noticed that as except for the depth with depth. This occurs inside and outside the stability region, except for superficial depths, for which the stiffn Fig. (12) shows a marked insensitivity of the oscillation with depth. This occurs inside and
outside the stability region, except for superficial depths, for which the stiffness is high, *i.e.*,
where *L* is small. Roughl suggests a disconnection of both stable portions in the subspace consideration of the stiffness is high, *i.e.*, where *L* is small. Roughly speaking, the smaller *L* and the larger the stiffness, the more feasible is tha

Fig. (14) shows the effect of the ROV thrust and the wave excitation on the system one feasible that the ROV can follow the harmonic motion of the jib. Conversely, the behavior shows a at sensibility with wave frequency.
 about a first bulk of a set of the harmonic motion of the jib. Conversely, the behavior shows a great sensibility with wave frequency.

A similar insensitivity, yet not so pronounced as in the case before, is encountered

Figure 12: Stability region: ROV depth vs. frequency for $C_D = 0.2$, $a = 1$ [m] and $EA_0 = 10^6$ [N]

 $EA_0 = 10^6 [N]$

larger is the thruster power, the larger would be the stability region even for increasing wave $E_A = 0$
 $\frac{1}{\frac{16}{10} + \frac{1}{10}} = 0$
 $\frac{1}{\frac{16}{10} + \frac{16}{10}} = 2$
 $\frac{1}{25} = 3$
 $\frac{1}{\frac{16}{10} + \frac{16}{10}} = 2$
 $\frac{1}{25} = 3$
 $\frac{1}{25} = 3$
 $\frac{1}{25} = 3$
 $\frac{1}{25} = 3$

Figure 13: Stability regions: ROV depth vs. w **Example 19** of $\frac{1}{\text{cm}}$ is $\frac{1}{10}$ **i** $\frac{1}{10}$ 2, $\omega = 1$ [rad/s] and
for increasing wave
(see at $F_t < 0$) the
. This indicates no
ifferent depths. It is
e tension sufficiently Figure 13: Stability regions: ROV depth vs. wave amplitude for $C_D = 0.2$, $\omega = 1$ [rad/s] and $EA_0 = 10^6$ [N]

larger is the thruster power, the larger would be the stability region even for increasing wave

amplitudes. Con and $\sigma = 10^6$ [N]
 $\sigma = 10^6$ [N]

for is the thruster power, the larger would be the stability region even for increasing wave

plitudes. Conversely, for the thrusters actuating in the other direction (see at $F_k < 0$) the Figure 13: Stability regions: ROV depth vs. wave amplitude for $C_D = 0.2$, $\omega = 1$ [rad/s] and $EA_0 = 10^6$ [N]
larger is the thruster power, the larger would be the stability region even for increasing wave
amplitudes. Conve $EA_0 - 10^6$ [N]
larger is the thruster power, the larger would be the stability region even for increasing wave
amplitudes. Conversely, for the thrusters actuating in the other direction (see at $F_t < 0$) the
ROV is pulled

ger is the thruster power, the larger would be the stability region even for increasing wave-
plitudes. Conversely, for the thrusteers actuating in the other direction (see at $F_i < 0$) the
V is pulled to the surface and t amplitudes. Conversely, for larger sound of the balance pelotic amplitudes. Conversely, for the thrusters actuating in the other direct ROV is pulled to the surface and the "slack" condition arises for large stable orbit The fact that the fact that with given the fact that $F_t < 0$) the urface and the "slack" condition arises for large F_t . This indicates no ded behavior.
the effect of the thruster force on the stability for different dep energy is the interest accuracy in the system (e.g., increases) the system (e.g., i.e., i slable orbit but unbounded behavior.

Fig. (15) illustrates the effect of the thruster force on the stability for different depths. It is

seen that stable oscillations occur when the actuators can maintain the cable tens Fig. (15) illustrates the effect of the thruster force on the stability for different depths. It is
seen that stable oscillations occur when the actuators can maintain the cable tension sufficiently
large. The same as bef Fig. (15) and
attach the check of the fundators can maintain the cable tension sufficiently
barge. The same as before is said for the portion shaded as "slack".
The last figure, Fig (16), depicts similar results as Fig. (stability.

and $EA_0=10^6[\rm N]$

 10^6 [N] and $L = 50$ [m]

 $\begin{vmatrix} 0 & 0 & \frac{\sqrt{3}}{2} & \frac{\sqrt{3}}{2}$ Figure 16: Stability region: wave frequency vs. thruster force for $C_D = 0.2$, $a = 1$ [m], $EA_0 =$
Figure 16: Stability region: wave frequency vs. thruster force for $C_D = 0.2$, $a = 1$ [m], $EA_0 =$
10⁶[N] and $L = 50$ [m]
C. St

Eigure 16: Stability region: wave frequency vs. thruster force for $C_D = 0.2$, $a = 1$ [m], $EA_0 = 10^6$ [N] and $L = 50$ [m]
C. Stability according to Model 2
In this section model (9)-(10) is considered with the same setup for ure 16: Stability region: wave frequency vs. thruster force for $C_D = 0.2$, $a = 1$ [m], $EA_0 =$
[N] and $L = 50$ [m]
Stability according to Model 2
this section model (9)-(10) is considered with the same setup for simulations Figure 16: Stability region: wave frequency vs. thruster force for $C_D = 0.2$, $a = 1$ [m], $EA_0 = 10^6$ [N] and $L = 50$ [m]

C. Stability according to Model 2

In this section model (9)-(10) is considered with the same setup f 10⁶[N] and $L = 50$ [m]
C. Stability according to Model 2
In this section model (9)-(10) is considered with the same setup for simulations as in model 1.
The main improvement of model 2 in comparison to model 1 is the inc C. Stability according to Model 2
In this section model (9)-(10) is considered with the same setup for simulations as in model 1.
The main improvement of model 2 in comparison to model 1 is the incorporation of a variable In this section model (9)-(10) is considered with the same setup for simulations as in model 1.
The main improvement of model 2 in comparison to model 1 is the incorporation of a variable
drag coefficient with motion depe Fig. (17) shows similitudes in the behavior diversity with respect to the homologonal similing proposed in the behavior of a variable goodflicient with motion dependence. The experiments are illustrated in Figs. (17) to (In the improvement of model ϵ in computable to model ϵ is the model provided in Figs. (17) to (23).
In general, it is noticed, that the diversity of orbits is qualitatively broader than in the
cases handled before. too significant and accordingly the drag coefficient does not vary to continue the much dependence. This feature was expected due to the complexity broader than in the cases handled before. This feature was expected due t or general, and the stability variable the stability regional than at many points in the stability regions remain almost the same as in the case earlier. Particular differences will be exalted comparatively with respect t

 $\left[\text{N} \right]$ and $% \left[\text{N} \right]$ and $% \left[\text{N} \right]$ and $\left[\text{N} \right]$ and $\left[\text{N} \right]$ and $\left[\text{N} \right]$

small.

Similar conclusions are worked out from Fig. (18) with respect to Fig. (11). A difference to

and on the rate of the specifical property for C_D variable, $EA_0 = 10^6$ [N] and

im the specific is to be to Fig. (18) with re Solution of the stability region: wave amplitude vs. frequency for C_D variable, $EA_0 = 10^6$ [N] and $L = 50$ [m]

Figure 17: Stability region: wave amplitude vs. frequency for C_D variable, $EA_0 = 10^6$ [N] and $L = 50$ [m]
 Fig. (19) illustrates a marked insensitivity of the oscillation with respect to F_A or B is B in B is B is Figure 17: Stability region: wave amplitude vs. frequency for C_D variable, $EA_0 = 10^6$ [N] and $L = 50$ [m]
small.
Similar conclusions are worked out from Fig. (18) with respect to Fig. (11). A difference to
stand out is t

Figure 17: Stability region: wave amplitude vs. frequency for C_D variable, EA
 $L = 50$ [m]

small.

Similar conclusions are worked out from Fig. (18) with respect to Fig. (11).

stand out is the enlargement of the stabi -50 [m]
Similar conclusions are worked out from Fig. (18) with respect to Fig. (11). A difference to
ad out is the enlargement of the stability region for small D and large stiffness.
Fig. (19) illustrates a marked ins small.

Similar conclusions are worked out from Fig. (18) with respect to Fig. (11). A difference to

stand out is the chargement of the stability region for small D and large stiffness.

Fig. (19) illustrates a marked Similar conclusions are worked out from Fig. (18) with respect to Fig. (11). A difference to stand out is the enlargement of the stability region for small D and large stiffness.

Fig. (19) illustrates a marked insensit Fig. (19) it is to make the related bor and ϵ (16) and related of ϵ (19). It canceled bordom of the stability region for small D and large stiffness.
Fig. (19) illustrates a marked insensitivity of the oscillation w Fig. (19) illustrates a marked insensitivity of the oscillation with respect to the length. The difference here with respect to the homologous case in Fig. (12) is that the diversity in the frequency is higher.

Similar c are view more than in the homologous case in Fig. (12) is that the diversity in the firence here with respect to the homologous case in Fig. (12) is that the diversity in the frequency is higher.

Similar conclusions are The two next guesses of the individual from Fig. (29) with respect to Fig. (13). The variant here interesting an extendion of the chaotic zone is broader and periodic solutions of odd periodicity are more common, creas in Similar conclusions are deduced from Fig. (20) with respect to Fig. (13). The variant here
is that the chaotic zone is broader and periodic solutions of odd periodicity are more common,
whereas in the previous homologous

and $EA_0=10^6[\rm N]$

 $EA_0=10^6[\rm N]$

[rad/s] and $EA_0 = 10^6$ [N]

di erences are signicative. Nevertheless the order of the diversity in homologous cases is not ⁶⁰⁰ ⁴⁰⁰ ²⁰⁰ ²⁰⁰ ²⁰⁰ ²⁰⁰ ⁶⁰⁰

⁶⁰⁰ $\frac{F_{t}[\mathbf{N}]}{F_{t}[\mathbf{N}]}$

²⁰⁰ $\frac{F_{t}[\mathbf{N}]}{F_{t}[\mathbf{N}]}$

²⁰¹ $\frac{F_{t}[\mathbf{N}]}{F_{t}[\mathbf{N}]}$

Figure 22: Stability region: ROV depth vs. thruster force for C_D v

Eigure 22: Stability region: ROV depth vs. thruster force for C_D variable, $a = 1[\text{m}], \omega = 1$
[rad/s] and $EA_0 = 10^6[\text{N}]$
comparison to those of their homologous cases of Figs. (15) and (16). However outside them the
dif Figure 22: Stability region: ROV depth vs. thruster force for C_D variable, $a = 1[m], \omega = 1$
[rad/s] and $EA_0 = 10^6[N]$
comparison to those of their homologous cases of Figs. (15) and (16). However outside them the
differenc (nd/s) and $EA_0 = 10^6$ [N]
comparison to those of their homologous cases of Figs. (15) and (16). However outside them the
differences are significative. Nevertheless the order of the diversity in homologous cases is not
to comparison to those of their homologous cases of Figs. (15) and (16). However outside them the differences are significative. Nevertheless the order of the diversity in homologous cases is not too dissimilar.
D. Stability comparison to those of their nomologous cases of Figs. (15) and (16). However outside them the differences are significative. Nevertheless the order of the diversity in homologous cases is not too dissimilar.

D. Stubilit differences are significative. Nevertheless the order of the diversity in homotogous cases is not
too dissimilar.
D. Stability according to Model 3
Finally, model (17)-(18) is simulated under the same scheduling as former D. Stability according to Model 3 and their same scheduling as former models. Apart from having a motion-dependent drag coefficient like model 2, the improvement provided by this model is the consideration of the potentia Stability according to Model 3
ally, model (17)-(18) is simulated under the same scheduling as former models. Apart from
ing a motion-dependent drag coefficient like model 2, the improvement provided by this model
the con Finally, model (17)-(18) is simulated under the same scheduling as former models. Apart from having a motion-dependent drag coefficient like model 2, the improvement provided by this model is the consideration of the pote in the periodicity. However, the chaotic behaviors remain in the same positions in the space of the consideration of the potential radiation force F_r . As this force is significant, mainly for shallow waters, the experim

 10^6 [N] and $L = 50$ [m]

Figure 24: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]

Fig. (25) depicts a scenario at a depth $d = 2[m]$. This is characterized by equal stability
ions with and without F_r , and almost identical variations in the periodicity in both cases.
chaotic region is comparatively slig Fig. (25) depicts a scenario at a depth $d = 2[m]$. This is characterized by equal stability regions with and without F_r , and almost identical variations in the periodicity in both cases.
The chaotic region is comparative Fig. (25) depicts a scenario at a depth $d = 2[m]$. This is characterized by equal stability
regions with and without F_r , and almost identical variations in the periodicity in both cases.
The chaotic region is comparative

Figure 25: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]

Fig. (26) illustrates the periodicity at a depth $d = 5$ [m] comparatively. The stability regions:

Fig. $\frac{6}{2}$ or $\frac{6}{2}$ or $\frac{6}{2}$ or $\frac{6}{2}$ or $\frac{6}{2}$
 $\frac{6}{24}$ or $\frac{6}{24}$ or $\frac{6}{24}$ or $\frac{6}{24}$ or Example $\frac{1}{2}$ or $\frac{1$ Figure 25: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]
and $L = 5$ [m]. Top: simulation without radiation force. Bottom: simulation with radiation force
Fig. (26) illustrates the p Figure 25: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]
and $L = 5$ [m]. Top: simulation without radiation force. Bottom: simulation with radiation force
Fig. (26) illustrates the p Figure 25: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]
and $L = 5$ [m]. Top: simulation without radiation force. Bottom: simulation with radiation force
Fig. (26) illustrates the p dicity at a depth $d = 5$ [m] comparatively. The stability regions
gions with presence of chaos are very similar. The periodicity
nly, however not so abrupt when considering F_r .
forces have an insignificant influence in t

In the sinking/lifting operation the state of the cable describes actually increases at some specific points only, however not so abrupt when considering F_r . It is concluded that radiation forces have an insignificant i changes at some specific points only, however not so abrupt when considering F_r .

It is concluded that radiation forces have an insignificant influence in the system dynamics

for depths $d \ge 5$ [m]. As the usual depths It is concluded that radiation forces have an insignificant influence in the system dynamics
for depths $d \geq 5[m]$. As the usual depths of the ROV in the operation are much larger than this
limit, it is inferred that mode

Figure 26: Stability regions: wave amplitude vs. wave frequency for C_D variable, $EA_0 = 10^6$ [N]

Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1$ m], frequency $\omega = 1.87$ [rad/s], stiffness constant $EA_n = 10^6$ [N] and cable length $L = 50$ [m]
taut through the taut-slack state. This qualitative ch Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1$ [m], frequency $\omega = 1.87$ [rad/s], stiffness constant $EA_0 = 10^6$ [N] and cable length $L = 50$ [m]
taut through the taut-slack state. This qualitative c Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1 \text{ [m]}$, frequency $\omega = 1.87 \text{ [rad/s]}$, stiffness constant $EA_0 = 10^6 \text{ [N]}$ and cable length $L = 50 \text{ [m]}$
taut through the taut-slack state. This qualita Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1|m$, frequency $\omega = 1.87[\text{rad/s}]$, stiffness constant $EA_0 = 10^6[N]$ and cable length $L = 50[m]$
taut through the taut-slack state. This qualitative change oc Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1$ [m], frequency $\omega = 1.87$ [rad/s], stiffness constant $EA_0 = 10^6$ [N] and cable length $L = 50$ [m]
taut through the taut-slack state. This qualitative chan Figure 27: Evolution of the cable force for a wave amplitude $a = 0.1|m$, frequency $\omega = 1.87[\text{rad/s}]$, stiffness constant $EA_0 = 10^6[\text{N}]$ and cable length $L = 50[\text{m}]$
taut through the taut-slack state. This qualitative cha 1.87[rad/s], stiffness constant $EA_0 = 10^6$ [N] and cable length $L = 50$ [m]
taut through the taut-slack state. This qualitative change occurs during the transient behavior
and is typically characterized by abrupt and hefty taut through the tant-slack state. This qualitative change occurs during the transient behavior and is typically characterized by abrupt and hefty increments of the force magnitude due to accelerations of the upper extrem Another advantage of preserving the taut state in the force magnitude of the state in the observations of the upper extreme of the cable during the slack condition followed by violent ks when the cable tows the ROV again. evolution of the upper extreme of the cable during the slack condition followed by violent
variables when the cable tows the ROV again. This scenario takes place at higher frequencies
depending of the natural frequency of (10)-(23) that the system behavior is always periodic P1 in the system constituted by ROV and cable.
The shorter is the cable length, the larger will be the frequency of the taut-slack state evolution.
The shorter is the

then the case of the maximum scalar chaos with the max-spring system constituted by ROV and cable.
The shorter is the cable length, the larger will be the frequency of the taut-slack state evolution.
Also the maximal magn the
paramog of the lands in equality of the limit separation of the shorten
is a symbol of the slack phenomenon. The shortenistic dispersion of
the maximal magnitude of the force in the taut-slack condition depend
magnitu Bearing the measure of the force in the taut-slack condition depends directly on the maximal magnitude of the force in the taut-slack condition depends directly on the guitade of the wave steepness $(a\omega)$.
Another advanta magnitude of the wave steepness $(a\omega)$.

Another advantage of preserving the taut state in the operation, is the more predictable

evolution of the ROV trajectory than under the taut-slack state. It is clearly seen from F

taut condition. Additionally, a practical requirement by the descent or ascent of the unit is to minimize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV taut condition. Additionally, a practical requirement by the descent or ascent of the unit is to
minimize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV t condition. Additionally, a practical requirement by the descent or ascent of the unit is to imize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV thrust tant condition. Additionally, a practical requirement by the descent or ascent of the unit is to minimize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV full the term for reaching a practical requirement by the descent or ascent of the unit is to minimize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV th the term of any controlled system and point of the stability requirement by the descent or ascent of the unit is to minimize the times required for these operations.
To achieve these controll objectives, the hoisting eran taut condition. Additionally, a practical requirement by the descent or ascent of the unit is to
minimize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV t condition. Additionally, a practical requirement by the descent or ascent of the unit is to unitize the times required for these operations.
To achieve these control objectives, the hoisting crane system and the ROV thr

namely the set points d_{eve} paradoxic requirement of the decline of decline of the unit is to minimize the times required for these operations.

To achieve these control objectives, the hoisting crane system and the ROV To achieve these control objectives, the hoisting crane system and the ROV thrusters are
involved in a controller design. They must properly be synchronized in a simultaneous, optimal
and secure form for reaching a desire To value of these control expective, the messuring order opposition and the recoverance are
involved in a controller design. They must properly be synchronized in a simultaneous, optimal
and secure form for reaching a des (28)) benefit of any controlled operation be a significative extension of the stability regions with
benefit of any controlled system.
To this end, the control system can be conceived as a dynamic system with two inputs,
nely t the state in the differential operator of extremes to denominate the state into the state of the state of the critical system. To this end, the control system can be conceived as a dynamic system with two inputs, namely t To this end, the control system can be conceived as a dynamic system with two inputs,
namely the set points d_{ref} and F_{eq} for depth and a suitable cable strength, respectively, and
an unavoidable wave perturbation $a \$

for any distribution of the ROV of the Capital and a suitable cable strength, respectively, and
an unavoidable wave perturbation $a\sin(\omega t)$. On the other side, it would have three measurable
outputs, namely the ROV velocit coupled through multiple feedbacks as seen from the proposed structure in Fig. (see Fig. (28))
In the control strategy can be achieved with the help of two mechanisms. First, the cable
(28))
So, the control strategy can b Interaction which were presentation with $\epsilon_{\rm w}$, the cable length L and the cable tension F_c (see Fig.)))

So, the control strategy can be achieved with the help of two mechanisms. First, the cable

sion is regula (28))
So, the control strategy can be achieved with the help of two mechanisms. First, the cable
tension is regulated from both extremes using controllers on the crane motor and the ROV
thrusters, respectively. On the oth So, the control strategy can be achieved with the help of two mechanisms. First, the cable
tension is regulated from both extremes using controllers on the crane motor and the ROV
thrusters, respectively. On the other han the help of two mechanisms. First, the cable
controllers on the crane motor and the ROV
to track desired trajectories for ascent/descent
led separately. All controllers are nonlinearly
the proposed structure in Fig. (28).

energy for example of the economic of the economic of the economic reaching control and the ROV densities, respectively. On the other hand, in order to track desired trajectories for ascent/descent fast and accurately, th observation is regarded than down additional wave, comparison on the state mathematic on the characteristic steep.

thrusters, respectively. On the other hand, in order to track desired trajectories for ascent/descent

fa

vector (the metric of a ROV)

(the metric of a ROV)

(domeontrol law is proposed with a control action

(t), $u_{cr}(t)|^T$, (29)

(d), where metric voltage (see Fig. (28)).

(d) as a fraction of the fracture tension. The cable

(h) is the thruster voltage and ucreation in lowering/lifting operation of a ROV in
ear control law
c the control goal, a two-degree-of-freedom control law is proposed with a control action
 $\mathbf{u}(t) = [u_t(t), u_{\sigma}(t)]^T$, (29)
is Figure 28: Control of depth and cable tension in lowering/lifting operation of a ROV
Nonlinear control law
achieve the control goal, a two-degree-of-freedom control law is proposed with a control action
for
 $\mathbf{n}(t) = [n_t(t),$ Figure 28: Control of depth and cable tension in lowering/lifting operation of a ROV
A. Nonlinear control law
To achieve the control goal, a two-degree-of-freedom control law is proposed with a control action
vector
 $u(t) =$ Nonlinear control law
achieve the control goal, a two-degree-of-freedom control law is proposed with a control action
for
 $\mathbf{n}(t) = [n_t(t), n_{cr}(t)]^T, \eqno{(29)}$
 $\text{are } n_t \text{ is the thrust voltage and } n_{cr} \text{ the came motor voltage (see Fig. (28)).}$
The set point for cable stre

$$
\mathbf{u}(t) = \left[u_t(t), u_{cr}(t)\right]^T,\tag{29}
$$

where u_t is the thruster voltage and u_{cr} the crane motor voltage (see Fig. (28)).

$$
F_v = -\frac{\pi \rho D^2}{8} C_D \text{ (} \text{P}cd \text{)} \text{ } \text{P}cd \text{ } |\text{P}cd| \tag{30}
$$

are included in the forces and moments, namely
\n
$$
F_v = -\frac{\pi \rho D^2}{8} C_D \text{ (Pcd) Pcd |Pcd|}
$$
\n
$$
F_c = \begin{cases}\n-\frac{EA_0}{L} z, \text{ for } z \ge 0 \\
0, \text{ otherwise}\n\end{cases}, \text{ with } z = d - L + b - a \sin(\omega t) \tag{31}
$$

are included in the forces and moments, namely
\n
$$
F_v = -\frac{\pi \rho D^2}{8} C_D (\text{Re}d) \text{ Re}d |\text{Re}d|
$$
\n(30)
\n
$$
F_c = \begin{cases}\n-\frac{E A_0}{L} z, \text{ for } z \ge 0 \\
0, \text{ otherwise}\n\end{cases}, \text{ with } z = d - L + b - a \sin(\omega t) \tag{31}
$$
\n(32)
\n
$$
F_t = \begin{cases}\n\frac{K_t}{s^2 + n^2 + s_0} u_t |u_t|, \text{ for } u_t \in [u_{tmin}, u_{tmax}] \\
F_{tmin}, \text{ for } u_t \le u_{tmin} \\
F_{tmin}, \text{ for } u_t \le u_{tmin}\n\end{cases}
$$
\n(32)
\n
$$
F_{tmax}, \text{ for } u_t \le u_{tmin}
$$
\n(33)
\n
$$
M_{cr} = \begin{cases}\n\frac{k_1 s}{\frac{r_{dr} s}{k_1 k_2} s^{2} + \frac{R_{td} s}{k_1 k_2} s + 1} u_{cr}, \text{ for } u_{cr} \in [u_{cr min}, u_{cr max}] \\
M_{cr min}, \text{ for } u_{cr} \ge u_{cr min}\n\end{cases}
$$
\n(33)
\nwhere γ_0 and γ_1 are coefficients of the thrust inductance and resistance of the crane motor,
\nrespectively, *J* the moment of inertia, *r* the radius of the wrapping drum, k_1 the transfer gain
\nbetween the armature current and the drum angular acceleration, and finally k_2 the transfer gain
\nbetween the armature current and the drawn angular acceleration, and finally k_2 the transfer gain
\nbetween the drum angular speed and the back c.f.m. The coefficients u_{rmin} , u_{rmin} , u_{rmin} , u_{rmin} , u_{rmin}

 $\begin{aligned} F_t &= \left\{ \begin{array}{ll} \frac{K_t}{s^2 + \gamma + t \gamma_0} u_t \left| u_t \right|, \text{ for } u_t \in [u_{\text{trmin}}, u_{\text{tmax}}] & (32) \\ F_{\text{train}}, \text{ for } u_t \leq u_{\text{tmin}} & (32) \end{array} \right. \right. \\ F_{\text{trmax}}, \text{ for } u_t \geq u_{\text{tmax}} & (32) \end{aligned}$
 $M_{\sigma\tau} &= \left\{ \begin{array}{ll} \frac{k_1 s}{s_{\text{tr}} s^2 + \frac{K_t s}{s_{\text{tr}} s^2 +$ F_t = $\begin{cases} \frac{s^2 + \gamma_1 s + \gamma_0 s^{-1} s^{-1} \gamma_1 \gamma_0 \dots \gamma_1 - \gamma_1 \gamma_0 \dots \gamma_1 - \gamma_1 \gamma_0 \dots \gamma_1$ between the armature current and the brackers.

From the $\sum_{n=1}^{\infty} h_n^2 \frac{k_1}{n_1^2 n_2^2 + 1} u_{\text{cr}}$, for $u_{\text{cr}} \leq u_{\text{cr,min}}$ (as) $M_{\text{cr,max}}$, for $u_{\text{cr}} \leq u_{\text{cr,min}}$ (as) $M_{\text{cr,max}}$, for $u_{\text{cr}} \geq u_{\text{cr,min}}$ (as) $M_{$ $M_{cr} = \begin{cases} \frac{L}{\frac{L}{2} + 2L_{cr}} \sum_{i=1}^{L} \frac{L_{12}}{2L_{cr} + \frac{L_{12}}{4L_{cr} + \$ $M_{cr} = \begin{cases} \frac{L_{c1}}{6\pi k_{2}^2}e^{2}+\frac{R_{c2}}{8\pi k_{2}^2+11}u_{cr}$, for $u_{cr} \leq u_{cr,min}$ for $u_{cr} \leq u_{cr,min}$ (33)
 $M_{cr,max}$, for $u_{cr} \leq u_{cr,min}$ and γ_1 are coefficients of the thruster motor dynamics, K_t is its gain, M_{cr} t $H_{\text{c}} = \int_{M_{\text{c}} \text{min}}^{M_{\text{c}} \text{min}} M_{\text{c}}$ is $u_{\text{cr}} \le u_{\text{c}}$ and γ_1 are coefficients of the thruster motor dynamics, K_i is its gain, M_{cr} the moment the crane drum, L_a and R_a the armature inductance (M_{ormax} , for $u_{\text{cr}} \geq n_{\text{or max}}$ and γ_4 are coefficients of the thruster motor dynamics, K_t is its gain, M_{or} the moment of the crane drum, L_a and R_a the armature inductance and resistance of the crane

since y and μ at examinate of the armature inductance and resistance of the crane motor,
of the crane drum, L_a and R_a the armature inductance and resistance of the crane motor,
respectively, J the moment of inert respectively, *J* the moment of inertia, *r* the radius of the wrapping drum, k_1 the transfer greepectively, *J* the moment of inertia, *r* the radius of the wrapping drum, k_1 the transfer greetween the armature cur anadate inductance and respectively, k_1 the transfer gain
e drum angular acceleration, and finally k_2 the transfer gain
the back e.f.m. The coefficients u_{tmin} , u_{tmax} , u_{cmin} , u_{cmax}
thrusters and c the lack of smoothness, nonlinear controls based on differential geometric scale of the rental smooth between the drum angular speed and the back e.f.m. The coefficients u_{train} , u_{train} , u_{train} , u_{train} , $u_{\text{train$ to between the druman university of the threat supplies and achiever the coefficients u_{min} , u_{tmax} , u_{cmin} , u_{cmin} , u_{cmin} and u_{cmin} and u_{cmin} are achiever are limiting saturation values of Exercis the thermal surgical operator and the basis cannot in the contribute origins, we many series are limiting saturation values of the thrusters and crane motor, respectively.

The nonlinearity (30) is nonconvex over The nonlinearity (30) is nonconvex over an interval that depends on the ROV diameter
The nonlinearity (30) is nonconvex over an interval that depends on the ROV diameter
Fig. (5)). Additionally, it is only two times deriv (see Fig. (5)). Additionally, it is only two times derivable with respect to Red because of the singularity at $\Re d = 0$. It similarly occurs with the nonlinearities (31) and (32), whose high derivatives with respect to z

corrections termed as $\delta \text{Re} L$ and $\delta \text{Re} d$ for the crane system and the ROV thrusters, respectively.
For these specific tasks, the equilibrium stress point of the cable given by the restriction
 $\text{Re} d - \text{Re} L - a \omega \cos(\omega$ rections termed as δ PeL and δ Ped for the crane system and the ROV thrusters, respectively.
For these specific tasks, the equilibrium stress point of the cable given by the restriction
 $\theta c d - Pc L - a\omega \cos(\omega t) = 0,$ (34)
a

$$
Pcd - PcL - a\omega \cos(\omega t) = 0,\t\t(34)
$$

$$
\delta \text{Red} \text{ for the crane system and the ROV thrusters, respectively.}
$$
\nequilibrium stress point of the cable given by the restriction

\n
$$
\text{Red} - \text{Re}L - a\omega \cos(\omega t) = 0,\tag{34}
$$
\n
$$
\text{Red} - \text{Re}L - a\omega \cos(\omega t) = \delta \text{Red},\tag{35}
$$

corrections termed as
$$
\delta \text{Re}L
$$
 and $\delta \text{Re}d$ for the crane system and the ROV thrusters, respectively.
For these specific tasks, the equilibrium stress point of the cable given by the restriction

$$
\text{Re}d - \text{Re}L - a\omega \cos(\omega t) = 0, \tag{34}
$$
is modified to

$$
\text{Re}d - \text{Re}L - a\omega \cos(\omega t) = \delta \text{Re}d, \tag{35}
$$
with the property

$$
\int_0^\infty |\delta \text{Re}d| dt = c_1 \tag{36}
$$
and $c_1 > 0$ being a constant for a bounded response. In this way, the lower point of the cable is

corrections termed as $\delta\!P\!c\!L$ and $\delta\!P\!c\!d$ for the crane system and the ROV thrus
For these specific tasks, the equilibrium stress point of the cable given by
 $\label{eq:2.1} \begin{split} \text{P}\!c\!d-\text{P}\!c\!L-a\omega\cos(\omega t)=0, \end{split}$ For these specific tasks, the equilibrium stress point of the cable given by the restriction
 $\text{Re}d-\text{Re}L-a\omega\cos(\omega t)=0, \tag{34}$ is modified to
 $\text{Re}d-\text{Re}L-a\omega\cos(\omega t)=\delta\text{Re}d, \tag{35}$ with the property
 $\int_0^\infty |\delta\text{Re}d|\,dt=c_$ then tensed conveniently by selecting the function $\delta \text{R} \cdot d - \delta \text{R} \cdot d$, (35)

with the property
 $\int_0^\infty |\delta \text{R} \cdot d| dt = c_1$ (36)

and $c_1 > 0$ being a constant for a bounded response. In this way, the lower point of th valid $\text{Red} - \text{Re}L - a\omega \cos(\omega t) = \delta \text{Re}d,$ (35)
 $\int_0^\infty |\delta \text{Re}d| dt = c_1$ (36)

a bounded response. In this way, the lower point of the cable is

ting the function $\delta \text{Re}d(t)$. Similarly, for the upper extreme it is
 $\text{Re}d - \text{Re$

$$
Pcd - PcL - a\omega \cos(\omega t) = \delta PcL,\tag{37}
$$

with $\sum_{i=1}^{n}$

$$
\int_0^\infty |\delta \mathbf{P} \mathbf{C} L| \, dt = c_2 \tag{38}
$$

 $\int_0^{\infty} |\partial \mathbf{r} \cdot d| \, d\mathbf{r} = c_1$ (36)

and $c_1 > 0$ being a constant for a bounded response. In this way, the lower point of the cable is

then tensed conveniently by selecting the function $\partial \mathbf{Rz}/(t)$. Similarly, and $c_1 > 0$ being a constant for a bounded response. In this way, the lower point of the cable is then tensed conveniently by selecting the function $\partial \bar{\partial}u(t)$. Similarly, for the upper extreme it is valid $\partial u(t) = \partial t + B$

then tensed conveniently by selecting the function $\delta \text{Rz}/(t)$. Similarly, for the upper extreme it is
valid
with
 $\text{Red} - \text{Rz}L - a\omega \cos(\omega t) = \delta \text{Rz}L$, (37)
with
 $\int_0^\infty |\delta \text{Rz}L| dt = c_2$ (38)
and $c_2 > 0$ being another c $\text{Re}L - \text{Re}L - a\omega \cos(\omega t) = \delta \text{Re}L,$ (37)

In
 $\int_0^\infty |\delta \text{Re}L| dt - c_2$ (38)
 $c_2 > 0$ being another constant for a bounded response. In the same way the upper extreme of

cable is then tensed conveniently by selecting the $\text{Re}d - \text{Re}L - a\omega \cos(\omega t) = \delta \text{Re}L,$ (37)

with $\int_0^\infty |\delta \text{Re}L| dt = c_2$ (38)

and $c_2 > 0$ being another constant for a bounded response. In the same way the upper extreme of

the cable is then tensed conveniently by selec

with
 $\int_0^\infty |\delta \mathbf{F}^L| dt = c_2$ (38)

and $c_2 > 0$ being another constant for a bounded response. In the same way the upper extreme of

the cable is then tensed conveniently by selecting the function $\delta \mathbf{F}^L(\ell)$. Thus $\int_0^\infty |\hat{\Lambda}vL| dt = c_2$ (38)
 $c_2 > 0$ being another constant for a bounded response. In the same way the upper extreme of

cable is then tensed conveniently by selecting the function $\delta v \cdot L(t)$. Thus the energy deployed
 and $c_2 > 0$ being another constant for a bounded response. In the same way the upper extreme of
the cable is then tensed conveniently by selecting the function $\partial \mathbf{R} \cdot L(t)$. Thus the energy deployed
by the cable force Another general requisite in the design, is that the use of high derivatives would be avoided to the calle is then tensed conveniently by selecting the function $\delta \text{Re} L(t)$. Thus the energy deployed by the calle force co for the able is calculated controllers for damping down a spurious is finite.

On the other side, since forces (30) and (32) are involved in the ROV dynamics, the ROV

velocity controller can compensate these nonlineariti

B. Design of a ROV velocity controller
In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference
controller with a tunable reference dynamics. It is proposed a realizable control l B. Design of a ROV velocity controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference dynamics. It is proposed a realizable control B. Design of a ROV velocity controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference dynamics. It is proposed a realizable control B. Design of a ROV velocity controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference dynamics. It is proposed a realizable control B. Design of a ROV velocity controller

In order to reach a high-quality controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference B. Design of a ROV velocity controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference dynamics. It is proposed a realizable control B. Design of a ROV velocity controller

In order to reach a high-quality control of the ROV kinematics, we focus the design of a reference

controller with a tunable reference dynamics. It is proposed a realizable control -quality control of the ROV kinematics, we focus the design of a reference
be reference oble reference synamics. It is proposed a realizable control law which be
velocity Red to track an auxiliary velocity Red_m, which i controller with a tunable reference dynamics. It is proposed a realizable control law which be
able to force the ROV velocity Red to track an auxiliary velocity Red_{ro}, which is the output of a
reference dynamics given b

$$
\text{Pcd}_m = \frac{\beta_0}{\Lambda(s)} \text{Pcd}_{ref},\tag{39}
$$

Denoting $v = u_t |u_t|$ as the auxiliary control action of the thrus

K^t s² + ¹s + ⁰ pccd+ + D² 8K^t s² + ¹s + ⁰ ^C^D (pccd) ^pccd [|]pccd[|] s² + ¹s + ⁰ 1 K^t F^c ⁰ m ^D³ 6 K^t g . (40) The last equation manifests a di erential relation of third order with a high degree of nonlin-(s) = ^s³ ⁺ ²s² ⁺ ¹^s ⁺ ⁰, (41)

Denoting $v = u_t |u_t|$ as the auxiliary control action of the thrusters and taking (9)-(10) and

(32) into account, one gets a basic equation of the system dynamics for controller design
 $v = \frac{m + m_{\infty}}{K_t} (s^2 + \gamma_1 s + \gamma_0) \cot$ (32) into account, one gets a basic equation of the system dynamics for controller design
 $v = \frac{m + m_{\infty}}{K_t} (s^2 + \gamma_1 s + \gamma_0) \text{Re}d +$
 $+ \frac{\pi \rho D^2}{8K_t} (s^2 + \gamma_1 s + \gamma_0) C_D (\text{Re}d) \text{Re}d [\text{Re}d] -$
 $- (s^2 + \gamma_1 s + \gamma_0) \frac{1}{K_t} F_r$ $+\frac{\pi \rho D^2}{8K_t} \left(s^2 + \gamma_1 s + \gamma_0\right) C_D \text{ (bcd) } \text{Pcd} \left| \text{Red} \right| - \\[2mm] \hspace{1.5mm} - \left(s^2 + \gamma_1 s + \gamma_0\right) \frac{1}{K_t} F_c - \gamma_0 \frac{m - \frac{\pi \rho D^3}{6}}{K_t} g \ . \end{1}$ The last equation manifests a differential relation of third order with a
earity $+\frac{\pi\rho D^2}{8K_t}(s^2 + \gamma_1 s + \gamma_0)C_D$ (keal) keal [keal] –
 $-(s^2 + \gamma_1 s + \gamma_0)\frac{1}{K_t}F_c - \gamma_0\frac{m - \frac{\pi\rho D^2}{K_t}}{K_t}g$. (40)

(equation manifests a differential relation of third order with a high degree of nonlin-

cen Red and $-(s^2 + \gamma_1 s + \gamma_0) \frac{1}{K_t} F_c - \gamma_0 \frac{m - \frac{\eta (D^2)}{6}}{K_t} g$. (40)
The last equation manifests a differential relation of third order with a high degree of nonlinity between ited and v. So the order of $\Lambda(s)$ has to be three i

$$
\Lambda(s) = s^3 + \alpha_2 s^2 + \alpha_1 s + \alpha_0,\tag{41}
$$

a suitable control law have to combined similar linear and nonlinear terms in (40) according to
\n
$$
v = \frac{\theta_1}{\Lambda_f} v + \frac{\theta_2}{\Lambda_f} \text{Ric}d + \frac{\theta_3}{\Lambda_f} F_v + \frac{\theta_4}{\Lambda_f} F_c + \frac{\theta_5}{\Lambda_f} \dot{F}_c + \frac{\theta_6}{\Lambda_f} \dot{F}_v + \frac{\theta_7}{\Lambda_f} \dot{F}_c + \frac{\theta_8}{\Lambda_f} \ddot{F}_c + \frac{\theta_9}{\Lambda_f} \ddot{F}_c + \theta_{13} \text{Ric}f + \theta_{14} \text{Rc}F_v + \theta_{13} \text{Rc}F_c + \theta_{15} \text{Rc}F_c + \theta_{17} \text{Rc}F_c + \theta_{18} \text{Rc}f_{v,f},
$$
\n(42)
\nwhere θ_i are the controller coefficients, $F_v = -C_D$ (Rzd) Rzd |Rzd | and Λ_f is an adjustable Hurwitz
\npolynomial, for instance, of the simple form
\n
$$
\Lambda_f = s + a_0,
$$
\n(43)
\nwhose minimal order helps to minimize the number of θ_i necessary to achieve the objective.
\nHence the control action is obtained through the inverse relation
\n $u_t = \text{sign}(v) \sqrt{|v|},$ \n(44)
\nsubject to saturation according to (32).

 $\label{eq:4.1} \begin{array}{l} \displaystyle +\theta_{13}\hbox{Re}F_{\rm e}+\theta_{17}\hbox{Re}F_{\rm e}+\vspace{10pt}\\ \displaystyle +\theta_{18}+\theta_{19}\hbox{Re}J_{\rm eef}, \end{array} \tag {42}$ $\displaystyle \begin{array}{l} \displaystyle \text{area of,} \hfill \begin{array}{l} \displaystyle \text{Area of,} \hfill \text{Area of,} \hfill \end{array} \begin{array}{l} \displaystyle \text{Area of,} \hfill \text{Area of,} \hfill \end{array} \begin{array}{l} \displaystyle \text{Area of,} \hfill \text{$ + θ_{19} Ped_{ref}, (42)
 $F_v = -C_D$ (Ped) Ped |Ped| and Λ_f is an adjustable Hurwitz

form
 $\Lambda_f = s + a_0$, (43)

the number of θ_i necessary to achieve the objective.

1 through the inverse relation
 $u_t = \text{sign}(v)\sqrt{|v|}$, where θ_i are the controller coefficients, $F_v = -C_D$ (levd) levd | levd| and Λ_f is an adjustable Hurwitz
polynomial, for instance, of the simple form
 $\Lambda_f - s + a_0$, (43)
whose minimal order helps to minimize the number

$$
\Lambda_f = s + a_0,\tag{43}
$$

$$
u_t = \text{sign}(v)\sqrt{|v|},\tag{44}
$$

From (42) and using
$$
\Lambda_f \Re d_{ref} = \Lambda_f \Lambda \Re d/\beta_0
$$
 one gets
\n
$$
(\Lambda_f - \theta_1) v = \theta_2 \Re d + \theta_3 F_v + \theta_4 F_c +
$$
\n
$$
+ \theta_3 \Re d + \theta_6 \Re c F_v + \theta_7 \Re c F_c
$$
\n
$$
+ \theta_8 \Re c F_v + \theta_9 \Re c F_c +
$$
\n
$$
+ \theta_{10} \Lambda_f \Re d + \theta_{11} \Lambda_f F_v + \theta_{12} \Lambda_f F_c +
$$
\n
$$
+ \theta_{13} \Lambda_f \Re d + \theta_{14} \Lambda_f \Re c F_v + \theta_{15} \Lambda_f \Re c F_c +
$$
\n
$$
+ \theta_{16} \Lambda_f \Re c F_v + \theta_{17} \Lambda_f \Re c F_c +
$$
\n
$$
+ \theta_{18} \alpha_0 +
$$
\n
$$
+ \theta_{19} \Lambda_f \Lambda \Re d/\beta_0,
$$
\n(45)
\nand with (40) one achieves
\n
$$
(\Lambda_f - \theta_1) v = \frac{m + m_{\infty}}{K_t} (\Lambda_f - \theta_1) (s^3 + \gamma_1 s^2 + \gamma_0 s) \Re d +
$$
\n
$$
+ \frac{\pi \rho D^2}{8K_t} (\Lambda_f - \theta_1) (s^2 + \gamma_1 s + \gamma_0) F_v -
$$
\n
$$
- (\Lambda_f - \theta_1) \left(\frac{1}{K_t} s^2 + \frac{\gamma_1}{K_t} s + \frac{\gamma_0}{K_t} \right) F_c -
$$

$$
+\theta_{19}\Lambda_f\Lambda \mathbf{R}d/\beta_0, \qquad (45)
$$

\nwith (40) one achieves
\n
$$
(\Lambda_f \cdot \theta_1) v = \frac{m+m_{\infty}}{K_t} (\Lambda_f \cdot \theta_1) \left(s^3 + \gamma_1 s^2 + \gamma_0 s\right) \mathbf{R}d +
$$
\n
$$
+\frac{\pi \rho D^2}{8K_t} (\Lambda_f \cdot \theta_1) \left(s^2 + \gamma_1 s + \gamma_0\right) F_v -
$$
\n
$$
-(\Lambda_f - \theta_1) \left(\frac{1}{K_t} s^2 + \frac{\gamma_1}{K_t} s + \frac{\gamma_0}{K_t}\right) F_c -
$$
\n
$$
-\gamma_0 \frac{m - \frac{\pi \rho D^2}{K_t}}{K_t} g \left(a_0 - \theta_1\right). \qquad (46)
$$
\nEqualing both last expressions one obtains a set of four equations to determine the controller
\nfficients θ_i 's, namely:

coefficients θ_i 's, namely:

1) a relation associated to a polynomial in
$$
Red
$$

\n
$$
\begin{bmatrix}\n\frac{m+m_{\text{max}}}{K_t} (\gamma_1 + a_0) \\
\frac{m+m_{\text{max}}}{K_t} (\gamma_0 + a_0 \gamma_1) \\
\frac{m+m_{\text{max}}}{K_t} a_0 \gamma_0\n\end{bmatrix} = (47)
$$
\n
$$
\begin{bmatrix}\n0 & 0 & 0 & 0 & 0 & \frac{1}{\beta_0} \\
\frac{m+m_{\text{max}}}{K_t} a_0 \gamma_0 \\
0\n\end{bmatrix}
$$
\n
$$
\begin{bmatrix}\n0 & 0 & 0 & 0 & 0 & \frac{1}{\beta_0} \\
\frac{m+m_{\text{max}}}{K_t} & 0 & 0 & 0 & 0 & \frac{\alpha_2 + a_0}{\beta_0} \\
\frac{(m+m_{\text{max}})m_0}{K_t} & 0 & 0 & 1 & 1 & a_0 & \frac{\alpha_0 + a_0 \alpha_1}{\beta_0} \\
\frac{(m+m_{\text{max}})m_0}{K_t} & 0 & 1 & 1 & a_0 & \frac{\alpha_0 + a_0 \alpha_1}{\beta_0} \\
0 & 1 & 0 & a_0 & 0 & \frac{\alpha_0 \alpha_0}{\beta_0}\n\end{bmatrix}
$$
\n
$$
= 2
$$
\n2) a relation associated to a polynomial in F_v

$$
\begin{bmatrix}\n\frac{m+m_{\infty}}{K_{t}}(\gamma_{0} + a_{0}\gamma_{1}) \\
\frac{m+m_{\infty}}{K_{t}}a_{0}\gamma_{0} \\
0\n\end{bmatrix} = (47)
$$
\n
$$
\begin{bmatrix}\n0 & 0 & 0 & 0 & \frac{1}{\beta_{0}} \\
\frac{m+m_{\infty}}{K_{t}} & 0 & 0 & 0 & \frac{2+aa}{\beta_{0}} \\
\frac{(m+m_{\infty})\gamma_{1}}{K_{t}} & 0 & 0 & 0 & 1 & \frac{a_{1}+a_{0}\alpha_{2}}{\beta_{0}} \\
\frac{(m+m_{\infty})\gamma_{0}}{K_{t}} & 0 & 1 & 1 & a_{0} & \frac{a_{0}+a_{0}\alpha_{1}}{\beta_{0}} \\
0 & 1 & 0 & a_{0} & 0 & \frac{a_{0}\alpha_{0}}{\beta_{0}}\n\end{bmatrix} \begin{bmatrix}\n\theta_{1} \\
\theta_{2} \\
\theta_{3} \\
\theta_{10} \\
\theta_{13}\n\end{bmatrix},
$$
\n
$$
\begin{bmatrix}\n\frac{m_{1}D^{2}}{K_{t}} \\
0 \\
\frac{m_{2}D^{2}}{8K_{t}}(\gamma_{1} + a_{0}) \\
\frac{m_{2}D^{2}}{8K_{t}}(\gamma_{0} + a_{0}\gamma_{1})\n\end{bmatrix} = (48)
$$

$$
\begin{bmatrix}\n\frac{m+m_{\infty}}{K_{t}} & 0 & 0 & 0 & 0 & \frac{\alpha_{2}+a_{0}}{\beta_{0}} \\
\frac{(m+m_{\infty})\gamma_{1}}{K_{t}} & 0 & 0 & 0 & 1 & \frac{\alpha_{1}+a_{0}\alpha_{2}}{\beta_{0}} \\
\frac{(m+m_{\infty})\gamma_{0}}{K_{t}} & 0 & 1 & 1 & a_{0} & \frac{a_{0}\alpha_{0}}{\beta_{0}} \\
0 & 1 & 0 & a_{0} & 0 & \frac{a_{0}\alpha_{0}}{\beta_{0}}\n\end{bmatrix}\n\begin{bmatrix}\n\theta_{1} \\
\theta_{10} \\
\theta_{10} \\
\theta_{13}\n\end{bmatrix},
$$
\n
$$
\begin{bmatrix}\n\frac{\pi\rho D^{2}}{8K_{t}} & 0 & 1 & 1 & a_{0} & \frac{\alpha_{0}\alpha_{0}}{\beta_{0}} \\
\frac{\pi\rho D^{2}}{8K_{t}} & (\gamma_{1}+a_{0}) \\
\frac{\pi\rho D^{2}}{8K_{t}} & (\gamma_{0}+a_{0}\gamma_{1}) \\
\frac{\pi\rho D^{2}}{8K_{t}} & 0 & 0 & 1 & 0 & 1 & a_{0} \\
\frac{\pi\rho D^{2}}{8K_{t}} & 0 & 0 & 1 & 0 & 1 & a_{0} \\
\frac{\pi\rho D^{2}}{8K_{t}} & 0 & 1 & 0 & 1 & a_{0} & 0 \\
\frac{\pi\rho D^{2}}{8K_{t}} & \gamma_{0} & 1 & 0 & 0 & a_{0} & 0 & 0\n\end{bmatrix}\n\begin{bmatrix}\n\theta_{1} \\
\theta_{2} \\
\theta_{3} \\
\theta_{4} \\
\theta_{5} \\
\theta_{5} \\
\theta_{6} \\
\theta_{7} \\
\theta_{8} \\
\theta_{9} \\
\theta_{10}\n\end{bmatrix},
$$
\n(48)

3) a relation associated to a polynomial in
$$
F_c
$$

\n
$$
\begin{bmatrix}\n-\frac{1}{Kc} \\
-\frac{2\pi + \alpha_2}{Kc} \\
-\frac{\alpha_2\alpha_3}{Kc}\n\end{bmatrix} = \n\begin{bmatrix}\n0 & 0 & 0 & 0 & 0 & 1 \\
-\frac{4\pi + \alpha_2}{Kc} & -\frac{\alpha_3\alpha_3}{Kc}\n\end{bmatrix}
$$
\n
$$
= \begin{bmatrix}\n0 & 0 & 0 & 0 & 0 & 0 & 1 \\
-\frac{1}{K_1} & 0 & 0 & 1 & 0 & 1 & a_0 \\
-\frac{1}{K_2} & 0 & 1 & 0 & 1 & a_0 & 0 \\
-\frac{2}{K_1} & 1 & 0 & 0 & a_0 & 0 & 0\n\end{bmatrix} \begin{bmatrix}\n\theta_1 \\
\theta_2 \\
\theta_3 \\
\theta_{12} \\
\theta_{13}\n\end{bmatrix},
$$
\n4) a relation associated to the independent term\n
$$
\begin{bmatrix}\n-\frac{\gamma_0 - \frac{\pi \rho^2 \theta^3}{K_1}}{K_1}g a_0\n\end{bmatrix} = \begin{bmatrix}\n-\frac{\gamma_0 - \frac{\pi \rho^2 \theta^3}{K_1}}{K_1}g a_0\n\end{bmatrix} = \begin{bmatrix}\n-\frac{\gamma_0 - \frac{\pi \rho^2 \theta^3}{K_1}}{K_1}g a_0\n\end{bmatrix} \begin{bmatrix}\n\theta_1 \\
\theta_{18} \\
\theta_{18}\n\end{bmatrix}.
$$
\n
$$
= \begin{bmatrix}\n-\frac{\gamma_0 - \frac{\pi \rho^2 \theta^3}{K_1}}{K_1}g a_0\n\end{bmatrix} \begin{bmatrix}\n\theta_1 \\
\theta_3\n\end{bmatrix}.
$$
\n
$$
(50)
$$
\nAs seen in (47)-(50), there exist more unknowns than equations for the identification of the
\nfficients θ_i 's. Eq. (47) describes an overparametrized system with one free parameter and five
\nnorows. Similarly, (48) and (49) have three free parameters and four unknowns each one, and

$$
\left[\begin{array}{c}\n-\gamma_0 \frac{m - \frac{\pi \rho D^3}{6}}{K_t} g a_0\n\end{array}\right] =\n\left[\begin{array}{c}\n-\gamma_0 \frac{m - \frac{\pi \rho D^3}{6}}{K_t} g a_0\n\end{array}\right] \left[\begin{array}{c}\n\theta_1 \\
\theta_{18}\n\end{array}\right].
$$
\n(50)

($-\frac{3\mu}{K_f}$ 1 0 0 a_0 0 0 θ_1)

4) a relation associated to the independent term

($-\gamma_0 \frac{m - \frac{\pi R_f^{D^2}}{K_f}}{\frac{m}{K_f}} g a_0$) =
 $-\left[-\gamma_0 \frac{m - \frac{\pi R_f^{D^2}}{K_f}}{K_f} g a_0\right] \begin{bmatrix} \theta_1 \\ \theta_{18} \end{bmatrix}$. (50)

As seen in (47)-(coefficients θ_i 's. Eq. (47) describes an overparametrized system with one free parameter and five on associated to the independent term
 $\left[-\gamma_0 \frac{m-3\omega^2}{K_1} g a_0\right] =$
 $=\left[-\gamma_0 \frac{m-3\omega^2}{K_3} g a_0\right] \left[\begin{array}{c} \theta_1 \\ \theta_{18} \end{array}\right]$. (50)

(47)-(50), there exist more unknowns than equations for the identification of the
 4) a relation associated to the independent term
 $\left[-\gamma_0 \frac{m - 2d\rho^3}{K_1} g a_0\right] =$
 $-\left[-\gamma_0 \frac{m - 2d\rho^3}{K_1} g a_0\right] \begin{bmatrix} \theta_1 \\ \theta_{18} \end{bmatrix}$. (50)

As seen in (47)-(50), there exist more unknowns than equations for the (50) As seen in (47)-(50), there exist more unknowns than equations for the identification of the coefficients θ_i 's. Eq. (47) describes an overparametrized system with one free parameter and five coefficients θ_i 's. $\left[\begin{array}{c} -\gamma_0 \frac{m-2\epsilon D^3}{K_1} g a_0 \end{array} \right] = \nonumber \\ \left[\begin{array}{c} -\gamma_0 \frac{m-2\epsilon D^3}{K_1} g a_0 \end{array} \right] \left[\begin{array}{c} \theta_1 \\ \theta_{18} \end{array} \right]. \eqno{(50)}$ As seen in (47)-(50), there exist more unknowns than equations for the identification of th $\begin{bmatrix} 1 & 0 \ -\gamma_0 \frac{m - \mu_0 p_0^2}{K_1 - g} g & a_0 \end{bmatrix} \begin{bmatrix} \theta_1 \\ \theta_{18} \\ \theta_{18} \end{bmatrix}$. (50)
As seen in (47)-(50), there exist more unknowns than equations for the identification of the
flicients θ_i 's. Eq. (47) describes an $-\left[-\frac{\gamma_0 - \frac{r_0 - r_0}{K_1}}{K_2} g a_0\right] \left[\frac{1}{\theta_{18}}\right]$. (50)

As seen in (47)-(50), there exist more unknowns than equations for the identification of the
coefficients θ_i 's. Eq. (47) describes an overparametrized syst As seen in (47)-(50), there exist more unknowns than equations for the identification of the coefficients θ_i 's. Eq. (47) describes an overparametrized system with one free parameter and five unknowns. Similarly, (48) a

free parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
respectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the free parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49), respectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
pectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the other si free parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
respectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the free parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
respectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the coefficients.

parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
occtively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be climinated due to singularity.
On the other si free parameters each one. It is observing that θ_{16} and θ_{17} are irremovable in (48) and (49),
respectively, and that the pairs $\{\theta_8, \theta_{14}\}$ and $\{\theta_{15}, \theta_9\}$ can not be eliminated due to singularity.
On the From (42) it is seen that for all derivatives and flux explored variables equal to singularity.

On the other side, the parameters to be potentially eliminated has to be just those involved

in terms with high derivatives configuration, that there is
the parameters to be potentially eliminated has to be just those involved
in terms with high derivatives. Under this criterion, for instance, θ_5 , θ_6 , θ_7 , θ_8 and θ_9 can
be e der this criterion, for instance, θ_5 , θ_6 , θ_7 , θ_8 and θ_9 can
to a minimal and optimal configuration of the controller
ment of damping down transients when the controller starts
This can be achieved, for minimal and optimal configuration of the controller
of damping down transients when the controller starts
can be achieved, for instance, by imposing $\nu(0) = 0$.
and filtered variables equal to zero at $t = 0$, it emerges
.

$$
v(0) = \theta_{12} F_c(0) + \theta_{18}
$$
\n
$$
= 0
$$
\n(51)

coefficients.

Besides, there exists a last requirement of damping down transients when the controller starts

at $t = 0$ from an equilibrium point. This can be achieved, for instance, by imposing $\nu(0) = 0$.

From (42) it Besides, there exists a last requirement of damping down transients when the controller starts

at $t = 0$ from an equilibrium point. This can be achieved, for instance, by imposing $\nu(0) = 0$.

From (42) it is seen that f at $t = 0$ from an equilibrium point. This can be achieved, for instance, by imposing $\nu(0) = 0$.
From (42) it is seen that for all derivatives and filtered variables equal to zero at $t = 0$, it emerges
another condition b From (42) it is seen that for all derivatives and filtered variables equal to zero at $t = 0$, it emerges
another condition between θ_{12} and θ_{18} , e.g.,
 $v(0) = \theta_{12} F_e(0) + \theta_{18}$ (51)
 $= 0$
So, from the set of redu another condition between θ_{12} and θ_{18} , e.g.,
 $v(0) = \theta_{12} F_c(0) + \theta_{18}$
 $= 0$

So, from the set of redundant parameters $\{\theta_5, \theta_6, \theta_7, \theta_8, \theta_9\}$ one chooses one of them to ace

minimal-set design and long- $\begin{array}{lcl} v(0)&=&\theta_{12}\,F_{\rm e}(0)+\theta_{18} \end{array} \eqno{(51)}$ $= &\ 0 \qquad \qquad \eqno{(51)}$ from the set of redundant parameters
 $\{\theta_5,\theta_8,\theta_7,\theta_8,\theta_9\}$ one chooses one of them to accomplish nimal-set design and long-term transient eliminati

In this way
$$
v(0) = 0
$$
.
\nBearing this reasoning in mind, one concludes that the minimal-set selection yielding to
\n
$$
v = \frac{\theta_1}{\Lambda_f} v + \frac{\theta_2}{\Lambda_f} \text{Re}d + \frac{\theta_3}{\Lambda_f} F_v + \frac{\theta_4}{\Lambda_f} F_e + \frac{\theta_7}{\Lambda_f} F_c + \frac{\theta_7}{\Lambda_f} F_c + \theta_{10} \text{Re}d + \theta_{11} F_v + \theta_{12} F_c + \theta_{13} \text{Re}d + \theta_{14} \text{Re}F_v + \theta_{15} \text{Re}F_c + \theta_{16} \text{Re}F_v + \theta_{17} \text{Re}F_c + \theta_{18} + \theta_{19} \text{Re}d_{ref}
$$
\n
$$
+ \theta_{18} + \theta_{19} \text{Re}d_{ref}
$$
\n
$$
(52)
$$
\nis quite suitable. Then the control action results from (44) with (52) and saturations given in
\n45

(32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of F_v and F_c , since these are commonly not measurable. A nonlinear observer for this purpose escribed in Jordán and Bustamante [15 (32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of
 Red, F_v and F_c , since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán (32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of

Bed, F_v and F_c , since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán

(32).

It is worth noticing the necessity of employing an observer to obtain hierd, F_v and F_e , since these are commonly not measurable. A nonlinear observer

is described in Jordán and Bustamante [15].

C. Force contr (32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of

Red, F_v and F_v since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán a (32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of

Red, F_c and F_c , since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán (32).

It is worth noticing the necessity of employing an observer to obtain high derivatives of

Red, F_c and F_c , since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán It is worth noticing the necessity of employing an observer to obtain high derivatives of

Red, F_c and F_c , since these are commonly not measurable. A nonlinear observer for this purpose

is described in Jordán and Bus ted, F_c and F_m since these are commonly not measurable. A nonlinear observer for this purpose
is described in Jordán and Bustamante [15].
C. Force controllers
The cable strength is controlled from the upper and lower Thus rolled from the upper and lower extremes of the cable according to the

. (28). From (35) one sees that the lower extreme of the cable can be

urbation $\delta \Re z d(t)$ about the equilibrium point of the cable force defined

no is eable strength is controlled from the upper and lower extremes of the cable according to the extreme proposed in Fig. (28). From (35) one sees that the lower extreme of the cable, can be seed by defining a perturbation structure proposed in Fig. (28). From (35) one sees that the lower extreme of the cable can be
tensed by defining a perturbation $\delta \text{Re} d(t)$ about the equilibrium point of the cable lorce defined
by (34). Taking also the urbation $\delta \Re d(t)$ about the equilibrium point of the cable force defined
nonlinearity (31) into account, a PD controller will be sufficient able
r, its gain has to be variable to compensate the cable length changes.
 $\Re d$

$$
\delta \mathbf{P}cd(t) = (d(t) + b) (K_{P_1} + K_{D_1} s) (F_{c_{ref}} - F_c).
$$
\n(53)

$$
\delta \mathbf{R} L(t) = (d(t) + b) (K_{P_2} + sK_{D_2}) (F_{c_{ref}} - F_c).
$$
\n(54)

Figures 1992.

The Poisson and Alexandre is spain has to be variable to compensate the cable length changes.

Similarly, using (37) and (31) for the upper extreme of the cable, the crane motor will be

thirded by acting d Thus
 $\delta \text{Re}d(t) = (d(t) + b) (K_{P_1} + K_{D_1}s) (F_{c_{\gamma q}} - F_c)$. (53)

Similarly, using (37) and (31) for the upper extreme of the cable, the crane motor will be

perturbed by acting directly on its voltage by means of another PD $\delta \text{Red}(t) = (d(t) + b) (K_{P_1} + K_{D_1}s) (F_{c_{ref}} - F_c)$. (53)

Similarly, using (37) and (31) for the upper extreme of the cable, the crane motor will be

perturbed by acting directly on its voltage by means of another PD controlle Similarly, using (37) and (31) for the upper extreme of the cable, the crane motor will be
perturbed by acting directly on its voltage by means of another PD controller which generates
 $\delta \Re(L(t) - (d(t) + b) (K_{P_2} + sK_{D_2}) (P_{e_{$ Similarly, using (37) and (31) for the upper extreme of the cable, the crane motor will be
perturbed by acting directly on its voltage by means of another PD controller which generates
 $\delta \text{Re} L(t) = (d(t) + b) (K_{P_2} + sK_{D_2}) (F$ perturbed by acting directly on its voltage by means of another PD controller which generates
 $\delta \text{Re} L(t) = (d(t) + b) (K_{P_2} + sK_{D_2}) (P_{\text{reg}} - F_c)$. (54)

The PD controller parameters in both cases are set constant for a desir $\delta \Re r L(t) = (d(t) + \hbar) (K_{P_2} + s K_{D_2}) (F_{c_{ref}} - F_c). \eqno{(54)}$ The PD controller parameters in both cases are set constant for a desired behavior of the eable tension. The tuning of these 4 coefficients is performed simultaneously The PD controller parameters in both cases are set constant for a desired behavior of the cable tension. The tuning of these 4 coefficients is performed simultaneously by numerically optimizing a quadratic cost functional

$$
u_{cr}(t) = (K_{P_3} + sK_{D_3})(L_{ref} - L). \tag{55}
$$

Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) with equation
 $u_{cr}(t) = (K_{P_3} + sK_{D_3}) (L_{ref} - L).$ (55)

The controller coefficients are tuned in the control loop of the hoisting system separ er is applied for the hoisting crane (see Fig. (28)) with equation
 $u_{cr}(t) = (K_{P_3} + sK_{D_3}) (L_{ref} - L).$ (55)

re tuned in the control loop of the hoisting system separately from

cal-ROV system, taking the model in (33) with oisting crane (see Fig. (28)) with equation

) $(L_{ref} - L)$. (55)

l loop of the hoisting system separately from

ng the model in (33) with saturations into Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) with equation
 $u_{cr}(t) = (K_{Ps} + sK_{Ds}) (L_{ref} - L).$ (55)

The controller coefficients are tuned in the control loop of the hoisting system separatel Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) with equation
 $u_{cr}(t) = (K_{P_3} + sK_{D_3}) (L_{ref} - L).$ (55)

The controller coefficients are tuned in the control loop of the hoisting system separ Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) wit
 $u_{cr}(t) = (K_{P_3} + sK_{D_3}) (L_{ref} - L)$.

The controller coefficients are tuned in the control loop of the hoisting system sepa

the controll Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) with equation
 $u_{cr}(t) = (K_{P_3} + sK_{D_3})(L_{ref} - L).$ (3)

The controller coefficients are tuned in the control loop of the hoisting system separat

Finally, a fixed PD controller is applied for the hoisting crane (see Fig. (28)) with equation				
	$u_{cr}(t) = (K_{P_3} + sK_{D_3}) (L_{ref} - L).$			(55)
The controller coefficients are tuned in the control loop of the hoisting system separately from				
the control loop of the umbilical-ROV system, taking the model in (33) with saturations into				
account for this purpose.				
D. Summary of control components				
The components of the controlled umbilical-ROV system are summarized in table (3).				
$Control$ components $Input(s)$			Output(s) Eq. number	
$\begin{array}{c c}\n\text{Control compontence} & \text{---} \\ \hline\n\end{array}\n\qquad\n\begin{array}{c}\nL \\ w_t\n\end{array}\n\qquad\n\begin{array}{c}\nF_c \\ \text{Red} \\ \end{array}\n\qquad\n\begin{array}{c}\n(4), (5) \\ (9), (10) \\ (17), (18)\n\end{array}\n\end{array}$				
Propulsion system	u_t	F_t	(32)	
Crane	u_{cr}	L	(33)	
	Pcd_{ref}			(56)
Cinematic controller $% \left\vert \cdot \right\rangle$	Pcd	u_t	(44), (52)	
	\mathcal{F}_c			
Reference model	$\mathbf{P}cd_{ref}$	$\mathbf{P}cd_m$	(39), (41)	
Force controller 1	$F_{c_{ref}} - F_c$	$\delta P C L$	(54)	
Force controller 2	$\mathcal{F}_{c_{ref}} - \mathcal{F}_c$	$\delta \mathbf{P}cd$	(53)	
Crane controller	$L_{ref} - L$	$\boldsymbol{u_{cr}}$	(55)	
	Table 3: Control system components			
In order to simulate the controlled umbilical-ROV system in a wide range of heave operations,				
	47			

System component Reference model	Coefficient set	Values in S.I. units	
	$\{K_m, \alpha_2, \alpha_1, \alpha_0\}$	$\{6.498, 4.80, 9.01, 6.498\}$	
		$\{\theta_1, \theta_2, \theta_3, \theta_4, \theta_7, \}$ $\{-3.80, -767.78, 359.89, -1.96 \times 10^{-13}, 0.89,$	
Kinematic controller	$\theta_{10}, \theta_{11}, \theta_{12}, \theta_{13}, \theta_{14},$	$111.32, 94.70, -1.12, -708.17, 227.30,$	
	$\{\theta_{15}, \theta_{16}, \theta_{17}, \theta_{18}, \theta_{19}\}$	$-0.56, 47.35, -0.11, -594.42, 656.45\}$	
Hoisting crane motor	$\left\{ k_{1}, \frac{L_{a}J}{k_{1}k_{2}}, \frac{R_{a}J}{k_{1}k_{2}} \right\}$	${0.015, 0, 5}$	
Force controller 1	$\{K_{P_1}, K_{D_1}\}\$	$\{0.0016, 4.5 \times 10^{-14}\}$	
Force controller 2	${K_{P_2}, K_{D_2}}$	$\{0.0021, 0.0034\}$	
Crane controller	${K_{P_3}, K_{D_3}}$	${1700.0, 1320.0}$	
Propulsion system	$\{K_t,\gamma_1,\gamma_0\}$	${8.5, 1, 2}$	
Thruster voltage saturation	${u_{t\min}, u_{t\min}}$	$\{-12, 12\}$	
Crane voltage saturation	${u_{cr\min}, u_{cr\min}\}$	$\{-110, 110\}$	
Umbilical cable	${E A_0, b}$	$\{10^6, 3\}$	
ROV dynamics	$\{D,m\}$	$\{1, 590.36\}$	
Hydrodynamics	${m_\infty, \rho, C_D(\text{Re})}$	${268.35, 1025, Eq. (7)}$	

(57)

 $\{m_\infty, \rho, C_D(\text{Re})\} \qquad \text{ \{268.35, 1025, Eq. (7)} \tag{57} \label{67}$ Table 4: Parameter settings for simulations of the uncontrolled dynamics is the model 2. WI. CONTROL STABILITY The taut-slack control system described in the pre

parameters. As in Sec. IV done, most of these parameters are suitable for a bifurcation study
and to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric
space for determining stability parameters. As in Sec. IV done, most of these parameters are suitable for a bifurcation study
and to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric
space for determining stability parameters. As in Sec. IV done, most of these parameters are suitable for a bifurcation study
and to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric
space for determining stability sot of these parameters are suitable for a bifurcation study

e of the taut-slack phenomenon. Similarly, the parametric

ns is defined here as
 $\boldsymbol{\mu} = [a, \omega, D, EA_0, L]^T$, (58)

e parameter since it is regulated by the ROV is not longer available as free parameters are suitable for a bifurcation study tablish stability regions free of the taut-slack phenomenon. Similarly, the parametric determining stability regions is defined here as $\mu = [$ parameters. As in Sec. IV done, most of these parameters are suitable f
and to establish stability regions free of the taut-slack phenomenon. Sin
space for determining stability regions is defined here as
 $\mu = [a, \omega, D, EA_0,$ ameters. As in Sec. IV done, most of these parameters are suitable for a bifurcation study

to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric

ce for determining stability regions parameters. As in Sec. IV done, most of these parameters are suitable for a bifurcation study
and to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric
space for determining stability

$$
\boldsymbol{\mu} = [a, \omega, D, EA_0, L]^T, \tag{58}
$$

where F_t is not longer available as free parameter since it is regulated by the ROV cinematic and

and to establish stability regions free of the taut-slack phenomenon. Similarly, the parametric
space for determining stability regions is defined here as
 $\mu = [a, \omega, D, EA_0, L]^T$, (58)
where F_t is not longer available as f Fig. (29) depicts the qualitative diversity of behavior that can be produced in the ROV cinematic and
 $\mu = [a, \omega, D, EA_0, L]^T$, (58)

The detection of high periods is carried out in the same way as done in Sec. III. Alike as $\mu = [a, \omega, D, EA_0, L]^T$, (58)
where F_t is not longer available as free parameter since it is regulated by the ROV cinematic and
force controllers.
The detection of high periods is carried out in the same way as done in Sec where F_i is not longer available as free parameter since it is regulated by the ROV cinematic and
force controllers.
The detection of high periods is carried out in the same way as done in Sec. III. Alike as
before, the once ϵ is in comparison with homologous case without control. The detection of high periods is carried out in the same way as done in Sec. III. Alike as before, the zone drawn in shades corresponds to regions where the The detection of high periods is carried out in the same way as done in Sec. III. Alike as
before, the zone drawn in shades corresponds to regions where the cable remains taut, at least
in steady state, for a monochromati significative in canceled or more increment of the stable regions where the cable remains taut, at least
in steady state, for a monochromatic perturbation.
Fig. (29) depicts the qualitative diversity of behavior that can in steady state, for a monochromatic perturbation.

In steady state, for a monochromatic perturbation.

Fig. (29) depicts the qualitative diversity of behavior that can be produced in the heave

operation under the contro solutions. The controlled case, the appearance of high-period orbits and be produced in the heave
operation under the control system with respect to the wave amplitude. In this sense, it is noticed
that the variety of per common. Fig. (30) shows also a significative extension of the stability region and decreased
isole it in comparison with homologous case without control. Additionally one observes that
control imposes a tendency to chaos, however contained in the tendency of parameteristic as before and anti-and outside to chaotic control imposes a tendency to chaos, however with damped energy. It is also appreciated a significative increment of the stable region stability region is registered mainly for small values of the board of the control imposes a tendency to chaoe, however with damped energy. It is also appreciated a significative increment of the stable region L versus a Fig. (31) illustrates the stability in the region are stable region at a mean depth of the stable region and seculated proportant property of results shown in Sec. III was that the stability region is exclusively represen

of the stability region is registered mainly for small wave amplitudes and even chaos is solutions. Now in the controlled case, the appearance of high-period orbits and even chaos is common.
Fig. (30) shows also a signifi The reason because the controlled case, the appearance of high-period orbits and even chaos is common.

Fig. (30) shows also a significative extension of the stability region d versus ω with the same

characteristic section. Fig. (30) shows also a significative extension of the stability region d versus ω with the same characteristic as before. The tendency to chaotic behaviors is found out inside and outside the stability regio

Figure 29: Comparisson of stability regions: ROV depth vs. wave amplitude for $EA_0 = 10^6$ [N]

Figure 30: Comparisson of stability regions: ROV depth vs. wave frequency for $EA_0 = 10^6$ [N]

restrained wave steepness (ωa). The behavior diversity has not changed too much comparatively
inside the stability region.
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
 restrained wave steepness (ωa) . The behavior diversity has not changed too much compainside the stability region.

Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects thence between ROV w Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
ce between ROV weight and volume, a restrained wave steepness (ωa) . The behavior diversity has not changed too much comparatively
inside the stability region.
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
den restrained wave steepness (ωa). The behavior diversity has not changed too much comparatively
inside the stability region.
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
 restrained wave steepness (ωa). The behavior diversity has not changed too much comparatively
inside the stability region.
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
 rained wave steepness (∞a). The behavior diversity has not changed too much comparatively
de the stability region.
Fig. (32) shows the region D versus EA_0 for a given wave and depth. It reflects the depen-
ce betw restrained wave steepness (ωa). The behavior diversity has not changed too much comparatively
inside the stability region. D versus EA₀ for a given wave and depth. It reflects the depen-
dence between ROV weight and Exerpiral (20). The bandinal antihing into antihing the sometime complicating
ty region.

Sity region D versus EA_0 for a given wave and depth. It reflects the depen-

ROV weight and volume, and cable stiffness. Also in D versus EA_0 for a given wave and depth. It reflects the dependent

a volume, and cable stiffness. Also in this case, it was possible to

solity region. It is inferred that a great volume with a relatively

controlled t

 $Pcd = -10 \text{ [m/s}^2$. The attractor shape is very common for other points considered in the study.

In this section and investigation of the stability region. It is inferred that a great volume with a relatively
small cable stiffness is easy to controlled than otherwise.
The appearance of chaotic behaviors is very commo small cable stiffness is easy to controlled than otherwise.

The appearance of chaotic behaviors is very common in the controlled case even in the stability

region. One strange attractor is depicted in Fig. (33) with a c The appearance of chaotic behaviors is very common in the controlled case even in the stability
region. One strange attractor is depicted in Fig. (33) with a cross section of its volume for
 $\text{Red} = -10[\text{m/s}^2]$. The attra **EVALUATE AS FIGURE 20** The Fig. (33) with a cross section of its volume for
eigen. One strange attractor shape is very common for other points considered in the study.
VII. CONTROL PERFORMANCE
In this section an investig In Figs. (34)-(35) and (36)-(37) the profile of the reference of the system in the sindy.

UII. CONTROL PERFORMANCE

this section an investigation of the overall control performance of the system in the sink-

//ifting op VII. CONTROI. PERFORMANCE.

In this section an investigation of the overall control performance of the system in the sinking/lifting operation is presented. The first experiments consist in preseribing a profile of the de VH. CONTROL PERFORMANCE
In this section an investigation of the overall control performance of the system in the sink-
ing/lifting operation is presented. The first experiments consist in preseribing a profile of the
desi In this section an investigation of the overall control performance of the system in the sink-
ing/lifting operation is presented. The first experiments consist in preseribing a profile of the
desired depth to be followed

In Fig. (34) the wave steepness amounts (ω) = 0.275[m rad/s]. The cable for maximal cable strength avoids as far as possible entering the tant-slack region.
In Figs. (34)-(35) and (36)-(37) the profiles of the referenc any many operation is presented. The the value of probability in presenting a particle in the value of desired depth to be followed in the shortest possible time, that cares for maximal cable strength and avoids as far as about the small about this reference included the cable tension.

In Figs. (34)-(35) and (36)-(37) the profiles of the reference L_{ref} are the same and are built

up as ramps for sinking up to 100[m] from a starting dept In Figs. (34)-(35) and (36)-(37) the profiles of the reference L_{ref} are the same and are built
up as ramps for sinking up to 100[m] from a starting depth $L = 15$ [m] $(i.e., d = 12$ [m]), pausing
and lifting again to the same The ROV velocity pcd behaves underdamped during the changes. It is noticed behaves under the same during and lifting again to the same depth at the beginning. The dynamics is subject to different perturbations explained i shown if it
and lifting again to the same depth at the beginning. The dynamics is subject to different
perturbations explained in the next.
In Fig. (34) the wave steepness amounts $(a\omega) = 0.275[m \text{ rad/s}]$. The cable force is

 $\overline{}$

Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m],

Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m],
 $\omega = 4.11$ [rad/s], EA Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m],
 $\omega = 4.11$ [rad/s], $EA_0 = 10^6$ and $L = 50$ [m]. Cross section for te $d = -10$ [m/s^{2]}

oscillations caused by transients of Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m],
 $\omega = 4.11$ [rad/s], $EA_0 = 10^6$ and $L = 50$ [m]. Cross section for $\text{Re}d = -10$ [m/s²]

oscillations caused by transients Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m],
 $\omega = 4.11$ [rad/s], $EA_0 = 10^6$ and $L = 50$ [m]. Cross section for $\text{P}el = -10$ [m/s²]

oscillations caused by transients Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45$ [m], $\omega = 4.11$ [rad/s], $EA_0 = 10^6$ and $L = 50$ [m]. Cross section for $\text{Re}d = -10$ [m/s²] oscillations caused by transients o Figure 33: Construction of a strange attractor for the control system behaviour with $a = 0.45[\text{m}]$,
 $\omega = 4.11[\text{rad/s}]$, $EA_0 = 10^6$ and $L = 50[\text{m}]$. Cross section for $\text{Per}d = -10 [\text{m/s}^2]$

oscillations caused by transie $\omega = 4.11$ [rad/s], $EA_0 = 10^6$ and $L = 50$ [m]. Cross section for $\text{Re}d = -10$ [m/s²]
oscillations caused by transients of the equivalent mass spring system. Fig. (35) shows the
evolution of the control actions on the R oscillations caused by transients of the equivalent mass-spring system. Fig. (35) shows the evolution of the control actions on the ROV thrusters and crane motor, respectively. In the first one, it is perceiving an increm the control of the control actions on the ROV thrusters and crane motor, respectively. In
first one, it is perceiving an increment of the energy of u_t with even a saturation for a sh
time. On the other side, the control The next experiment illustrated in Figs. (36)-(37), exemplifies the control behavior of the control action for the energy of u_k with even a saturation for a short c. On the other side, the control action for the crane m lare one, i. E perfering an interaction of the energy of a mini-ordinal extractions of the chine. On the other side, the control action for the crane motor shows a continuous oscillatory behavior with steps at the break p behavior with steps at the break points of L_{ref} . The frequency of these oscillations correspond
to the wave frequency, which indicates that during the sinking/lifting of the ROV, the crane
motor attempts to follow the w

is to the transient phase is stronger than earlier but less than 20% of the ROV, the crane motor attempts to follow the wave perturbation in order to care for the cable strength and simultaneously diminish the error $(d_{ref}$ motor attempts to follow the wave perturbation in order to care for the cable strength and
simultaneously diminish the error $(d_{ref} - d)$. In summary, the overall achievable performance in
this operation is of high quality.

 $\omega=0.55[\mathrm{rad/s}]$ and $EA_0=10^6[\mathrm{N}]$

motor for $a=0.5[{\rm m}],\,\omega=0.55[{\rm rad/s}]$ and $EA_0=10^6[{\rm N}]$

 $0.55[\text{rad/s}]$ and $EA_0 = 10^6[\text{N}]$

For the case of the called term is nevertheless very good. The evolution of the thruster excitation saturates during
the second intervention of the cable length, cable force and ROV velocity for $a = 0.75$ [m], $\omega = 0.55$ [r The ascent and descent, and turns off in the pause. On the other side, the other big after an ascent and descent, and turns off in the pause. On the other side, the control action for the cancer motor saturates form time Figure 36: Evolution of the cable length, cable force and ROV velocity for $a = 0.75$ [m], $\omega = 0.55$ [rad/s] and $EA_0 = 10^6$ [N]
operation is nevertheless very good. The evolution of the thruster excitation ν saturates du Figure 36: Evolution of the cable length, cable force and ROV velocity for $a = 0.75$ [m], $\omega = 0.55$ [rad/s] and $EA_0 = 10^6$ [N]
operation is nevertheless very good. The evolution of the thruster excitation ν saturates du Figure 36: Evolution of the cable length, cable force and ROV velocity for $a = 0.75$ [m], $\omega = 0.55$ [rad/s] and $EA_0 = 10^6$ [N]
operation is nevertheless very good. The evolution of the thruster excitation ν saturates du Figure 36: Evolution of the cable length, cable force and ROV velocity 1
0.55[rad/s] and $EA_0 = 10^6$ [N]
operation is nevertheless very good. The evolution of the thruster excitatie
the ascent and descent, and turns off in 5[rad/s] and $EA_0 = 10^6$ [N]
ration is nevertheless very good. The evolution of the thruster excitation ν saturates during
ascent and descent, and turns off in the pause. On the other side, the control action for
crane operation is nevertheless very good. The evolution of the thruster excitation ν saturates during
the ascent and descent, and turns off in the pause. On the other side, the control action for
the crane motor saturates f Incometric and descent, and turns off in the parts. On the other side, the control action for
erane motor saturates from time to time, recovering sometimes the low-frequency oscillation
h a wave-shaped appearance. The err

the crane motor saturates from time to time, recovering sometimes the low-frequency oscillation
with a wave-shaped appearance. The error $(L_{ref} - L)$ is mainly perceived in the starting phase,
after an ascent or descent, how with a wave-shaped appearance. The error $(L_{ref} - L)$ is mainly perceived in the starting phase,
after an ascent or descent, however it amounts a maximal value less than 5% of the total change
of the length.
The next couple which whose samples oppositence. The critic ($\mu_{eff} = x$) to manny posteror an the stating prace,
after an ascent or descent, however it amounts a maximal value less than 5% of the total change
of the length.
The next compl by the elongation of the cable. This effect does not appear by the control performance for the creation operation about a fixed depth $d_{ref} = L_{ref} - b = 47$ [m] under wave perturbations.
In the first case, the control variables The next couple of figures (38)-(39) and (40)-(41) show the control performance for the regulation operation about a fixed depth $d_{ref} = L_{ref} - b = 47$ [m] under wave perturbations.
In the first case, the control variables L , Figs. (40)-(41) depict the control performance for a significative larger wave steepness (ω) and $\ln t$ depth $d_{ref} = L_{ref} - b = 47$ [m] under wave perturbations.
In the first case, the control variables L, F_e and $\Re d$ sh

motor for $a=0.75[{\rm m}],\,\omega=0.55[{\rm rad/s}]$ and $EA_0=10^6[{\rm N}]$

 $a=0.45[{\rm m}]$ and frequency $\omega=0.86[{\rm rad/s}]$ and $EA_0=10^6[{\rm N}]$

for a wave amplitude $a = 0.45$ [m] and frequency $\omega = 0.86$ [rad/s] and $EA_0 = 10^6$ [N]

For a wave amplitude a = 0.45[m] and frequency ω = 0.86[rad/s] and EA_0 = 10⁶[N]

Figure 39: Explittion of the cable, length, square tension of the thrusters and tension crime motor

for a wave amplitude a = 0.45[m] Figure 39: Evolution of the cable, length, square tension of the thrusters and tension crane motor
Figure 39: Evolution of the cable, length, square tension of the thrusters and tension crane motor
for a wave amplitude a $\frac{1}{2} \sqrt{\frac{1}{2} \sqrt{\frac{1$ Figure 39: Evolution of the cable, length, square tension of the thrusters and tension crane motor
Figure 39: Evolution of the cable, length, square tension of the thrusters and tension crane motor
for a wave amplitude a **Than 2% of the reference values and the taut-slack phenomenon of the single values and tension crane motor

frame: 30: Evolution of the cable, length, square tension of the thrusters and tension crane motor

for a wave a** mgth, square tension of the thrusters and tension crane motor

and frequency $\omega = 0.86$ [rad/s] and $EA_0 = 10^6$ [N]

5[m rad/s]. In this case the behavior becomes chaotic for all

16 maintaining the cable tense is achieved.

for a wave amplitude $a = 0.45$ [m] and frequency $\omega = 0.86$ [rad/s] and $EA_0 = 10^6$ [N]
than the case before, equal to 0.645[m rad/s]. In this case the behavior becomes chaotic for all
variables, however the control goal of than the case before, equal to 0.645[m rad/s]. In this case the behavior becomes chaotic for all
variables, however the control goal of maintaining the cable tense is achieved. Despite the almost
permanent saturation of t variables, however the control goal of maintaining the cable tense is achieved. Despite the almost
permanent saturation of the control action for the thrusters, the depth and length errors are less
than 2% of the referenc The cable is the cable tension for the christeneously and the capital depth and length errors are less
than 2% of the reference values and the taut-slack phenomenon is quite afar.
VIII. CABLE TENSION
The presence of the t Fig. (42) reproduces the end, some selected scenarios of the ROV operation. The presence of the taut-slack phenomenon during the sinking/lifting operation of the ROV demands a significative stress resistance from the umbi VIII. CABLE TENSION

The presence of the taut-slack phenomenon during the sinking/lifting operation of the ROV

demands a significative stress resistance from the umbilical cable. The rampant rising and large

strengths m VIII. CABLE TENSION
The presence of the taut-slack phenomenon during the sinking/lifting operation of the ROV
demands a significative stress resistance from the umbilical cable. The rampant rising and large
strengths may Eq. (42) reproduces the evolution of the forces for a relation of the ROV represence of the taut-slack phenomenon during the smiking/lifting operation of the ROV reparation significative stress resistance of premature fat The protone of the out start protonnant during the unming proton of the tast.

demands a significative stress resistance from the umbilical cable. The rampant rising and large

strengths may not only be the cause of premat

 $a=0.75[\mathrm{m}]$ and frequency $\omega=0.86[\mathrm{rad/s}]$ and $EA_0=10^6[\mathrm{N}]$

for a wave amplitude $a=0.75[\mathrm{m}]$ and frequency $\omega=0.86[\mathrm{rad/s}]$ and $EA_0=10^6[\mathrm{N}]$

EVALUE AND SET AND SET AND MANUSUPARTIZE AT A CAUTE AND SET AND SET $\frac{2}{\omega}$ $\frac{760}{90}$ $\frac{100}{100}$ $\frac{100}{100}$ to $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{60}$ $\frac{1}{60}$ Figure 42: Cable force comparisson for $a = 0.15$ [m], $\omega = 1, 87$ [md/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
with evidence of hefty hauls of the cable. On the contrary, the controlled system can succe Figure 42: Cable force comparisson for $a = 0.15$ [m], $\omega = 1,87$ [rad/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
with evidence of hefty hauls of the cable. On the contrary, the controlled system can succe without control. Bottom: with control
with evidence of hefty hauls of the cable. On the contrary, the controlled system can successfully
regulate the force about the reference in the tant zone.
The next four figures illus with evidence of hefty hauls of the cable. On the contrary, the controlled system can successfully regulate the force about the reference in the taut zone.
The next four figures illustrate the force evolution for a relati tension of circa 10 times larger than in the controlled case. Figs. (44) and the controlled equal to 0.75[m] and high frequencies, ranging from 0.86[rad/s] till 1.27[rad/s], and lengths in the span starting at $L = 4.12$ [m The next four figures illustrate the force evolution for a relatively large wave amplitude equal
to 0.75[m] and high frequencies, ranging from 0.86[rad/s] till 1.27[rad/s], and lengths in the
span starting at $L = 4,12$ [m] Let note that the control of the control of the control of the cable temperature of the control of the control of the control of the span starting at $L = 4, 12$ [m] up to $L = 50$ [m]. Fig. (43) represents the force progre is an starting at $L = 4, 12$ [ni] up to $L = 50$ [ni]. Fig. (43) represents the force progressing under a wave steepness of $(\omega \omega) = 0.645$ [ni rad/s]. Similarly as before, in the control system the cable remains tense and the and watures at $Z = 1.12 \mu m$ or $Z = 2.0435$ maniform and $|z|$. Similarly as before, in the control system the cable remains tense and the force regulated within a relatively narrow band, while on the other side, the uncon the cable remains tense and the force regulated within a relatively narrow band, while on the other side, the uncontrolled dynamics of the system produces large and stark increments of the tension of circa 10 times larger the state controlled dynamics of the system produces large and stark increments of the tension of circa 10 times larger than in the controlled case. Figs. (44) and (45) characterize a similar situation for an increment of based star), the uncontrolled system. In the controlled case. Figs. (44) and (45) characterize a similar situation for an increment of the wave steepness to $(a\omega) = 0.75[m \text{ rad/s}]$ and two different lengths. It is noticing th

the controller case than in the uncontrolled one.

The control experiments as earlier but with a different south of $L = 14.2$ /m], $L = 47.66$ /m] and $L = 73.80$ /m]. The same experiments of $L = 14.2$ /m], $L = 47.66$ /m] and L $T_{\text{B}} = \frac{1}{2} \begin{bmatrix} 0.00 &$ $\frac{Z}{\omega}$ and $\frac{2000}{400}$ $\frac{1}{100}$ $\frac{1}{100}$ Eigure 44: Cable force comparisson for $a = 0.75$ [m], $\omega = 1$ [rad/s] and $L = 4.12$ [m]. Top: without controll. Bottom: with controll
ethe controller case than in the uncontrolled one.
The next three figures, Figs. (47)-(49) Figure 44: Cable force comparisson for $a = 0.75$ [m], $\omega = 1$ [rad/s] and $L = 4.12$ [m]. Top: without controll. Bottom: with controll
the controller case than in the uncontrolled one.
The next three figures, Figs. (47)-(49), Figure 44: Cable force comparisson for $a = 0.75$ [m], $\omega = 1$ [rad/s] and $L = 4.12$ [m]. Top: without controll. Bottom: with controll
the controller case than in the uncontrolled one.
The next three figures, Figs. (47)-(49), control. Bottom: with control
the controller case than in the uncontrolled one.
The next three figures, Figs. (47)-(49), illustrate the force evolution for a greater wave
amplitude than the previous cases, but with smalle

Figure 50: Cable force comparisson for $a = 1.05$ [m], $\omega = 0.86$ [rad/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
without source. Bottom: with control
Fig. (50) displays an extreme situation where the wave

Figure 50: Cable force comparisson for $a = 1.05$ [m], $\omega = 0.86$ [rad/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
Figure 50: Cable force comparisson for $a = 1.05$ [m], $\omega = 0.86$ [rad/s] and $L = 50$ [m]. Top: Summarizing, in the majority of the experiments, the control system and successfully,
the material space of the majority of the majority of the majority of the majority of the periodic successfully,
ever the operation sta Figure 50: Cable force comparisson for $a = 1.05$ [m], $\omega = 0.86$ [rad/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
Fig. (50) displays an extreme situation where the wave steepness amounts a relatively large Figure 50: Cable force comparisson for $a = 1.05$ [m], $\omega = 0.86$ [rad/s] and $L = 50$ [m]. Top:
without control. Bottom: with control
Fig. (50) displays an extreme situation where the wave steepness amounts a relatively large without control. Bottom: with control

Fig. (50) displays an extreme situation where the wave steepness amounts a relatively large

value of $(a\omega) = 0.903$ [m rad/s] for a middle length. Here the control system works succes

IX. Conclusions
bilical-ROV system under nonlinear oscillations in heave
methods for the uncontrolled and controlled cases compar-
taut-slack phenomenon on the umbilical cable produced by IX. CONCLUSIONS
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the IX. Conclusions
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the IX. CoNCLUSIONS
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the IN this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the appearance of t IX. CONCLUSIONS
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the IX. CONCLUSIONS
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the IX. CoNCLUSIONS
In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the In this paper the stability of an umbilical-ROV system under nonlinear oscillations in heave
motion was analyzed using numerical methods for the uncontrolled and controlled cases compar-
atively. Mainly the appearance of t means analyzed using numerical methods for the uncontrolled and controlled cases comparatively. Mainly the appearance of the taut-slack phenomenon on the umbilical cable produced by the interaction of monochromatic waves w sively. Mainly the appearance of the taut-slack phenomenon on the umbilical cable produced by
the interaction of monochromatic waves with the ROV is particularly focused. Nonlinear elements
were considered in the dynamics the interaction of monochromatic waves with the ROV is particularly focused. Nonlinear elements
were considered in the dynamics in three models with different degrees of physical knowledge.
These encompass nonlinear drag d number and presents a nonconvex in the span of the present in the span of the span of the dynamics in three models with different degrees of physical knowledge.
These encompass nonlinear drag damping, bilinear restoring fo se encompass nonlinear at an objection and the sinking of the sinking of the cabing of the actuators and aration of the actuators. It is concluded that the most complex model including all nonlinear nents produces the wide race oncompact nonlinear and searching similar restering serve, relation potential releating all nonlinear staturation of the detuators. It is concluded that the most complex model including all nonlinear elements produces

relation of the total
relation of redistion forces only contributes with improvements at superficial depths. In order to
simplify the analysis, a ROV with spherical shape was selected and a nonlinear drag characteris-
simp simulations of radiation forces only contributes with improvements at superficial depths. In order to simplify the analysis, a ROV with spherical shape was selected and a nonlinear drag characteristic for this shape was in simplify the analysis, a ROV with spherical shape was selected and a nonlinear drag characteristic for this shape was introduced in the model. This characteristic is variable with the Reynolds number and presents a nonconv consequence its shape was introduced in the model. This characteristic is variable with the Reynolds
number and presents a nonconvex zone just in the span of the operating ROV velocities.
The sinking/lifting operation in a is based on the composition in the component of the composition of the component of two criteria. The sinking/lifting operation in a wide interval of the colle length is characterized by the appearance of the taut-slack ph The sinking/lifting operation in a wide interval of the cable length is characterized by the
appearance of the taut-slack phenomenon, which is described by hefty hands of the cable with
tension magnitudes close to the tole The simularly interactively and in order and is due to the singular interactional operation
appearance of the taut-slack phenomenon, which is described by hefty hands of the cable with
tension magnitudes close to the toler represented at the toterance procedure in the design and the magnitudes cluster of the toterable limits. This unpredictable behavior was observed in simulations of the uncontrolled ROV dynamics, mainly for significative wa embers and restortion of the uncontrolled ROV dynamics, mainly for significative wave steepness and great
depths of operation. In the paper, a solution via control to avoid this phenomenon and in
consequence its negative e species the desired reference behavior variable strength was presented. The control system design
depths of operation. In the paper, a solution via control to avoid this phenomenon and in
consequence its negative effects o

of diverse operations in steady state by means of the controlled system and the free system are
comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed usin of diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed usin liverse operations in steady state by means of the controlled system and the free system are
uparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed using phy for diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed usi for diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed usi for diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings. The comparative stability study is performed usi of diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings. The comparative stability study is performed usin of diverse operations in steady state by means of the controlled system and the free system are comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed usin behavior and possible routes to chaos from the stability regions and parameter settings.
The comparatively investigated under equal perturbations and parameter settings.
The comparative stability study is performed using p The comparative stability study is performed using physical bifurcation parameters and de-
ion methods of high periods based on Poincaré maps and analysis of Cauchy series. The
proximon parameters are divided into two sets The comparative standary orday is performed tame, projected soluted
only perfection methods of high periods based on Poincaré maps and analysis of Cauchy series. The
bifurcation parameters are divided into two sets, namely

bifurcation parameters are divided into two sets, namely once analysis of elastic region parameters (rable length, wave amplitude and frequency, thruster force) and design parameters (ROV shape, mass and cable stiffness). Extra description parameters are triviata mot one beto, andery operators (ROV shape, mass and cable stiffness). One of the main results is the construction of stability regions that are free of those phenomenon on the free rate dominance of the restoring force. They indicate a qualitative diversity in the restoriation of stability regions that are free of these phenomenon on the free parameter space. They indicate a qualitative diversity in doese bullimits). One of the filmit relation of the contributed of othering regions that the technical diversity in the behavior and possible routes to chaos from the stability regions can be extended considerably with the whose square value represents the carriery of the behavior and perturbation and possible routes to chaos from the stability regions to outside.
From the results it was clear that stability regions can be extended considera From the results it was clear that stability regions can be extended considerably with the
use of control, e.g., the control system can avoid the slackness of the cable in a heave operation
despite the presence of wave per From an research of wave perturbations. A particularity of the system is that stability regions
despite the presence of wave perturbations. A particularity of the system is that stability regions
can exhibit not only perio recentred vagy, the control optical can other and starlards of the classe in a neces o-pointed in
the he presence of wave perturbations. A particularity of the system is that stability regions
exhibit not only period-one b respect to proteins of the pretainsment is particularly of the plasmining is applied to each form of that is the dominance of the restoring force of the cable against the hydrodynamic drag force. The limits between the tau

due dumber are only performed to combined but the due of symmics. The function for this in the dominance of the restoring force of the colled against the hydrodynamic drag force. The limits between the taut and taut-slack imits between the taut and taut-slack zones are significantly influence by the wave steepness, whose square value represents the energy of the perturbation. From a practical point of view, the effectiveness of the control mates severe the entire mate stand as the constant and pitch and pitch and pitch and pitch and pitch and pitch angles so that of the control system proposed here begins to fall off when the energy of the actuators is not s where equals that repleads the langy of an perturbation. Trom a practical point of their the effectiveness of the control system proposed here begins to fall off when the energy of the actuators is not sufficient to counte

Acknowledgment

The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Techni-
University of Hamburg-Harburg for the theoretical support. Also it is thanked the National
mcil for Science and Technology, Argenti The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Technical University of Hamburg-Harburg for the theoretical support. Also it is thanked the National Council for Science and Technology, A The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Technical University of Hamburg-Harburg for the theoretical support. Also it is thanked the National Council for Science and Technology, A The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Technical University of Hamburg-Harburg for the theoretical support. Also it is thanked the National Council for Science and Technology, A The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Techni-
University of Hamburg-Harburg for the theoretical support. Also it is thanked the National
meil for Science and Technology, Argenti The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Techni-University of Hamburg-Harburg for the theoretical support. Also it is thanked the National uncil for Science and Technology, Argenti The authors thank Prof. Dr.-Ing. Edwin Kreuzer and Dr.-Ing. Volker Schlegel at the Technical University of Hamburg-Harburg for the theoretical support. Also it is thanked the National Council for Science and Technology, A University of Hamburg-Harburg for the theoretical support. Also it is thanked the National
meil for Science and Technology, Argentine, Universidad Nacional del Sur and Scientific Coop-
folion Project with Germany (AL/A99 -

References

Council for Science and Technology, Argentine, Universidad Nacional del Sur and Scientific Cooperation Project with Germany (AL/A99 - EX II/17) for the financial support of this investigation.
 References

1. AQWA: AQWA eration Project with Germany (AL/A99 - EX II/17) for the financial support of this investigation.
 References

1. AQWA: AQWA Reference Manual, Version 5.3A. Century Dynamics Ltd, UK (2002)

2 Behbahani-Nejad, M., Perkins References

1. AQWA: AQWA Reference Manual, Version 5.3A. Century Dynamics Ltd, UK (2002)

2. Behbahani-Nejad, M., Perkins, N.C.: Hydrodynamic and Geometric Stiffening Effects on

0ut-of-Plane Waves of Submerged Cables. No 1. AQWA: AQWA Reference Manual, Version 5.3A. Century Dynamics Ltd, UK (2002)

2 Behbahani-Nejad, M., Perkins, N.C.: Hydrodynamic and Geometric Stiffering Effects on

the Out-of-Plane Waves of Submerged Cables. Nonlinear 1. Fragma Experimental Relations, N.C.: Hydrodynamic and Geometric Stiffening Effects on

Out-of-Planc Waves of Submerged Cables. Nonlinear Dynamics 13(3), 243-257(1997)

3. Dmitrieva, I., Longovsky, V.: Non-linear harmoni Control-Plane Waves of Submerged Cables. Nonlinear Dynamics 13(3), 243-257(1997)

3. Dmitrieva, I., Longovsky, V.: Non-linear harmonic, subharmonic and chaotic motion of

offshore structures, In: Proceedings of the 8th Int 3. Dmitrieva, I., Longovsky, V.: Non-linear harmmonic, subharmonic and chaotic motion of
3. Dmitrieva, I., Longovsky, V.: Non-linear harmonic, subharmonic and chaotic motion of
shore Structures, In: Proceedings of the 8th 7. Ellermann, K., Kreuzer, E., Markiewicz, M.: Nonlinear Dynamics of the Behaviour of
hore structures, vol 2, pp.205-218, Delft, Netherlands, 1997
4. Feng, Z., Allen, R.: Evaluation of the effects of the communication cabl

Million Structures, vol 2 , pp.205-218, Delft, Netherlands, 1997

4. Feng, Z., Allen, R.: Evaluation of the effects of the communication cable on the dynamics

of an underwater flight vehicle. Ocean Engineering 31, 1019-1 8. Fong, Z., Allen, R.: Evaluation of the effects of the communication cable on the dynamics
in underwater flight vehicle. Ocean Engineering 31, 1019-1035 (2003)
5. Guckenheimer, J., Holmes, P.: Nonlinear Oscillations, Dyn

(1994)

9. S. Guckenheimer, J., Holmes, P.: Nonlinear Oscillations, Dynamical Systems and Bifurcations

2. Guckenheimer, J., Holmes, P.: Nonlinear Oscillations, Dynamical Systems and Bifurcations

2. Stability analysis of the heav 1. F. Hawary, F.: The Ocean Engineering Handbook. CRC Press LLC, Florida, USA (2001)

1. Ellermann, K., Kreuzer, E., Markiewicz, M.: Nonlinear Dynamics of Floating Cran

1. Ellermann, K., Kreuzer, E., Markiewicz, M.: Nonli 10. Huang, Primag, Consequence, Handbook, CRC Press LLC, Florida, USA (2001)

16. El-Hawary, F.: The Ocean Engineering Handbook. CRC Press LLC, Florida, USA (2001)

17. Ellermann, K., Kreuzer, E., Markiewicz, M.: Nonlinear recovery process. In: Proceedings of the 2nd. International Offshore and Polar Engineering 26, 531-546 (1999)

10. Huang, S.: Stability analysis of the heave motion of marine cable-body systems. Ocean

1994)

10. Huang, S. Experiment, A., Helast, A., Helasteries, M., Helasteries, M., Helasteries, M., Helasteries, Shannako of Francisco, Thichester, UK

(1994)

8. Fossen, T.I.: Guidance and Control of Ocean Vehicles. John Wiley&Sons, Chicheste

11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. Uni-
sity of Genova (1998)
12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and
13. Jordán, M.A., Bel 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, P
versity of Genova (1998)
12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-
Drag-Force Models for Moored Floatin 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. Unisity of Genova (1998)
12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and
13. Jordán, M.A., Beltr 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. University of Genova (1998)

12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and

Drag-Force Models

11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. Unisity of Genova (1998)

12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and

14. Jordán, M.A., Be 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. University of Genova (1998)

12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and

Drag-Force Models 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. Unisity of Genova (1998)

12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and

12. Jordán, M.A., Del 11. Indiveri, G.: Modelling and Identification of Underwater Robotic Systems, Ph.D. University of Genova (1998)
12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and
Drag-Force Models f 11 marten, cm monetary dia recommendation of cultural resolute system
12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitati

Drag-Force Models for Moored Floating Structures. Ocean Engineering 33, 1 12. Jordán, M.A.: On-line Identification and Convergence Analysis of Excitation-Force and
12. Jordán, M.A.; On-line Identification and Convergence Analysis of Excitation-Force and
13. Jordán, M.A., Beltrán-Aguedo, R.: Nonl 2. Jostain, Jamel Val me Jacaniacator and Unitaryces Emayles of Entitated Price and

Drag-Force Models for Moored Floating Structures. Ocean Engineering 33, 1161-1213 (2006)

13. Jordán, M.A., Beltrán-Aguedo, R.: Nonlinear Eng Fried Instant Friedman, Scientification of morring tori Trist (2000)

13. Jordán, M.A., Beltrán-Aguedo, R.: Nonlinear identification of mooring lines in dynamic

operation of floating structures. Ocean Engineering 31, 26. Jordan, Ericular Espacio, 46. Fossima, Tamineca en Indonesia en Indonesia en Indiana.
14. Jordán, M.A., Beltrán-Aguedo, R.: Optimal identification of potential-radiation hydrody-
14. Jordán, M.A., Beltrán-Aguedo, R.: O

11. Wackaii, First, Sciencia, Equitor, E.: Openina Reinfindation is potential relation by the system (Figure 1559-1914 (2004).

15. Jordán, M.A., Bustamante, J.L.: Diseño de un observador no-lineal para Robots sub-

46. Jo 31, 1859-1914 (2004).

15. Jordán, M.A., Bustamante, J.L.: Diseño de un observador no-lineal para Robots sub-

acuáticos en operación de ascenso/descenso. In: Proceedings of the III Jornadas Argentinas de

Robótica, San Ju 15. Jordán, M.A., Bustamante, J.L.: Diseño de un observador no-lineal para Robots sub-
16. Jordán, M.A., Bustamante, J.L.: Diseño de un observador no-lineal para Robots sub-
26. San Juan, June 3-4, 2004
16. Kijima, K., Fos 16. Journal on Maximilian on Sound and Vibration 140 Hardware Rest Angentinas de
Rolloftica, San Juan, June 3-4, 2004
16. Kijima, K., Fossen, T.I. (ed.): Control Applications in Marine Systems, Pergamon Press
(2000)
17. Kl

(2000)

19. Plaut, R.H., Fossen, T.I. (ed.): Control Applications in Marine Systems, Pergamon Press

19. Kijima, K., Fossen, T.I. (ed.): Control Applications in Marine Systems, Pergamon Press

19. Kijima, K., Fossen, T.I. (ed.): C

16. Kijima, K., Fossen, T.I. (ed.): Control Applications in Marine Systems, Pergamon Press
(2000)
17. Kleezka, W., Kreuzer, E.: On the systematic analytic-numeric bifurcation analysis. Non-
linear Dynamics 7, 149-163 (1995 20. Rosenwasser, E.C. On the systematic analytic-numeric bifurcation analysis. Non-
20. Rieczka, W., Kreuzer, E.: On the systematic analytic-numeric bifurcation analysis. Non-
20. Rosenwasser, E.149-163 (1995)
18. Papazogl 21. Smith, R.J: Taut-slack dynamics of a vertically suspended subseau of MHz. Proceding in water. Journal on Sound and Vibration 140(1),103-115 (1990)
18. Papazoglou, V.J., Mavrakos, A., Triantaffilou, M.S.: Nonlinear cabl of the International Conference on Offshore Mechanics and Arctic Engineering (OMAE) vol 1,
B. Papazoglou, V.J., Mavrakos, A., Triantaffilou, M.S.: Nonlinear cable response and model
testing in water. Journal on Sound and N 18. Papazoglou, V.J., Mavrakos, A., Triantaffilou, M.S.: Nonlinear cable respetesting in water. Journal on Sound and Vibration $140(1), 103-115$ (1990)
19. Plant, R.H., Farmer, A.L., Holland, M.M.: Bouncing-Ball Model of '